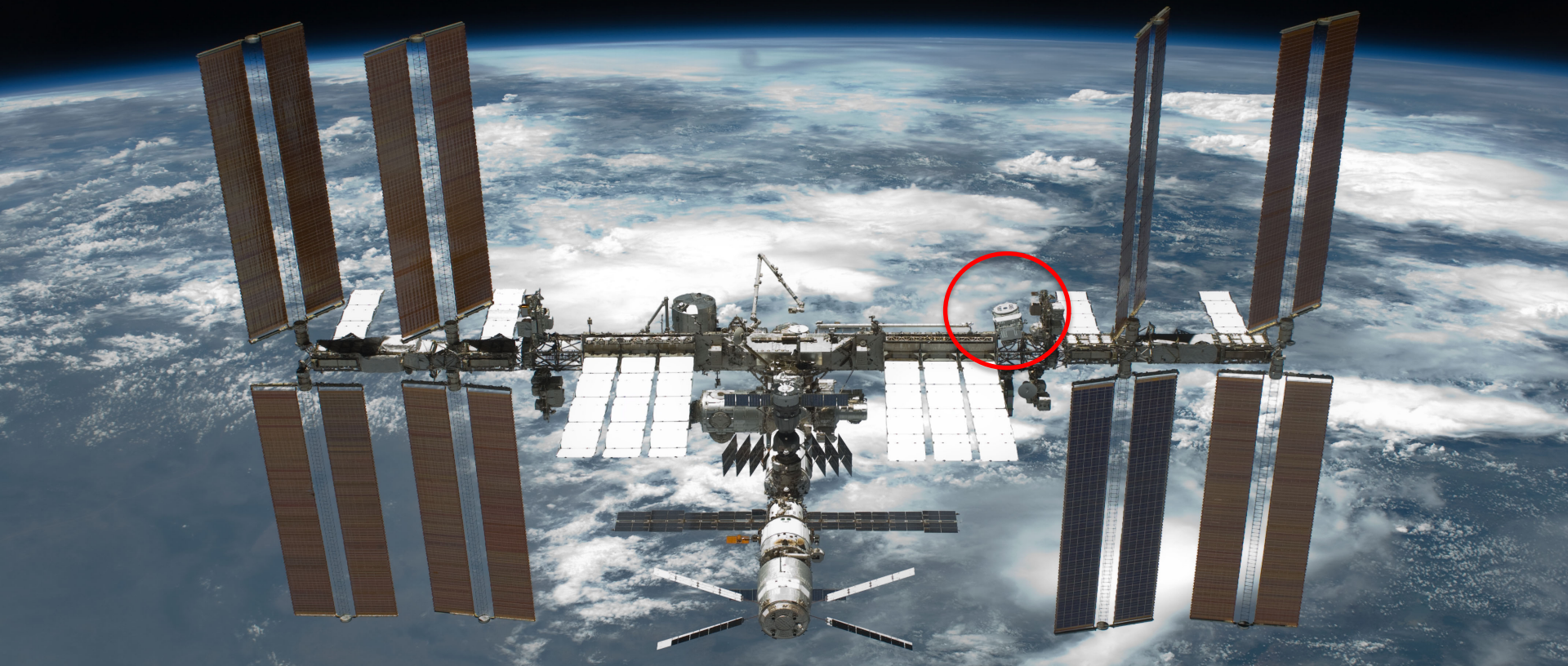


The AMS latest results and the impact on the design of future Cosmic Ray Space Experiments



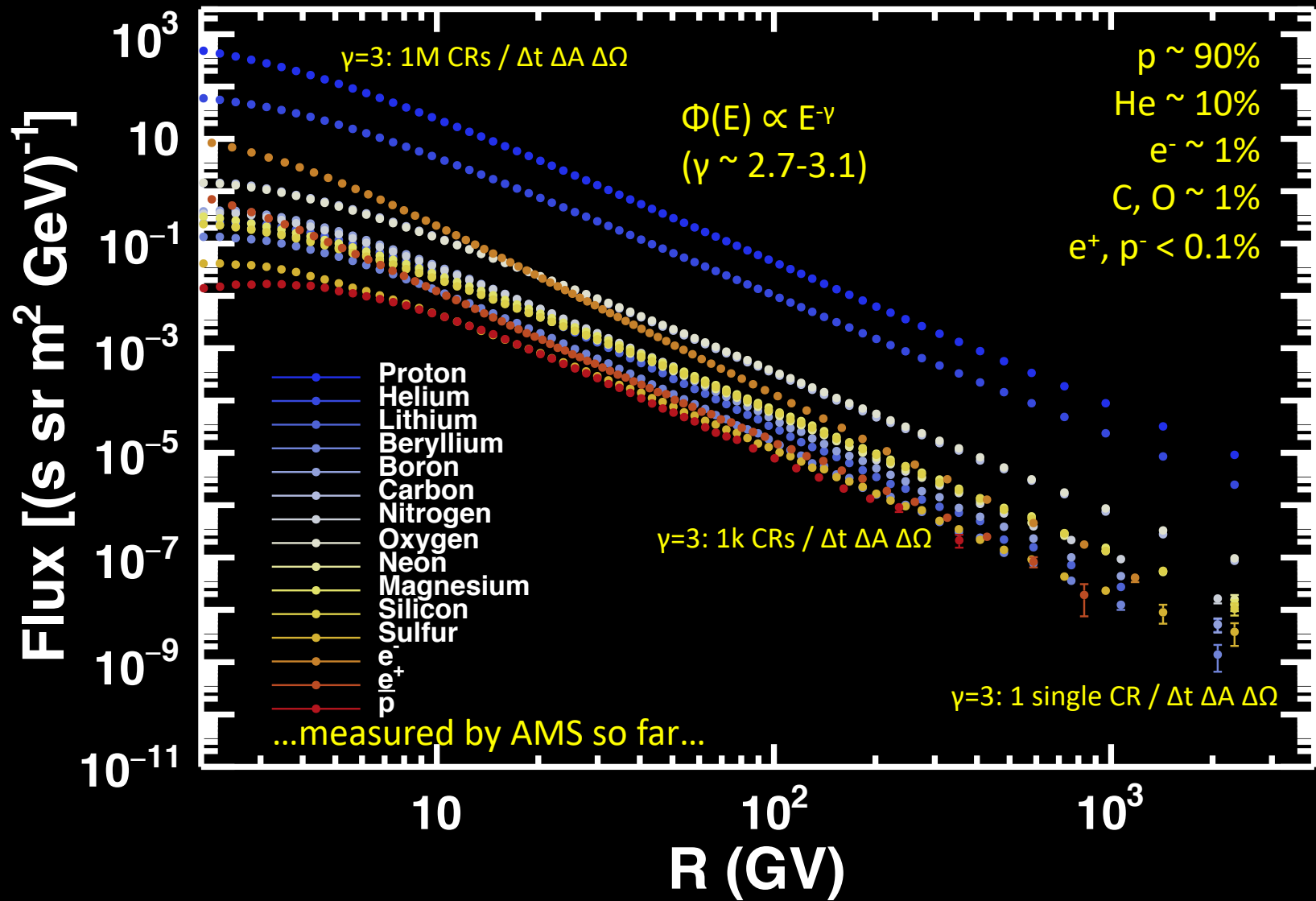
Charged Cosmic Rays in Space

The talk is focused on:

- Charged Cosmic Rays experiments
- Experiments in space

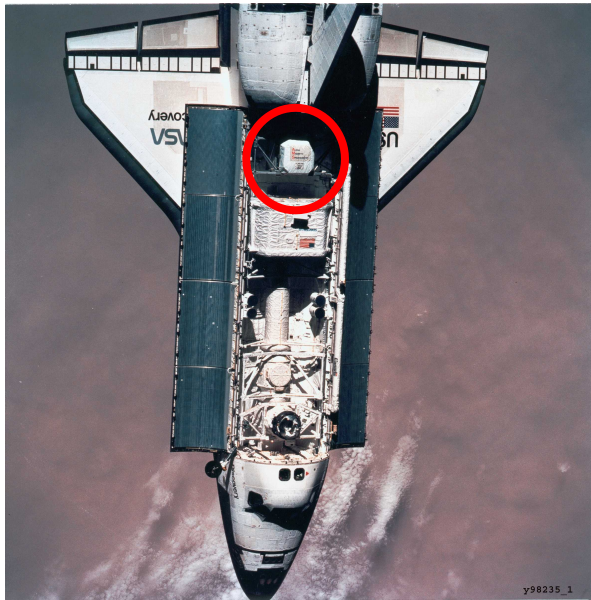


Charged cosmic rays



- Current and past experiments – key concepts/detectors
- Latest AMS results
- Future/proposed 4π experiments
 - HERD
 - ALADInO
 - AMS-100

Current and past experiments – key concepts/detectors



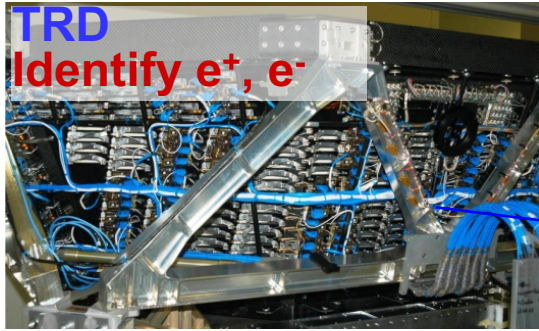
Alpha Magnetic Spectrometer - 01 (AMS-01)

- ❑ ~ 2 tons
- ❑ Same orbit of the ISS and of AMS-02
- ❑ 10 days of mission on board the Space Shuttle Discovery mission STS-91, June 1998

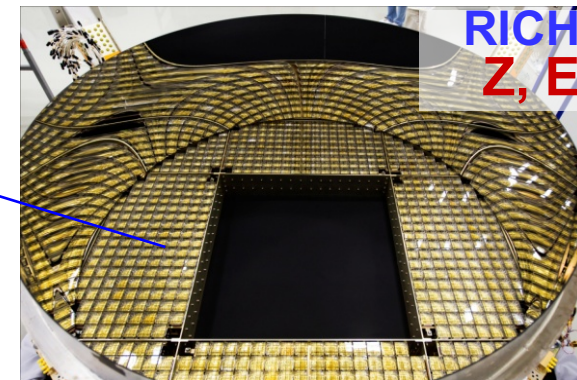
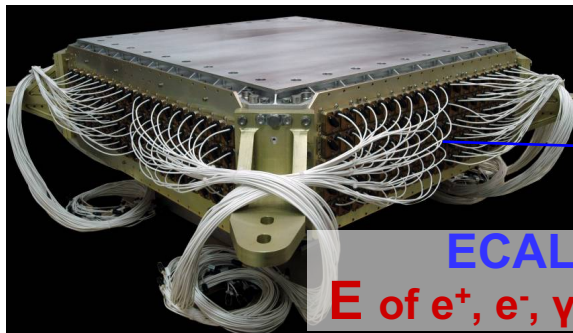
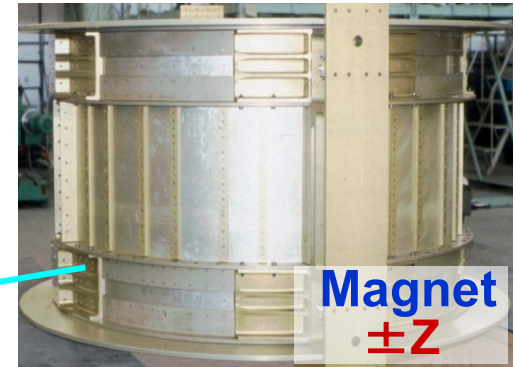
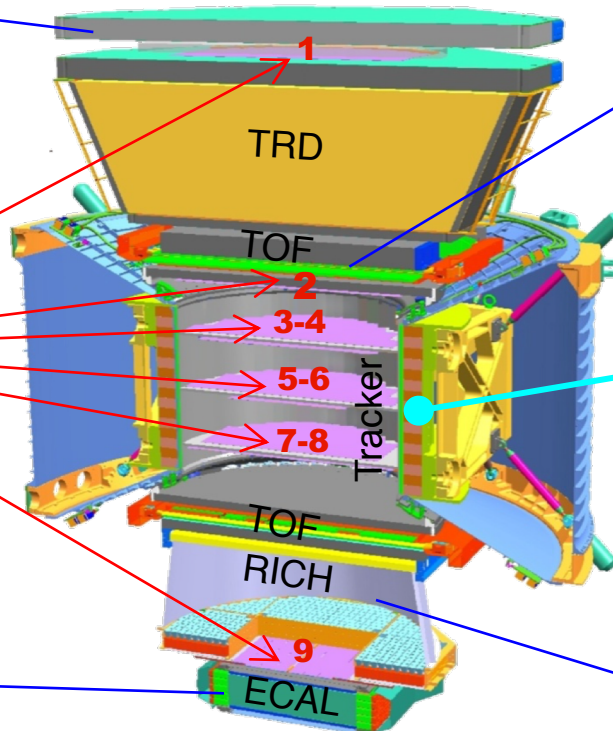
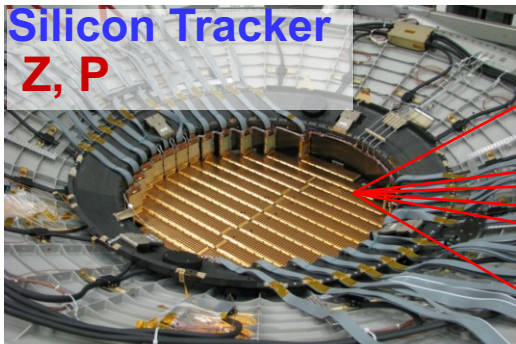
Payload for Antimatter Matter Exploration and Light-nuclei Astrophysics (PAMELA)

- ❑ 470 Kg
- ❑ On board Resurs-DK1 satellite
- ❑ 15 June 2006 – 7 February 2016



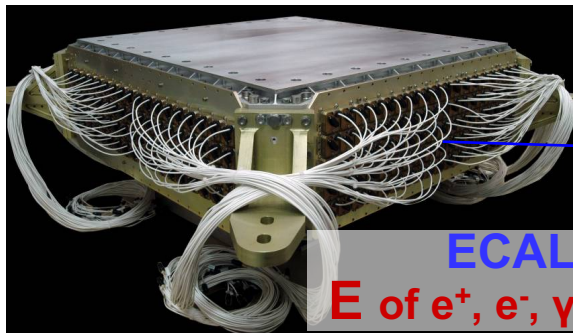
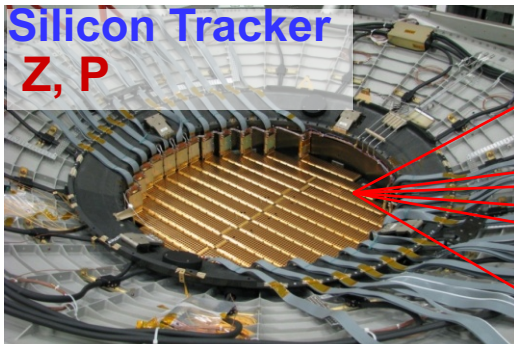


Z , P independently measured by
Tracker, RICH, TOF and ECAL



On ISS since 16 May 2011

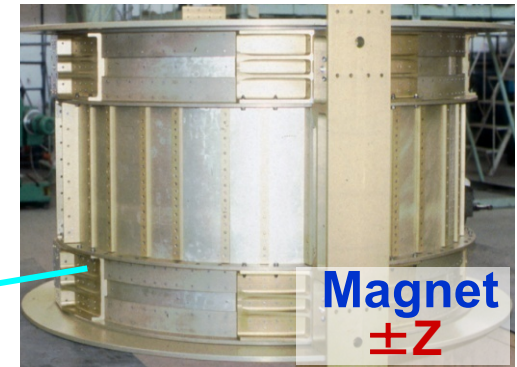
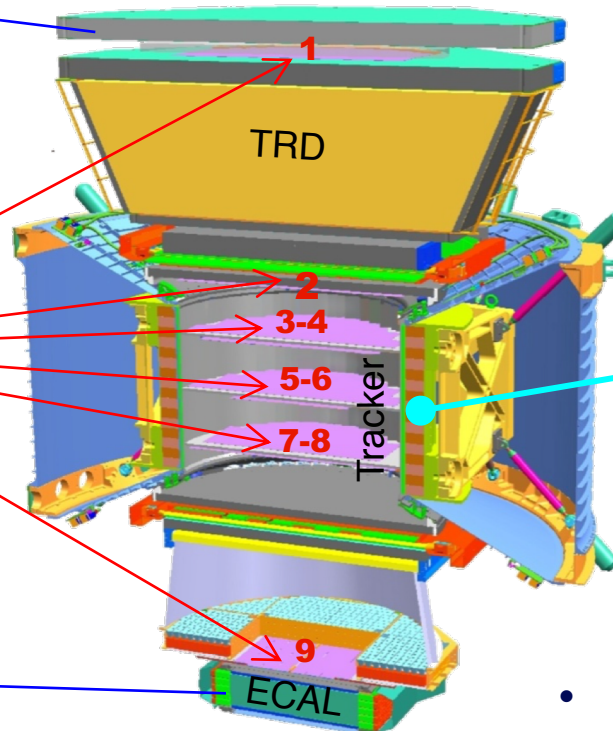
Key concepts/detectors



Techniques:

- Transition Radiation
- Shower development topology

- Energy/Momentum (E/p) match
- neutrons produced in the hadronic shower

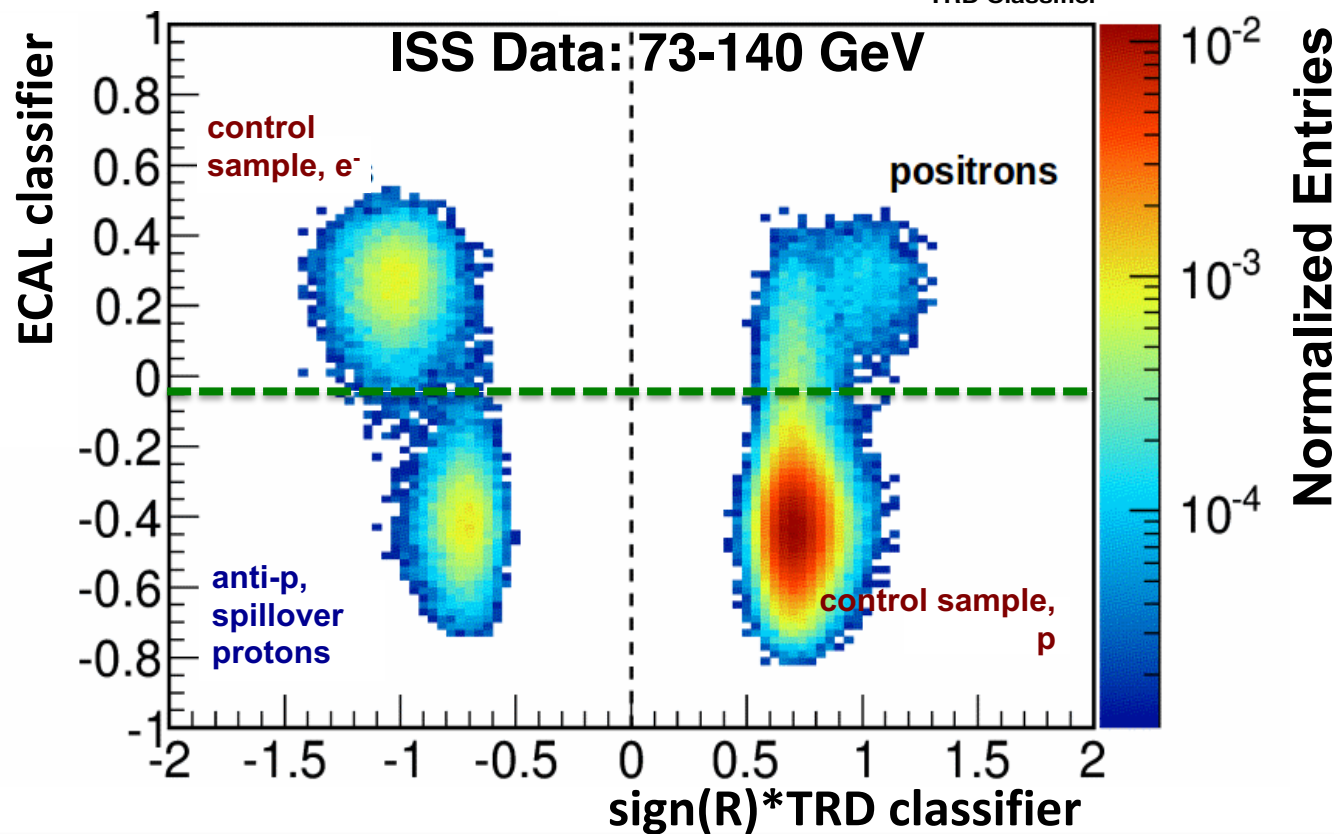
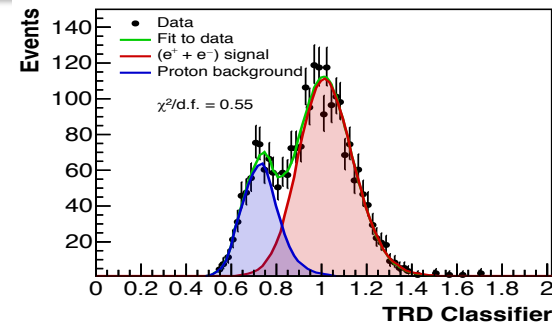
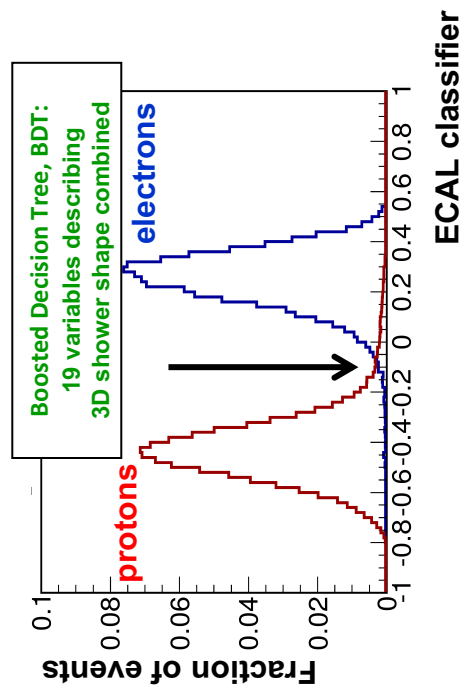


Electron/proton separation:

- e^- wrt the p background
- e^+ wrt to the p background
- anti- p wrt to the e^- background
- γ 's wrt the p background

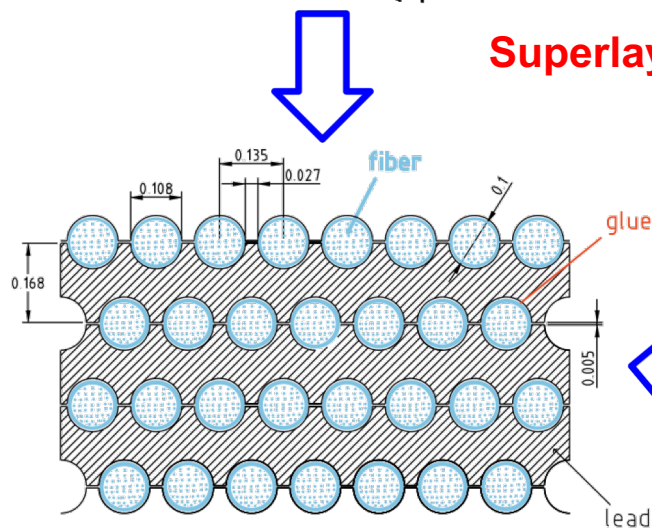
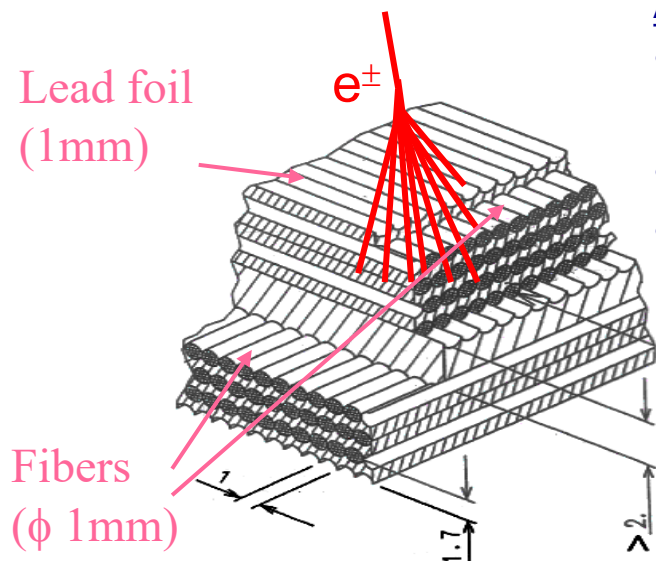
Key concepts/detectors

One important lesson from the AMS experiment is the importance of the redundancy: use one detector to create control sample for another one.

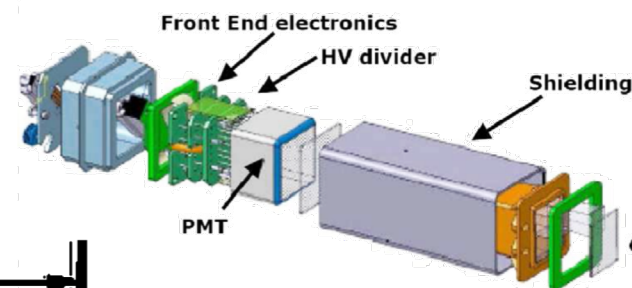
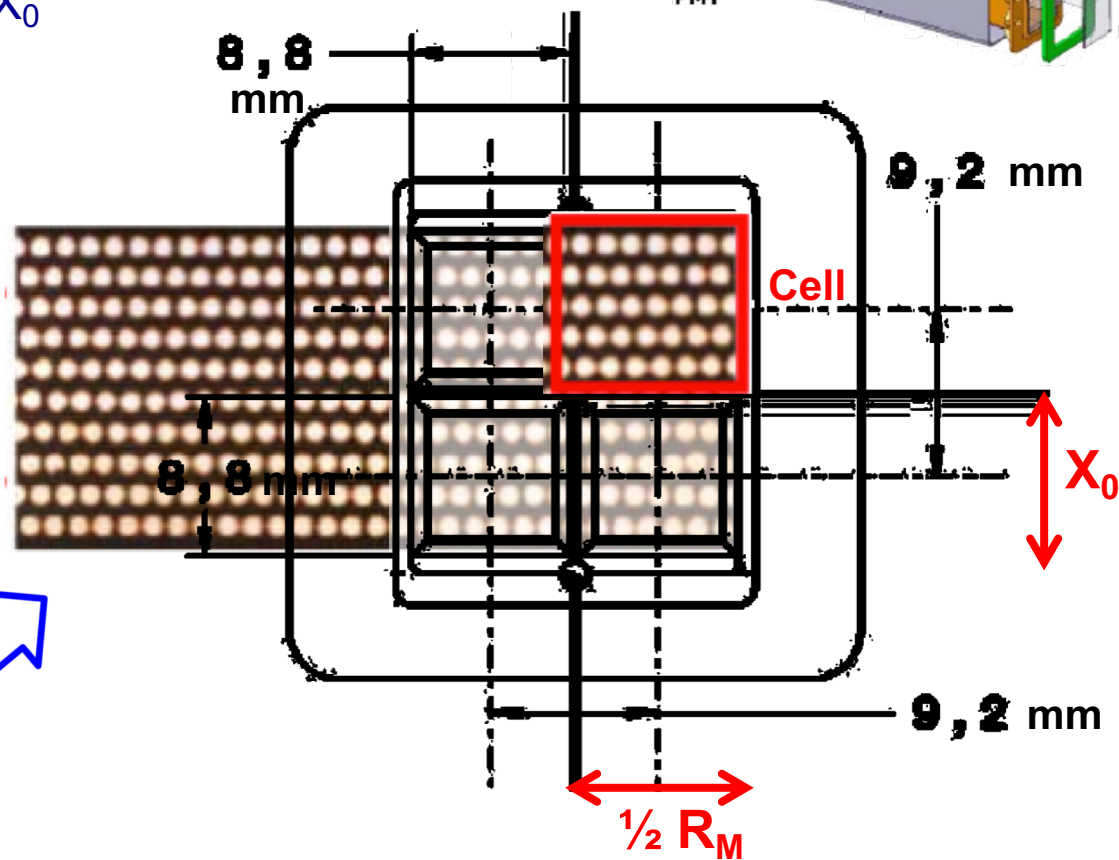


AMS ECAL:

- Lead-SciFi sampling calorimeter
- 18 layers (9 super-layers)
- $17 X_0$



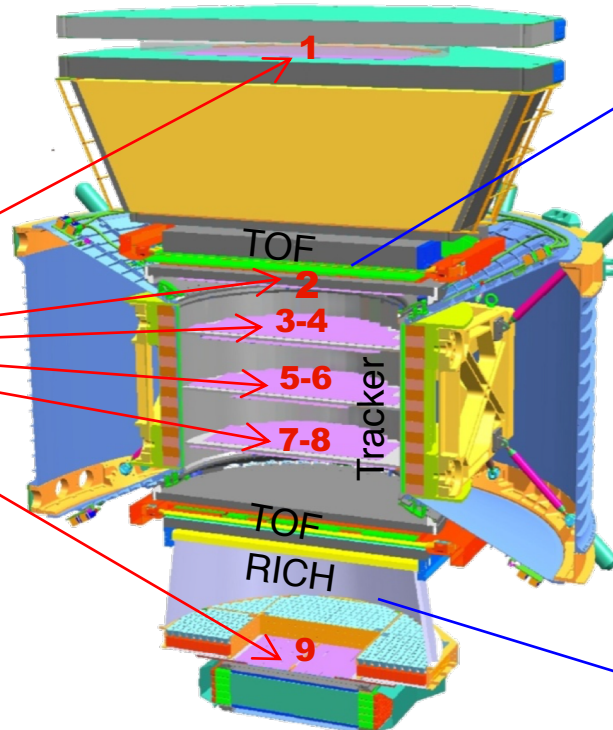
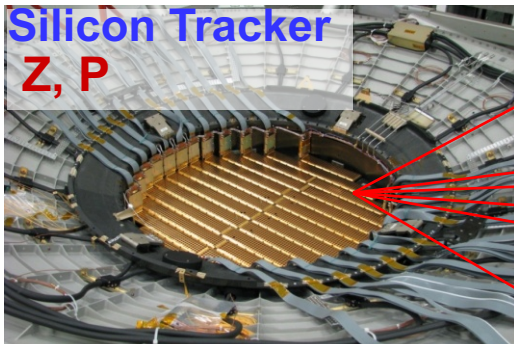
Superlayer



Key concepts/detectors

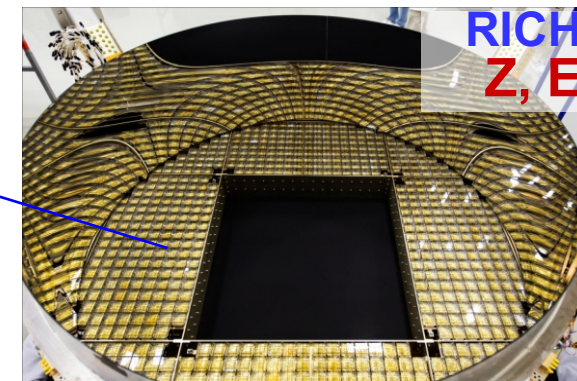
Techniques:

- dE/dx
- number of photons in the Cherenkov radiation

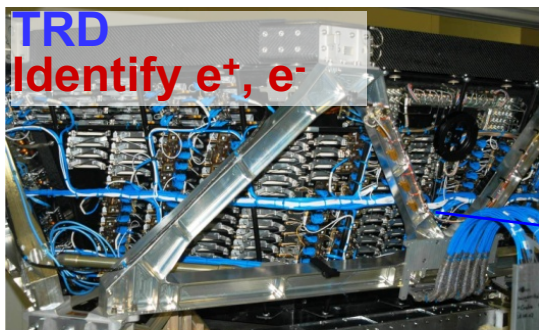


Charge measurement:

- identify the different nuclear species
 - control the fragmentations (if multiple measurements along the detector)

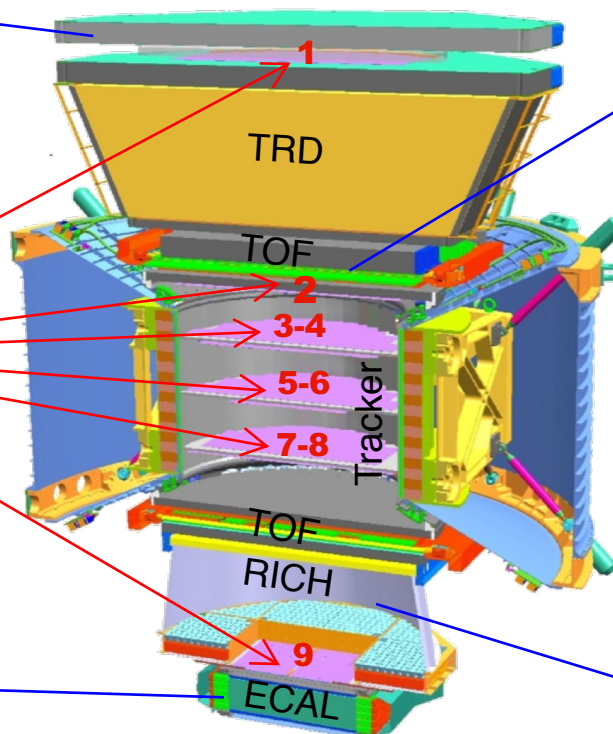
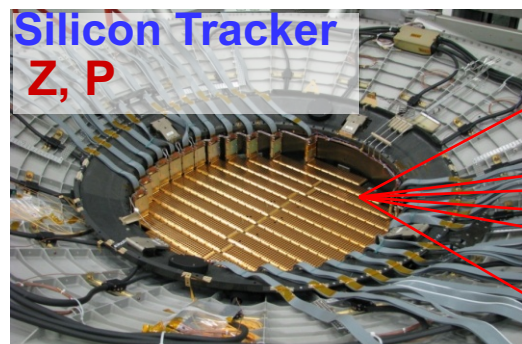


Key concepts/detectors



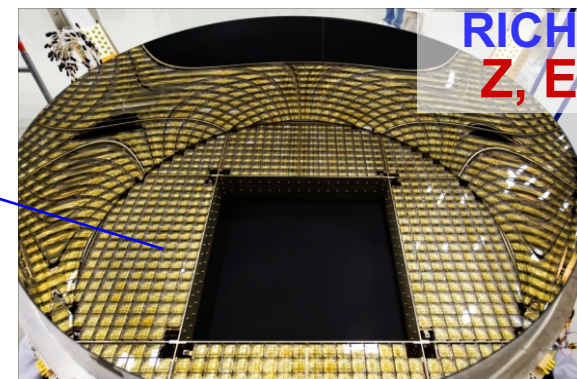
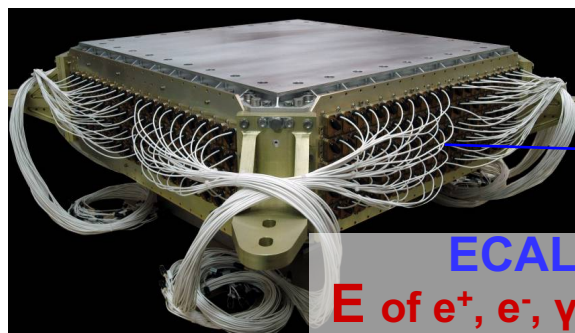
Techniques:

- dE/dx
- number of photons in the Cherenkov radiation

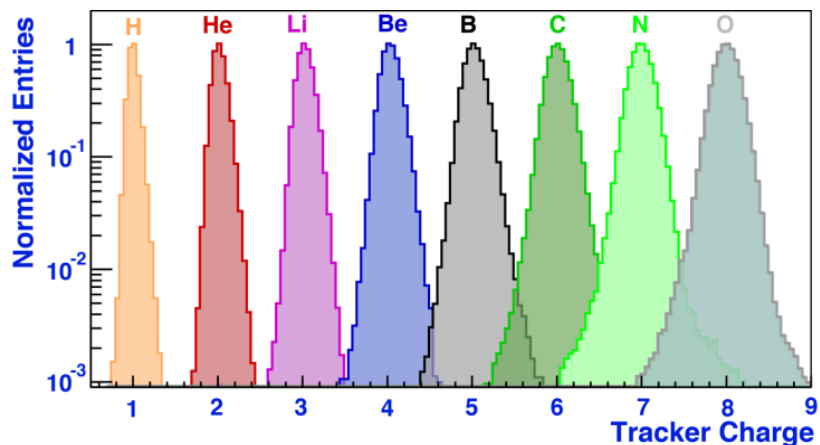
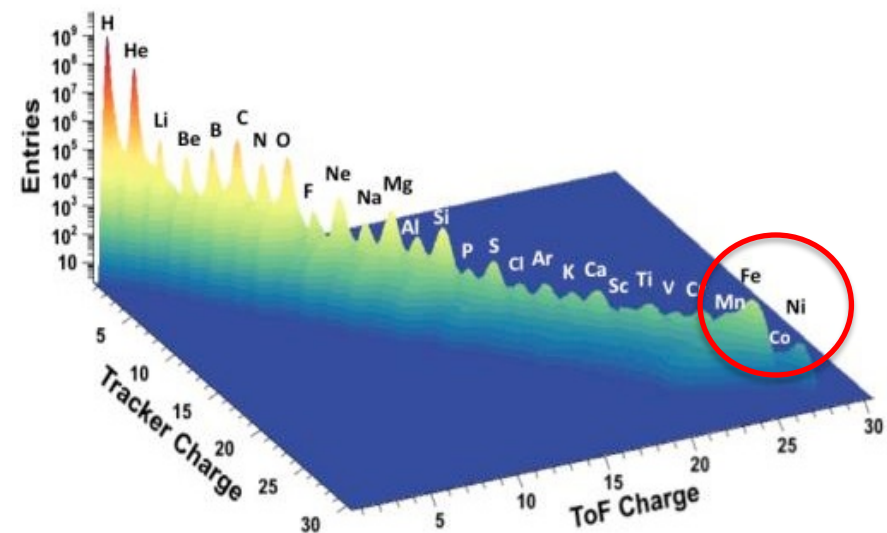
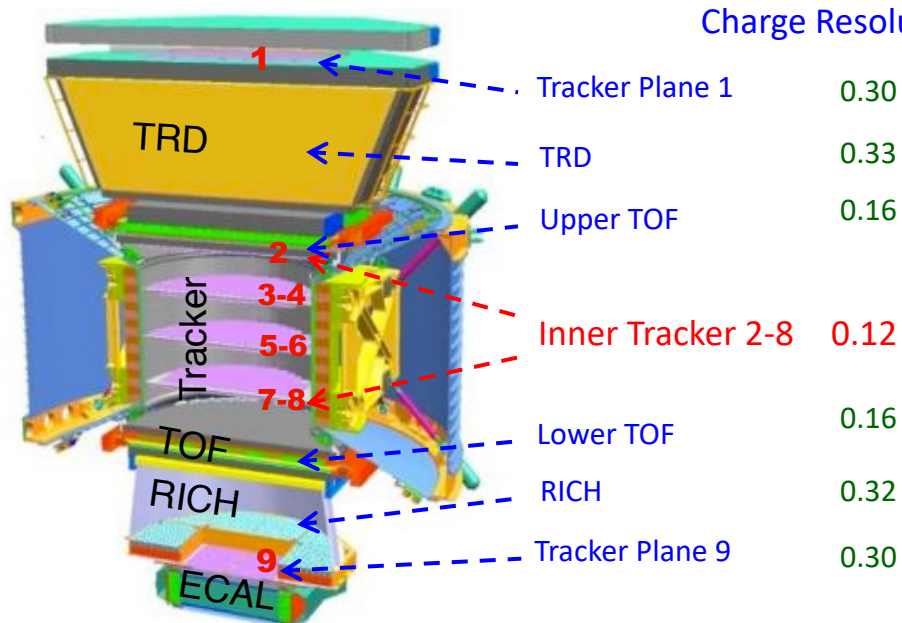


Charge measurement:

- identify the different nuclear species
 - control the fragmentations (if multiple measurements along the detector)



Key concepts/detectors



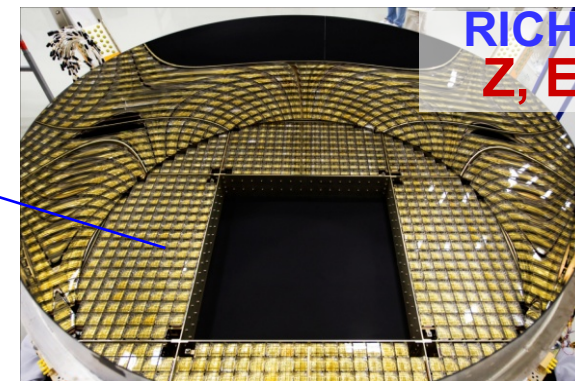
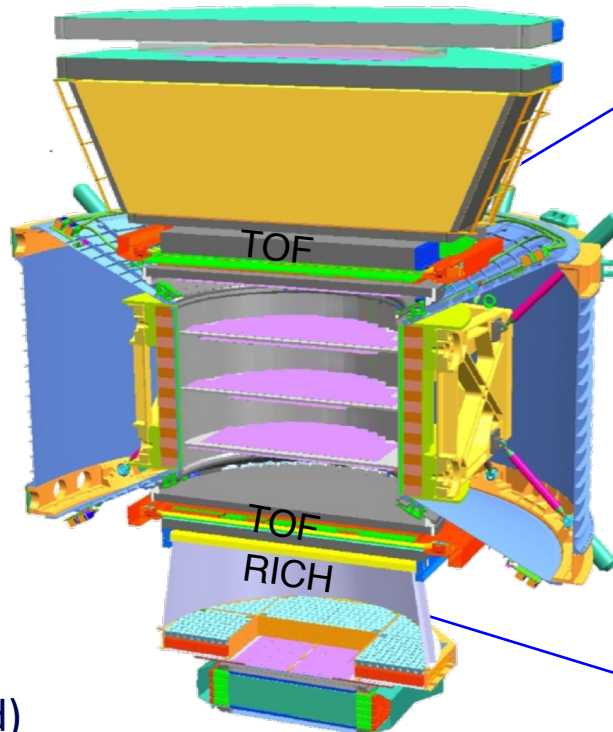
Key concepts/detectors

β measurement:

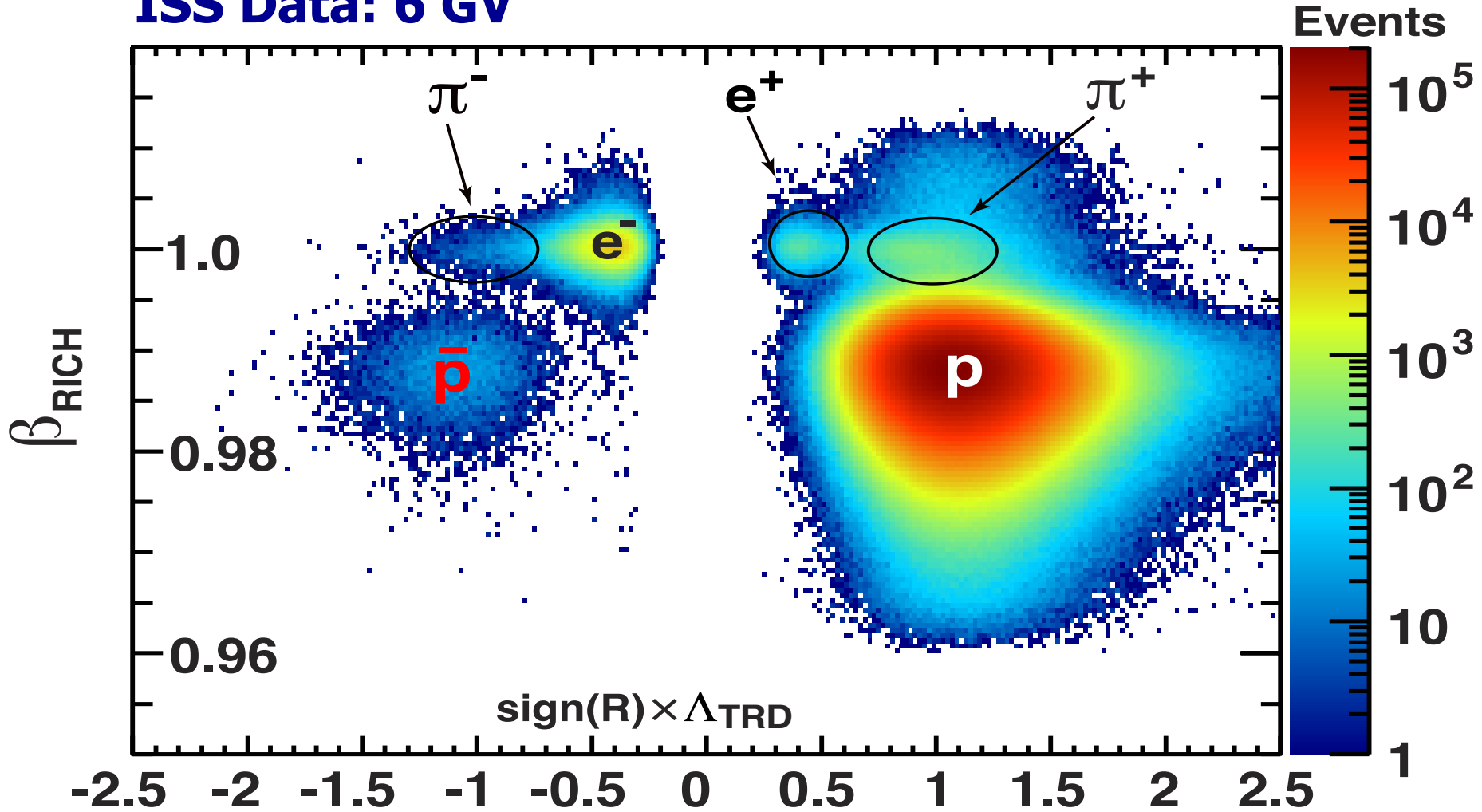
- identify the different isotopes (d/p, $^3\text{He}/^4\text{He}$, $^7\text{Li}/^6\text{Li}$, $^{10}\text{Be}/^9\text{Be}$, $^{27}\text{Al}/^{26}\text{Al}$, ...)
- control the quality of the momentum/energy measurement (check on the mass)

Techniques:

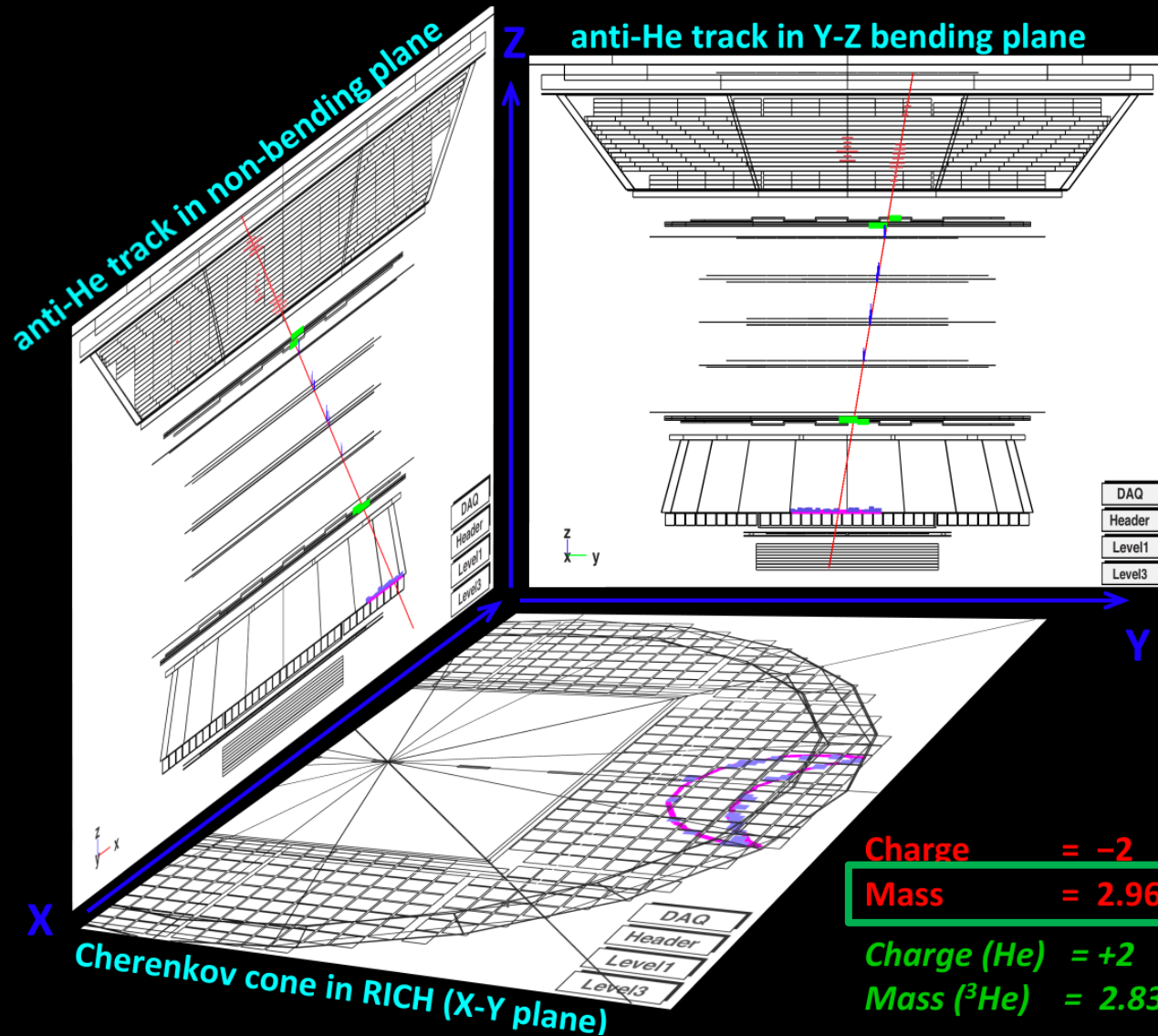
- Time of Flight (ToF)
- Cherenkov (ring or threshold)
- Transition Radiation (measuring γ)



ISS Data: 6 GV



Key concepts/detectors



Charge = -2

Mass = $2.96 \pm 0.33 \text{ GeV}/c^2$

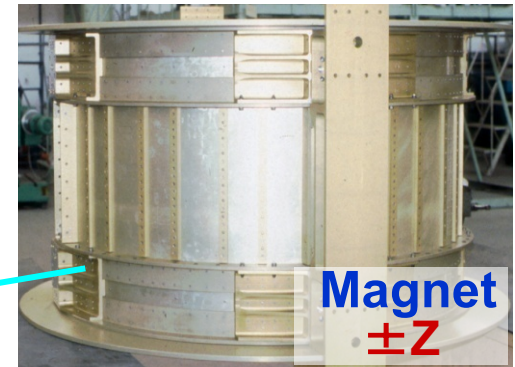
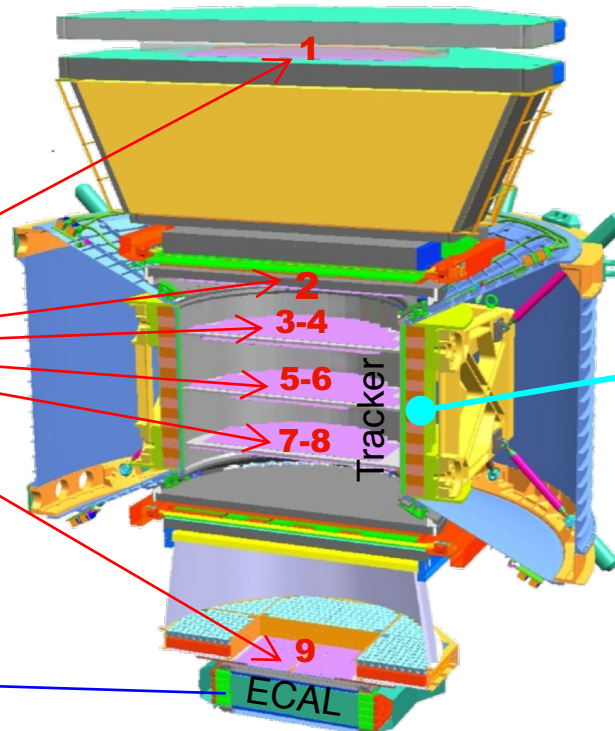
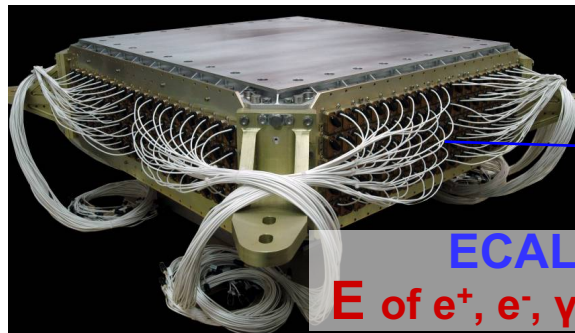
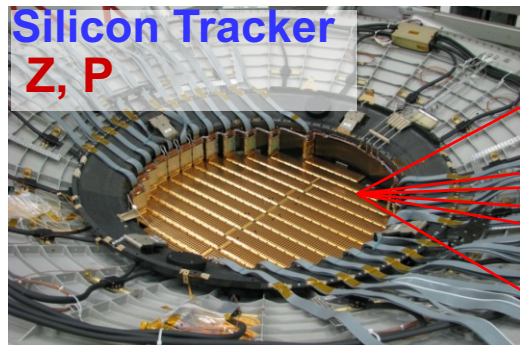
Charge (He) = +2

Mass (^3He) = $2.83 \text{ GeV}/c^2$

Key concepts/detectors

Energy/momentum measurement:

- search for spectral features



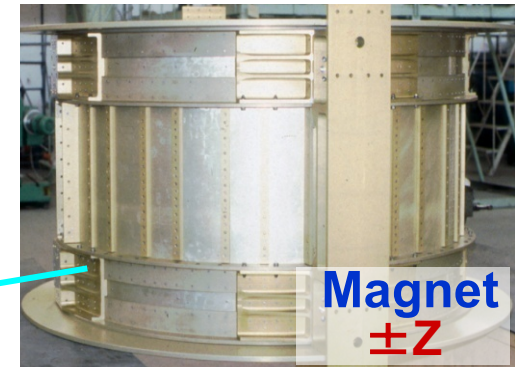
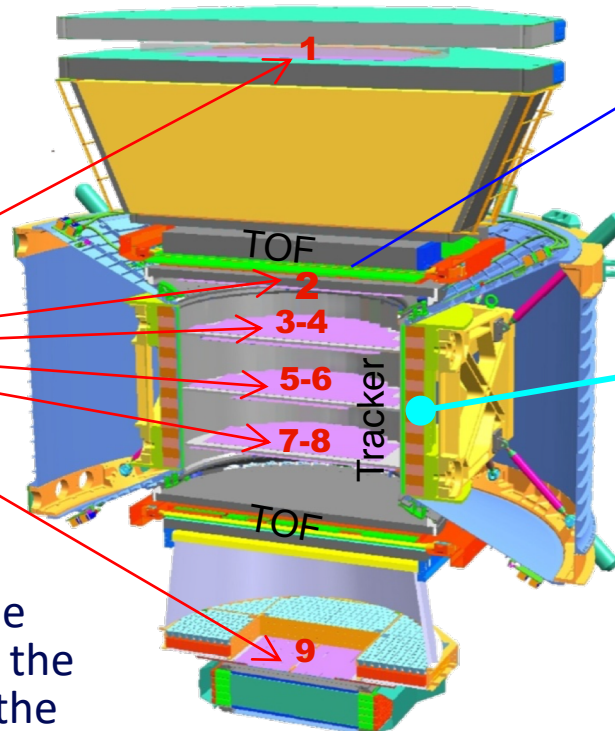
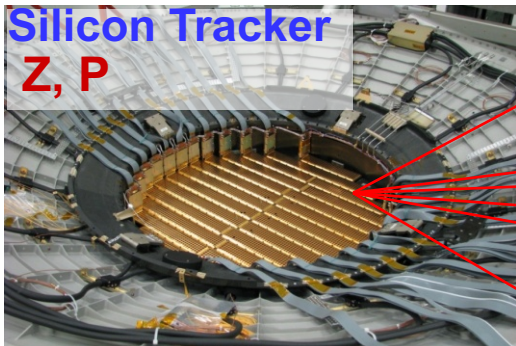
Techniques:

- Spectrometry
- Calorimetry
- Transition Radiation (measuring γ)

Key concepts/detectors

Charge sign measurement:

- matter/anti-matter

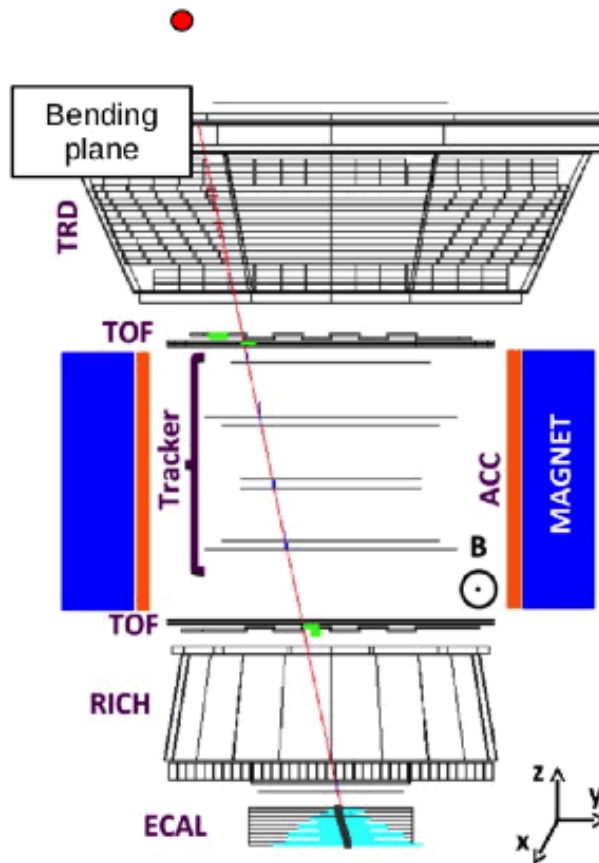


The intensity of the magnetic field (B), the lever arm (L) and the spatial resolution (σ_x) determine the momentum resolution (δp) and the detector Maximum Detectable Rigidity, MDR ($\delta p/p=1$):

$$\text{MDR} \propto B L^2 / \sigma_x$$

Techniques:

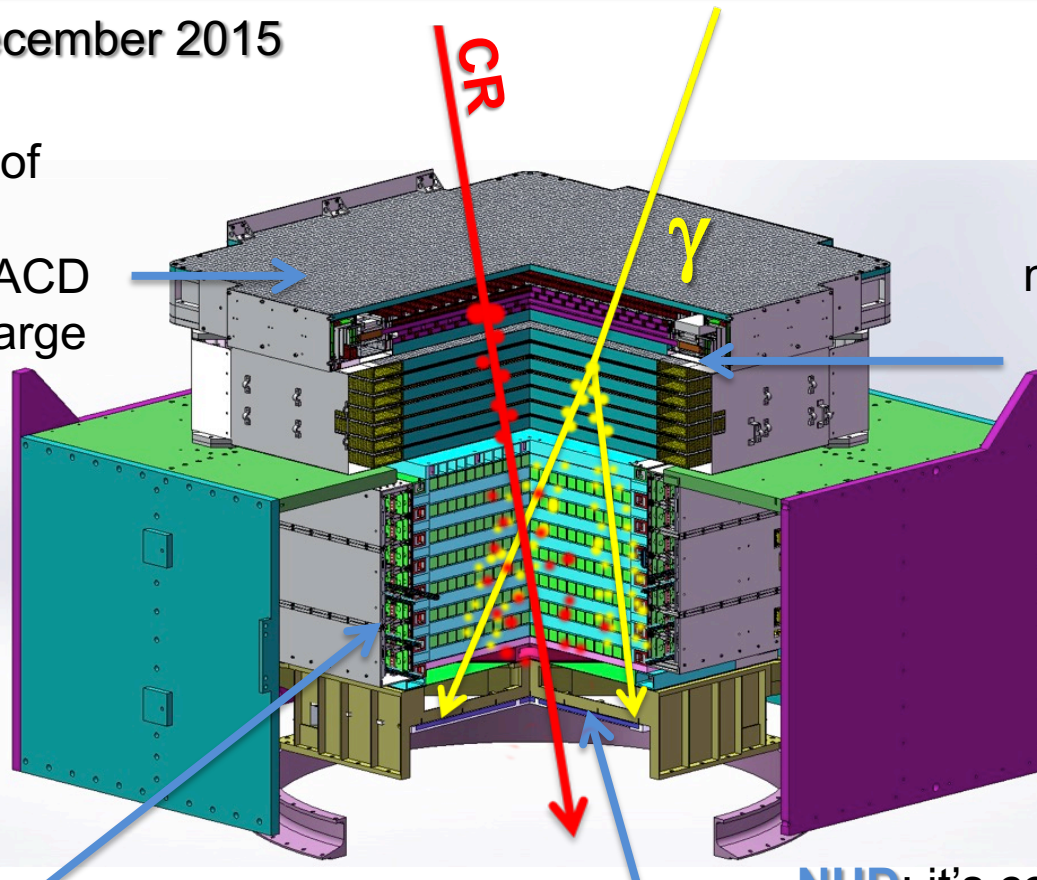
- Spectrometry + ToF



	e^-	p	He
TRD 20 layers			
TOF 4 layers			
TRK 9 layers			
RICH			
ECAL 20 layers			

In orbit since 17 December 2015

PSD: double layer of scintillating strip detector acting as ACD (anti-counter) + charge measurement



STK: 6 tracking double layer + 3 mm tungsten plates. Used for particle track, charge measurement and photon conversion ($\sim 2 X_0$)

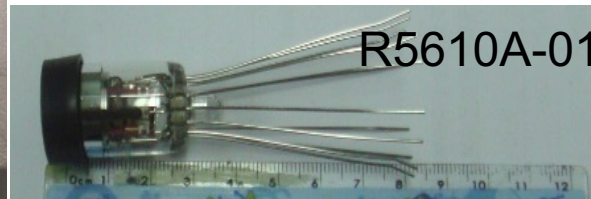
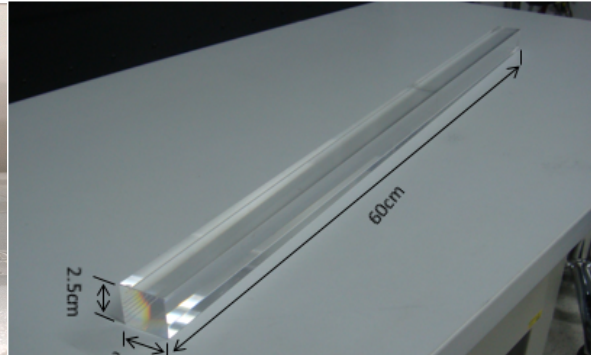
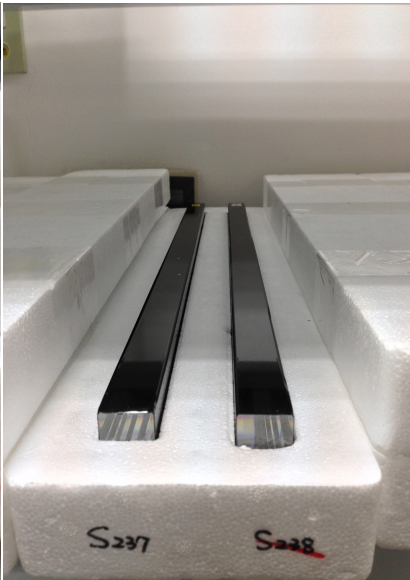
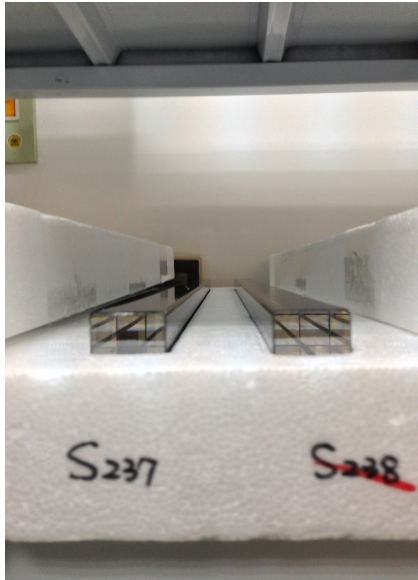
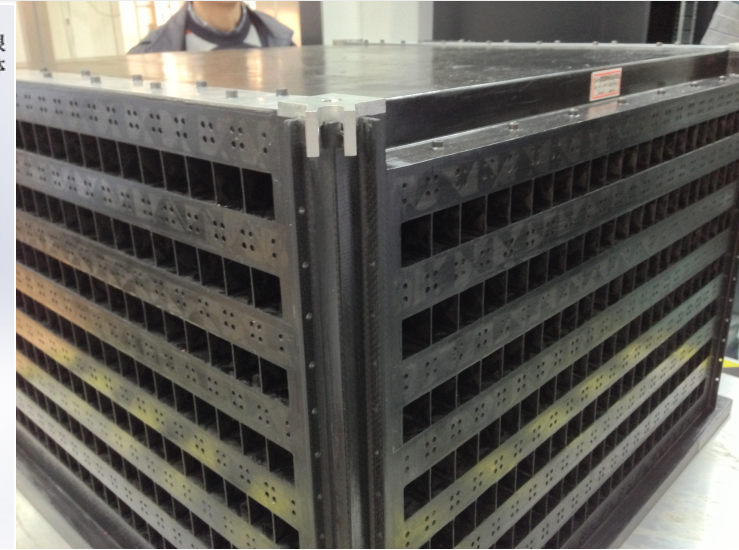
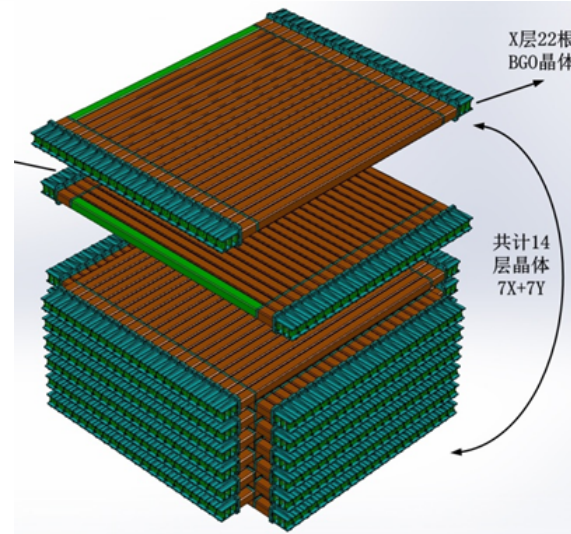
BGO: the calorimeter is made of 308 BGO bars in hodoscopic arrangement ($\sim 31 X_0$). Performs energy measurements, hadron/lepton identification (*e/p rejection*), and trigger

NUD: it's complementary to the BGO e/p rejection, by measuring the thermal neutron shower activity. Made up of boron-doped plastic scintillator

Shower development topology: segmentation

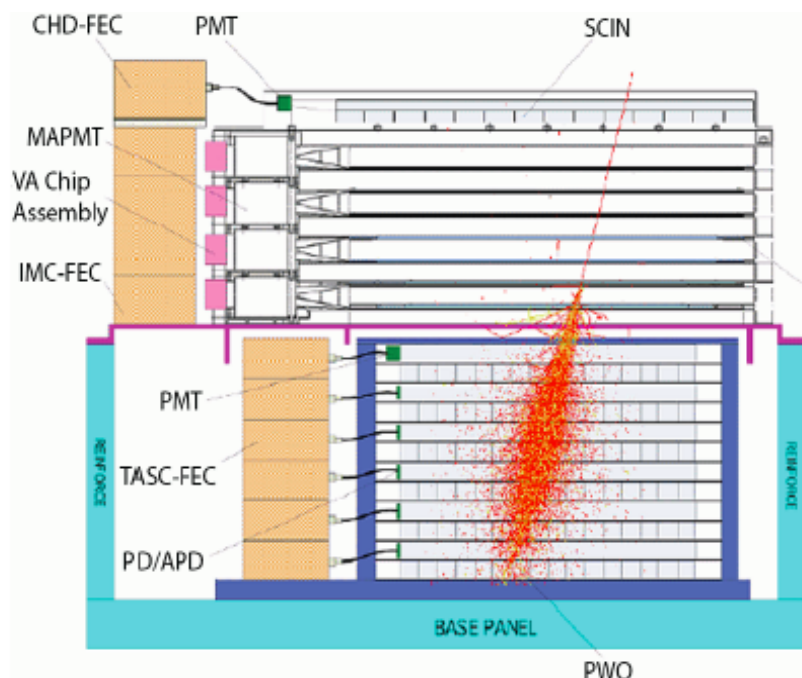
DAMPE BGO:

- homogeneous calorimeter
- $\sim 31 X_0$
- 14 layers
($\sim 2X_0$ per layer)



308 bars
616 PMTs

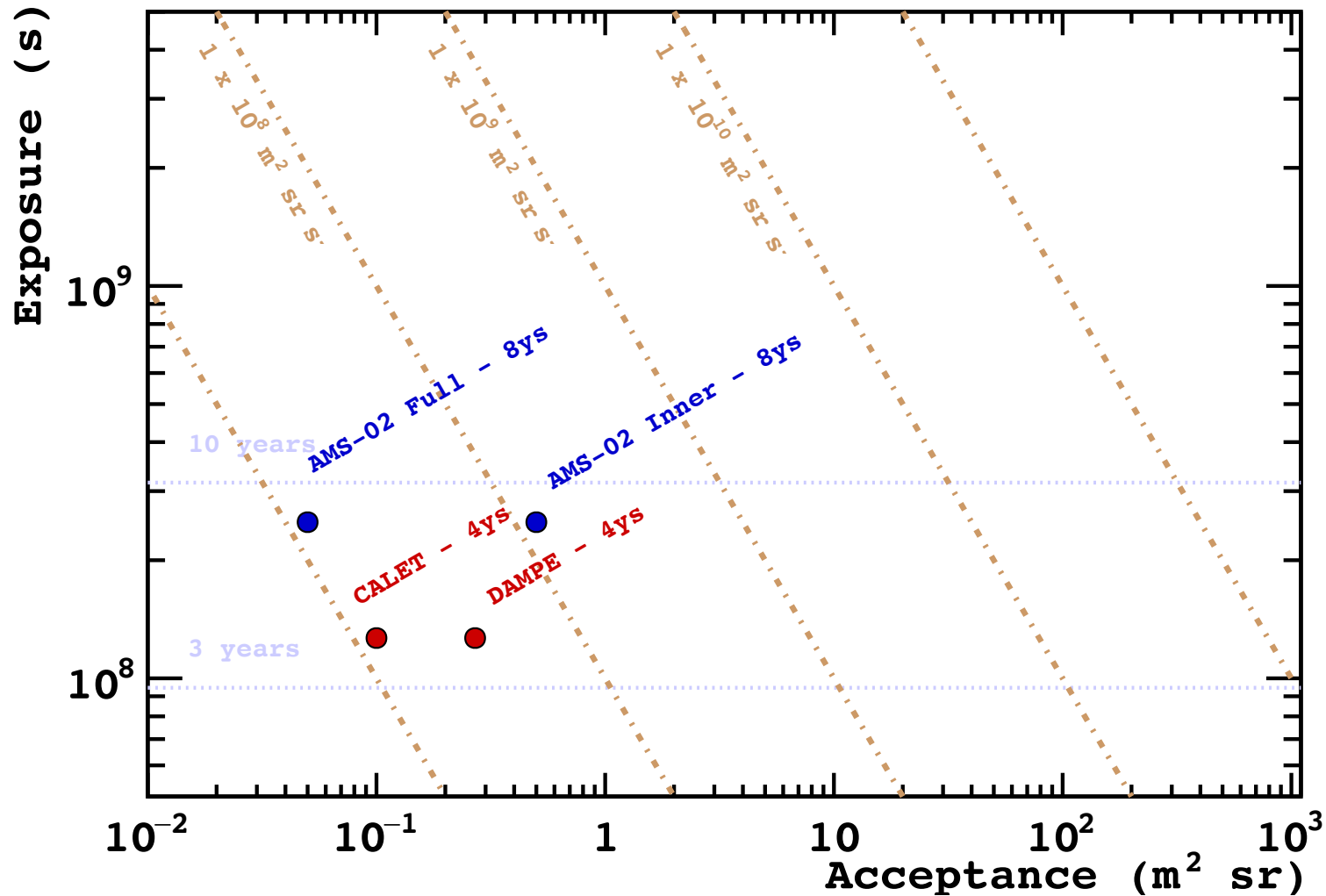
In orbit since 19 August 2015

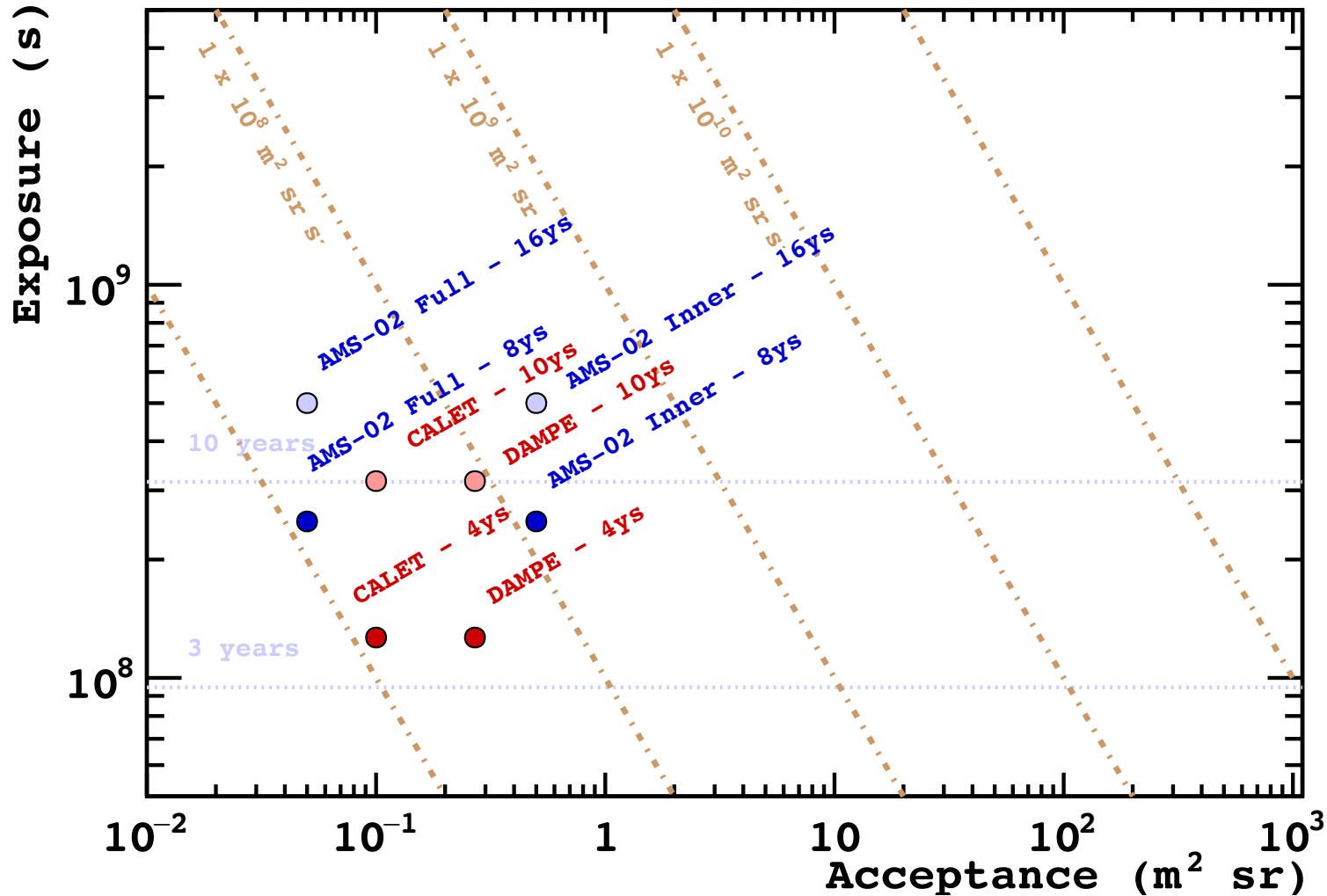


- CHD - CHarge Detector (CHD)**
(Charge Measurement $Z=1-40$)
- IMC - Imaging Calorimeter (IMC)**
(Particle ID, Direction)
Total Thickness of Tungsten (W): $3 X_0$, $0.1 \lambda_I$
Layer Number of SciFi Belts: 8 Layers $\times 2(X,Y)$
- TASC - Total Absorption Calorimeter (TASC)**
(Energy Measurement, Particle ID)
PWO 20mm x 20mm x 320mm
Total Depth of PWO: $27 X_0$ (24 cm), $1.2 \lambda_I$

	CHD (Charge Detector)	IMC (Imaging Calorimeter)	TASC (Total Absorption Calorimeter)
Function	Charge Measurement ($Z = 1 - 40$)	Arrival Direction, Particle ID	Energy Measurement, Particle ID
Sensor (+ Absorber)	Plastic Scintillator : 2 layers Unit Size: 32mm x 10mm x 450mm	SciFi : 16 layers Unit size: 1mm ² x 448 mm Total thickness of Tungsten: $3 X_0$	PWO log: 12 layers Unit size: 19mm x 20mm x 326mm Total Thickness of PWO: $27 X_0$
Readout	PMT+CSA	64 -anode PMT+ ASIC	APD/PD+CSA PMT+CSA (for Trigger)

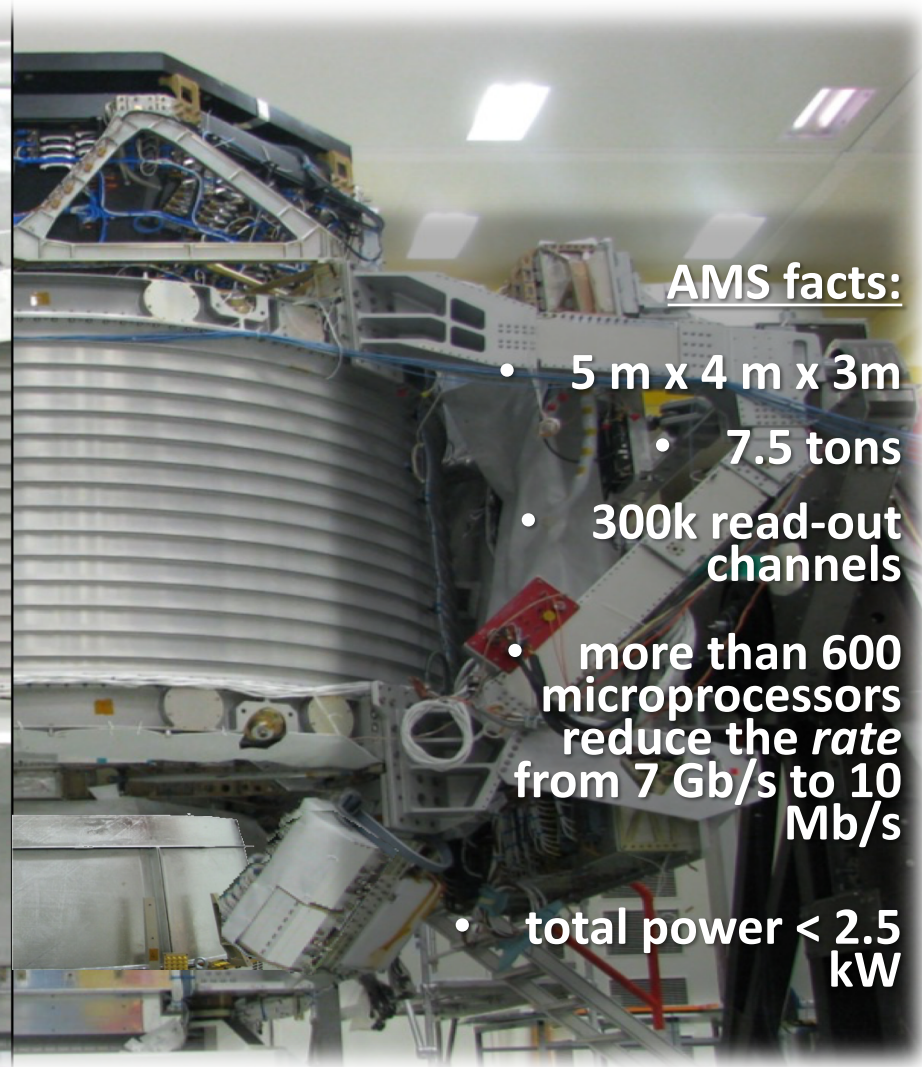
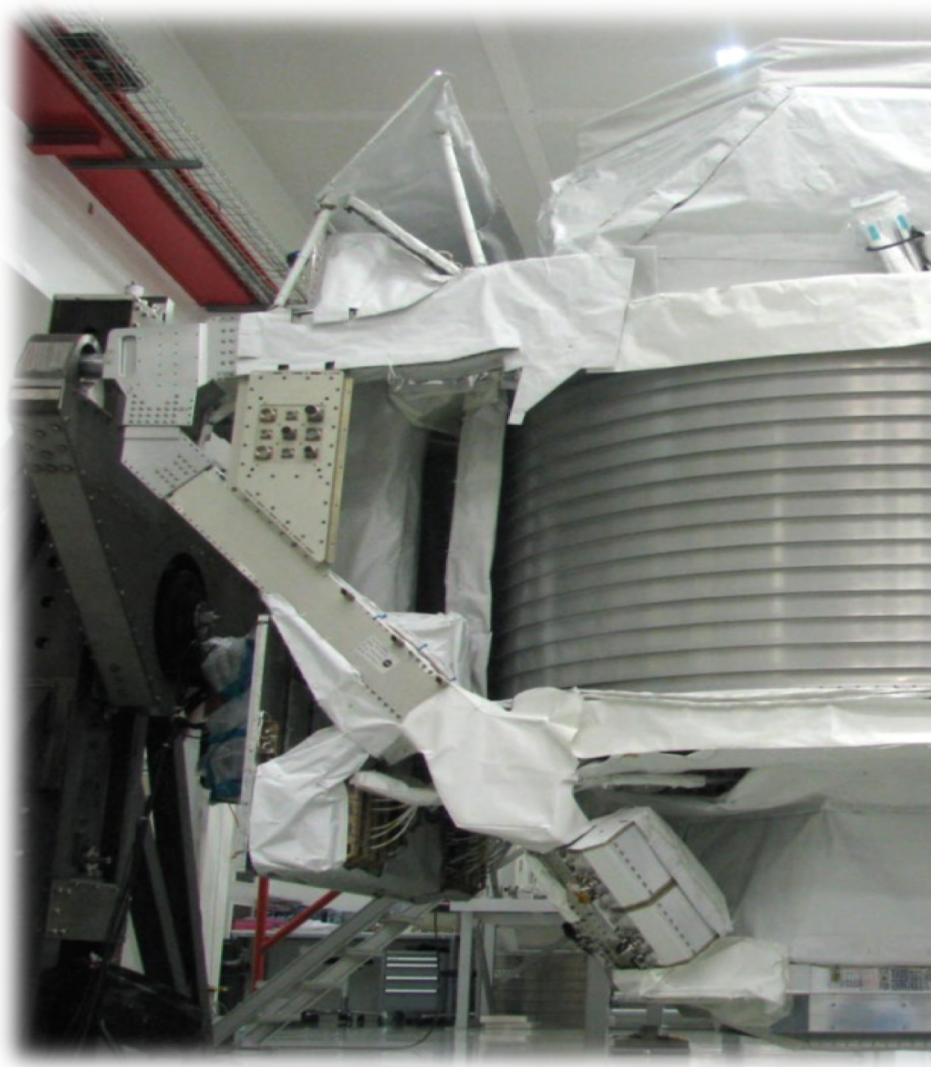
Current operating experiments





Latest AMS results

2010: AMS-02 assembled



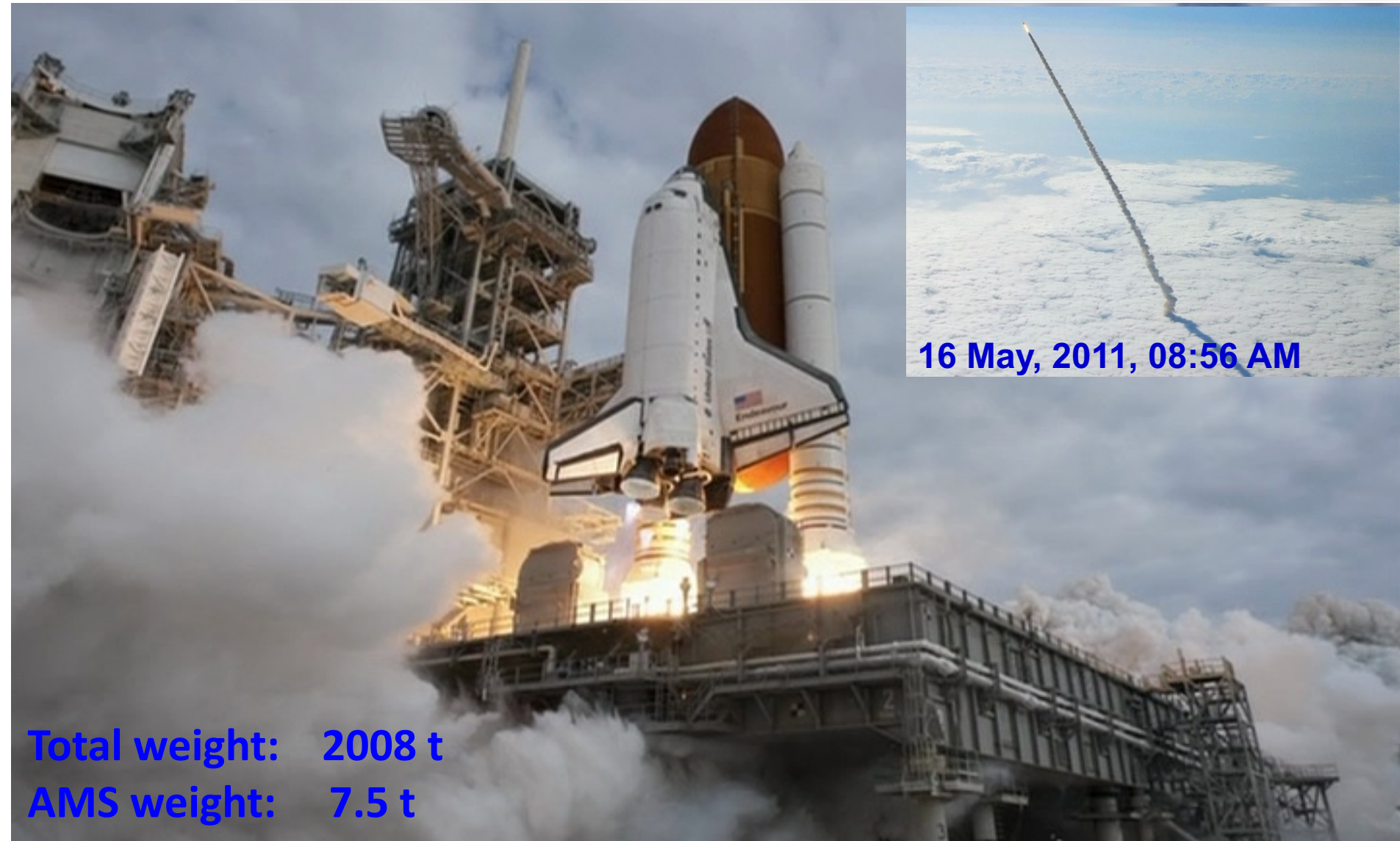
AMS facts:

- 5 m x 4 m x 3m
- 7.5 tons
- 300k read-out channels
- more than 600 microprocessors reduce the *rate* from 7 Gb/s to 10 Mb/s
- total power < 2.5 kW

Houston, JSC – 16 May, 2011@ 07:56 AM



2011: AMS launch - @ KSC, Florida



A particle physics experiment on ISS

AMS-02 time on ISS since May 19th, 5:46 a.m. EDT:

3203

DAYS

23

HOURS

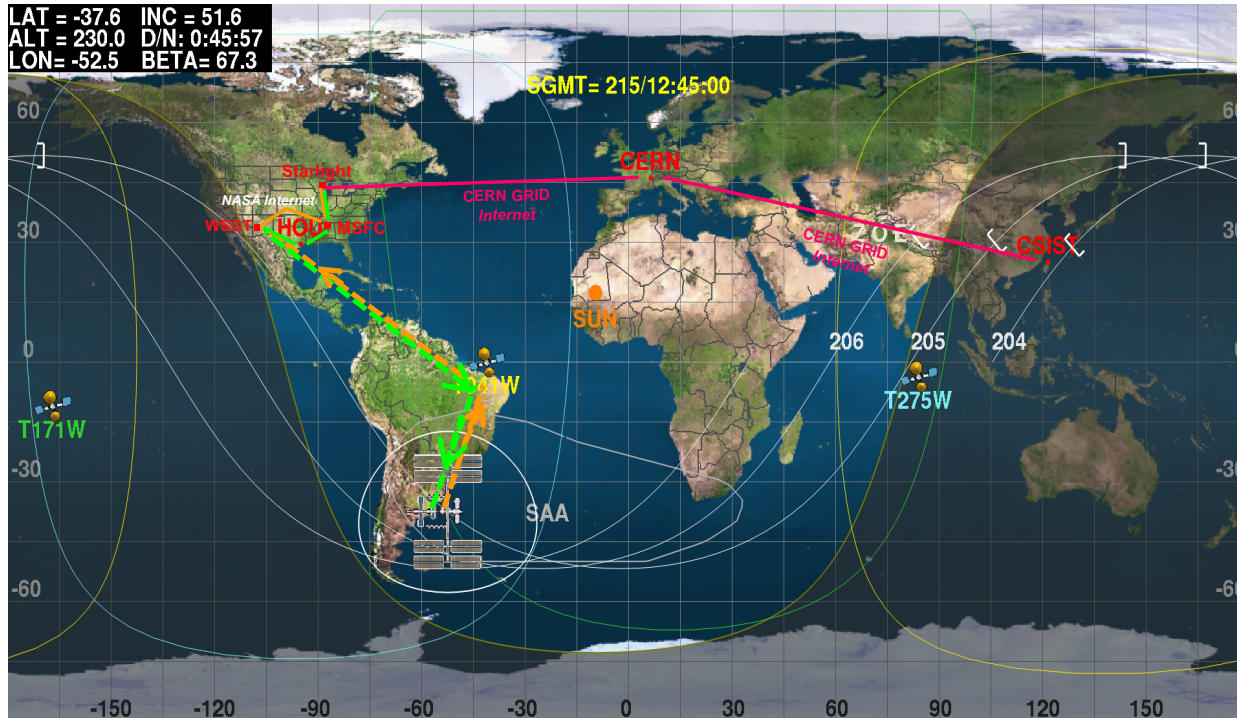
54

MINUTES

14

SECONDS

LAT = -37.6 INC = 51.6
ALT = 230.0 D/N: 0:45:57
LON = -52.5 BETA = 67.3



AMS has collected

153,733,562,077

cosmic ray events

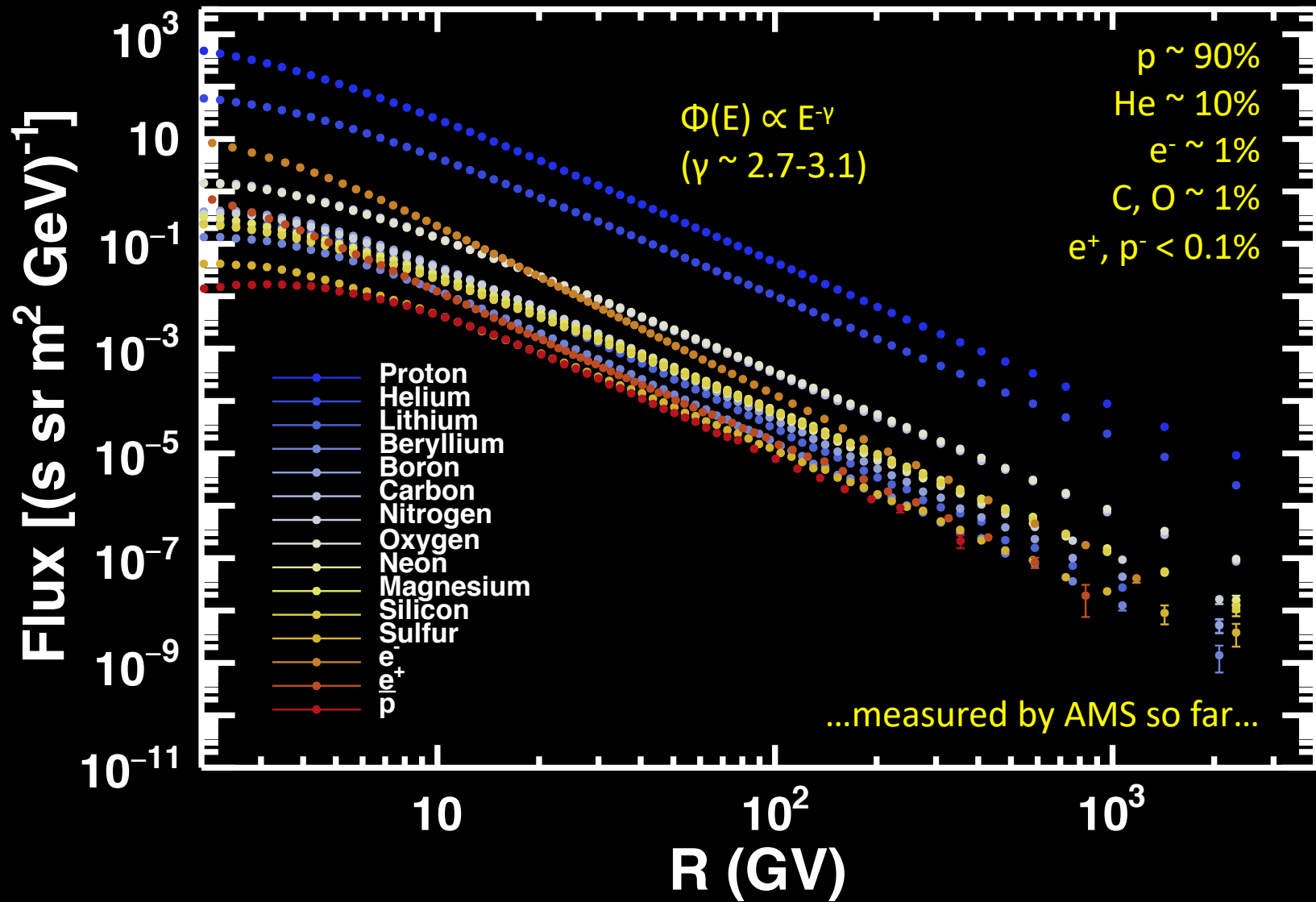
Last update: February 25, 2020, 8:48 AM



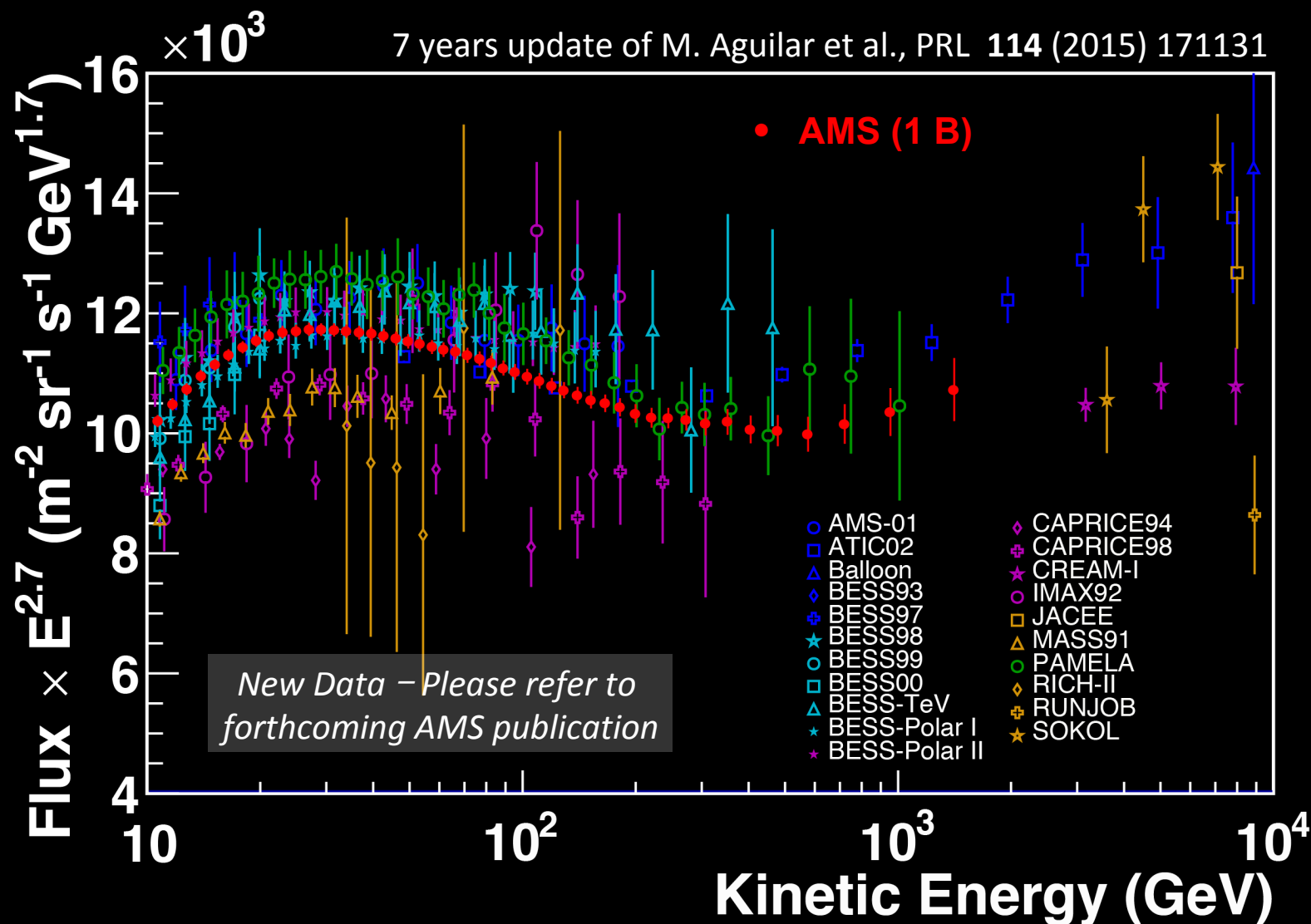
**Payload Operation
Control Center @CERN**



Goal: to measure all the CR fluxes

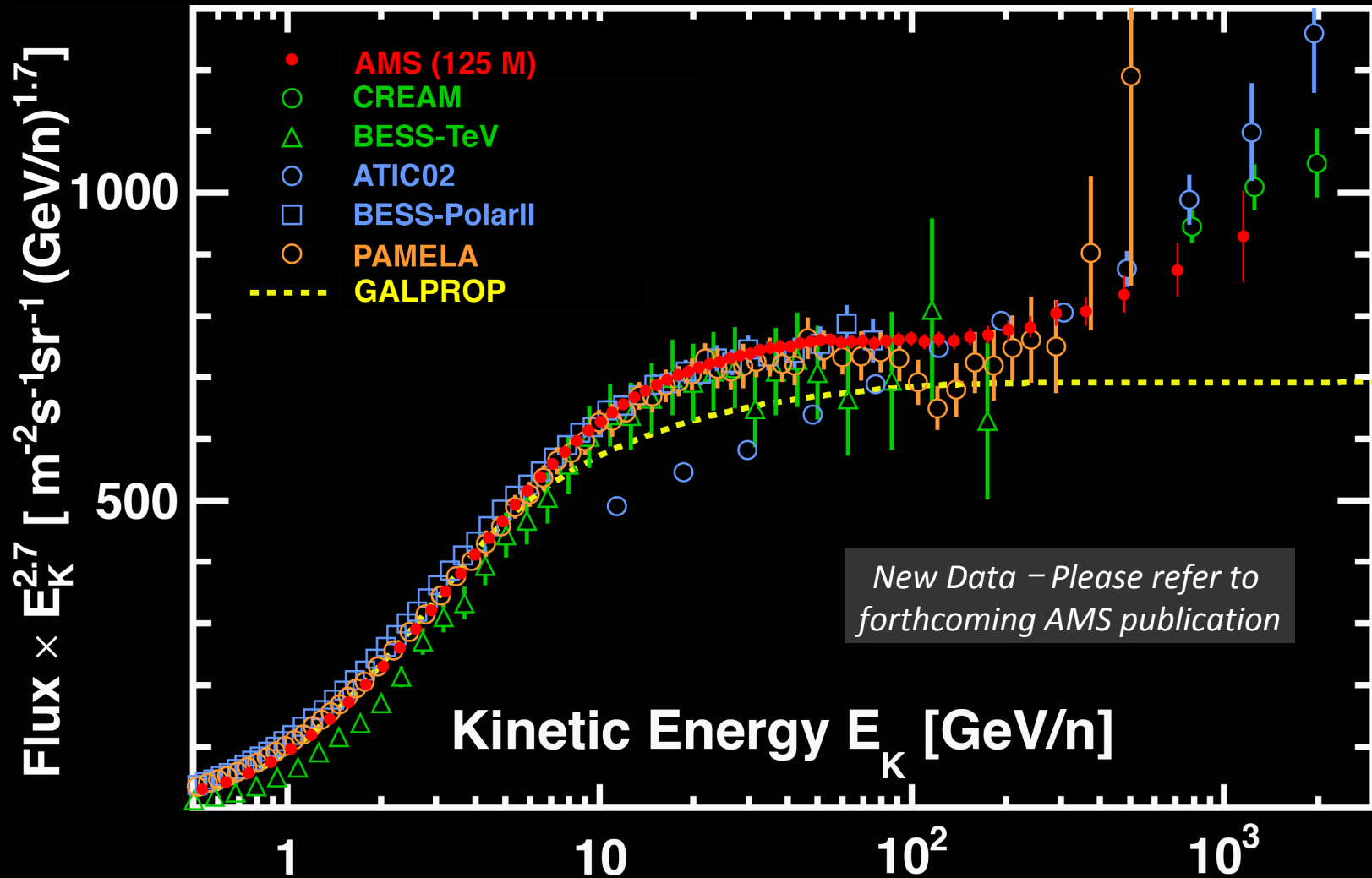


Proton Flux



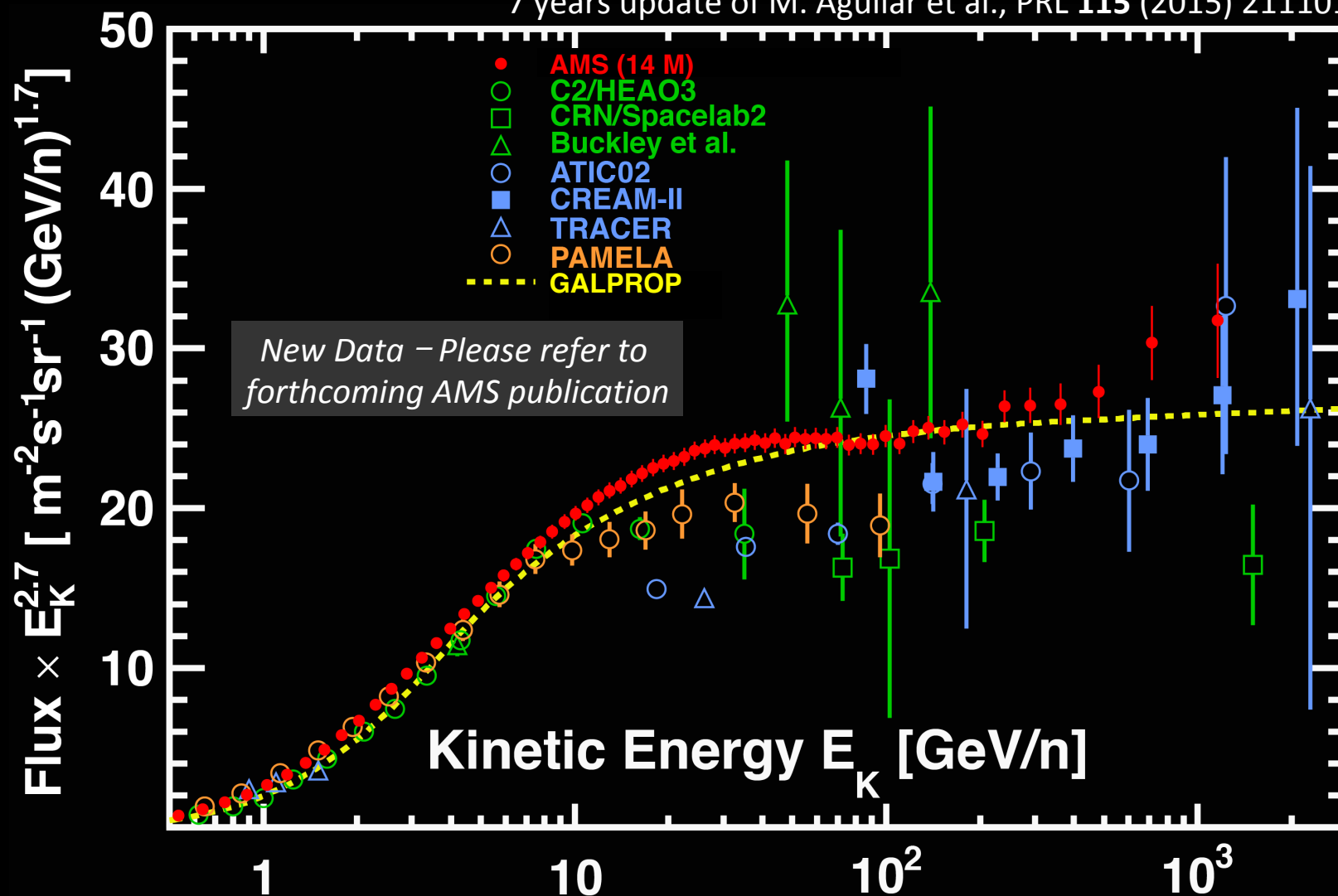
Helium Flux

7 years update of M. Aguilar et al., PRL **115** (2015) 211101



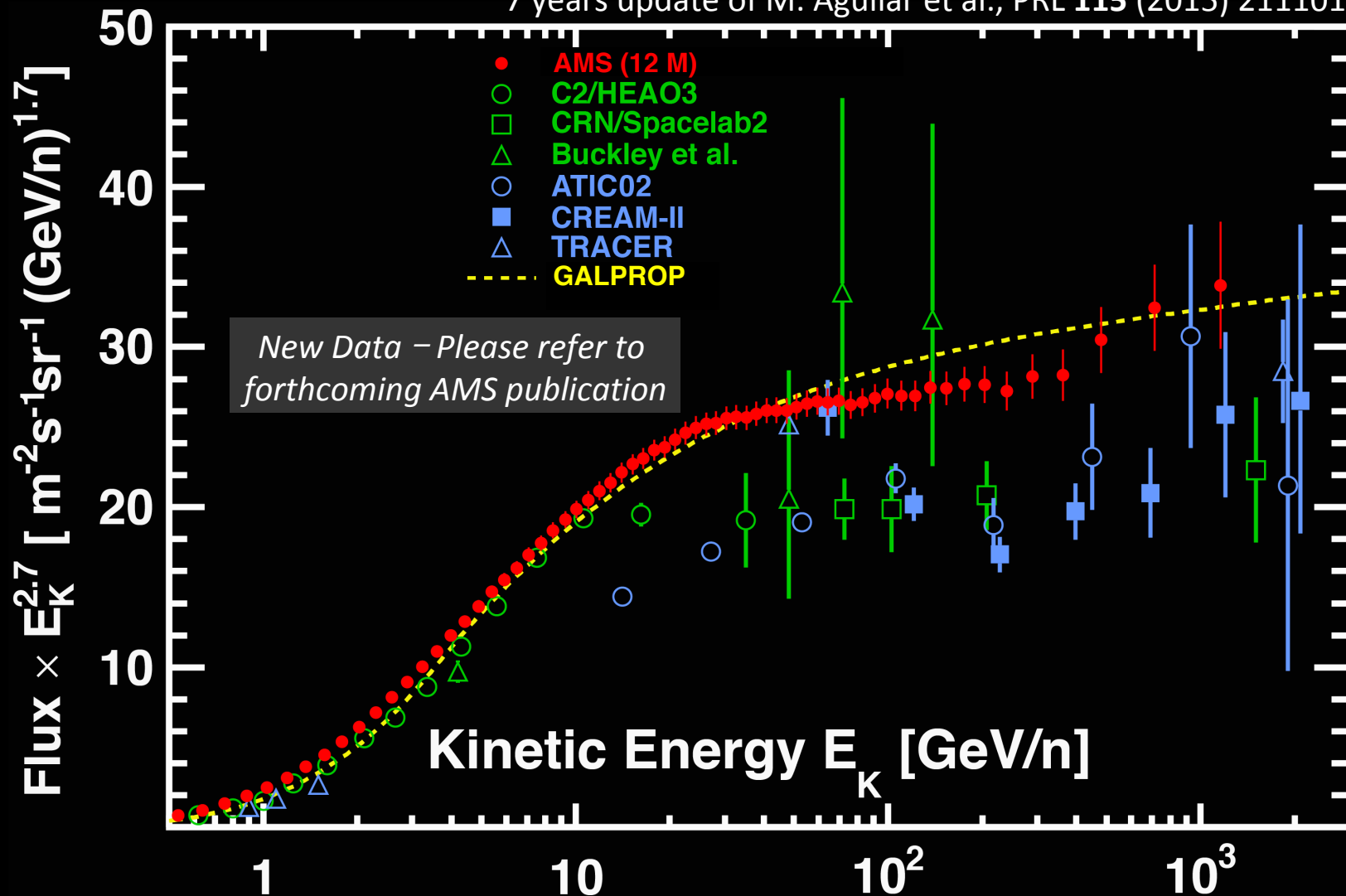
Carbon Flux

7 years update of M. Aguilar et al., PRL **115** (2015) 211101



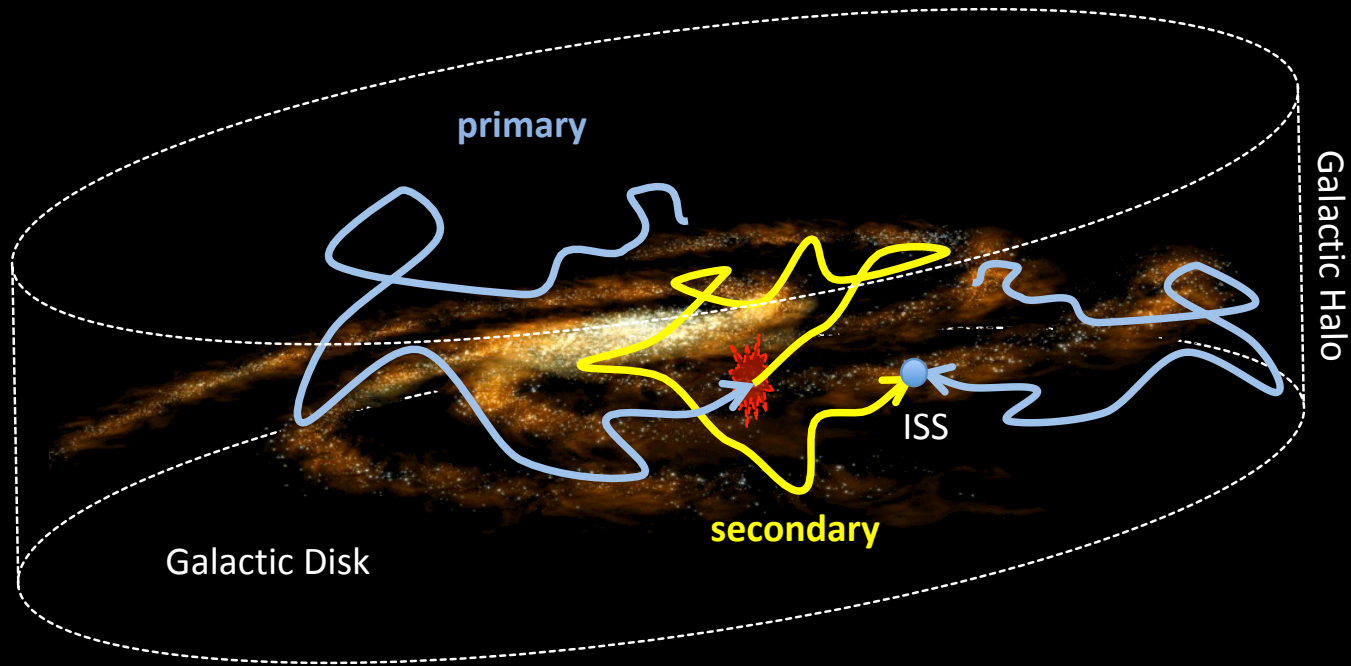
Oxygen Flux

7 years update of M. Aguilar et al., PRL **115** (2015) 211101



Secondary Cosmic Rays

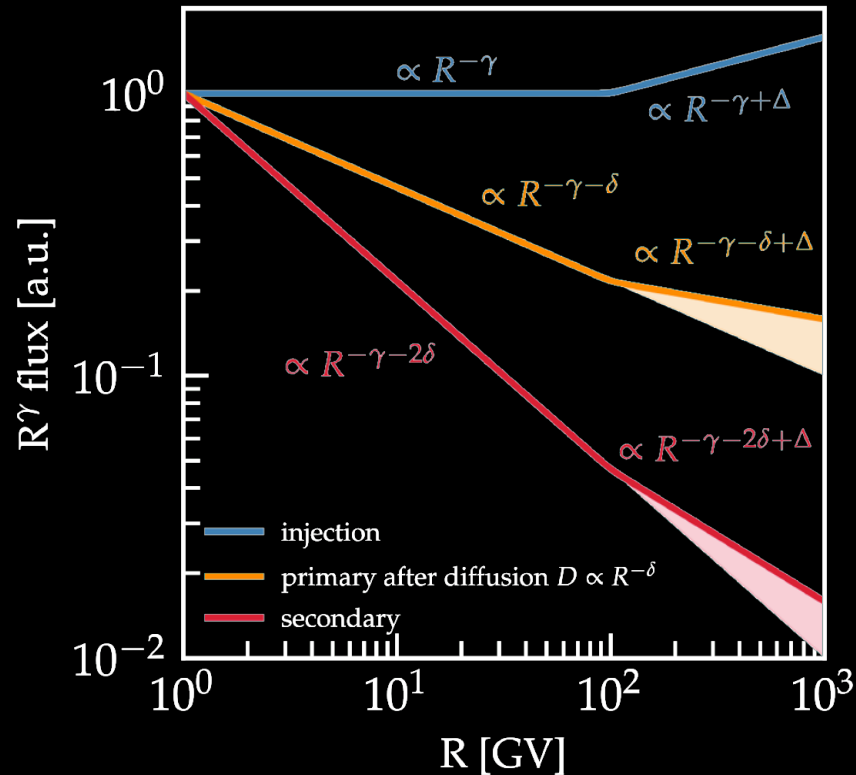
Cosmic rays **primaries** are mostly produced at astrophysical sources (ex. e^- , p , He , C , O , ...), **secondaries** (ex. Li , Be , B , ...) are mostly produced by the collision of cosmic rays with the ISM.



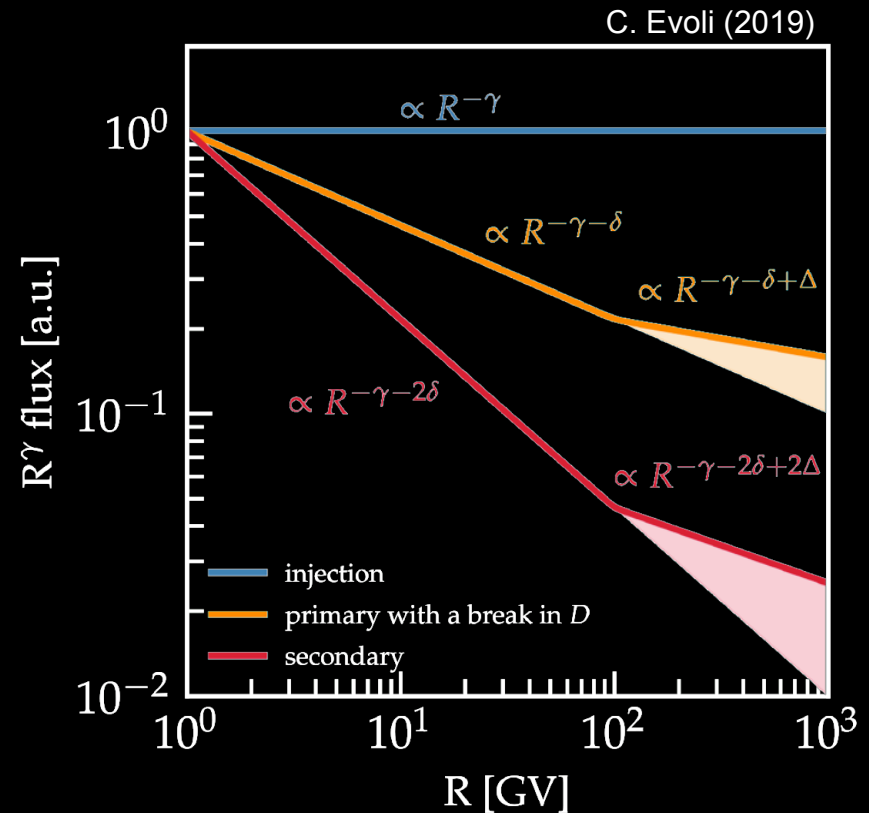
The understanding of primary and secondary cosmic rays nuclei reveal details of sources and propagation of all CRs species, specially for the secondary production of e^+ , \bar{p} , \bar{D} , ...

- The cosmic ray fluxes of their “parents” (p , He)
- Behaviour of their propagation in the Milky Way (B/C , Be/B , ...)

Cosmic Ray Propagation

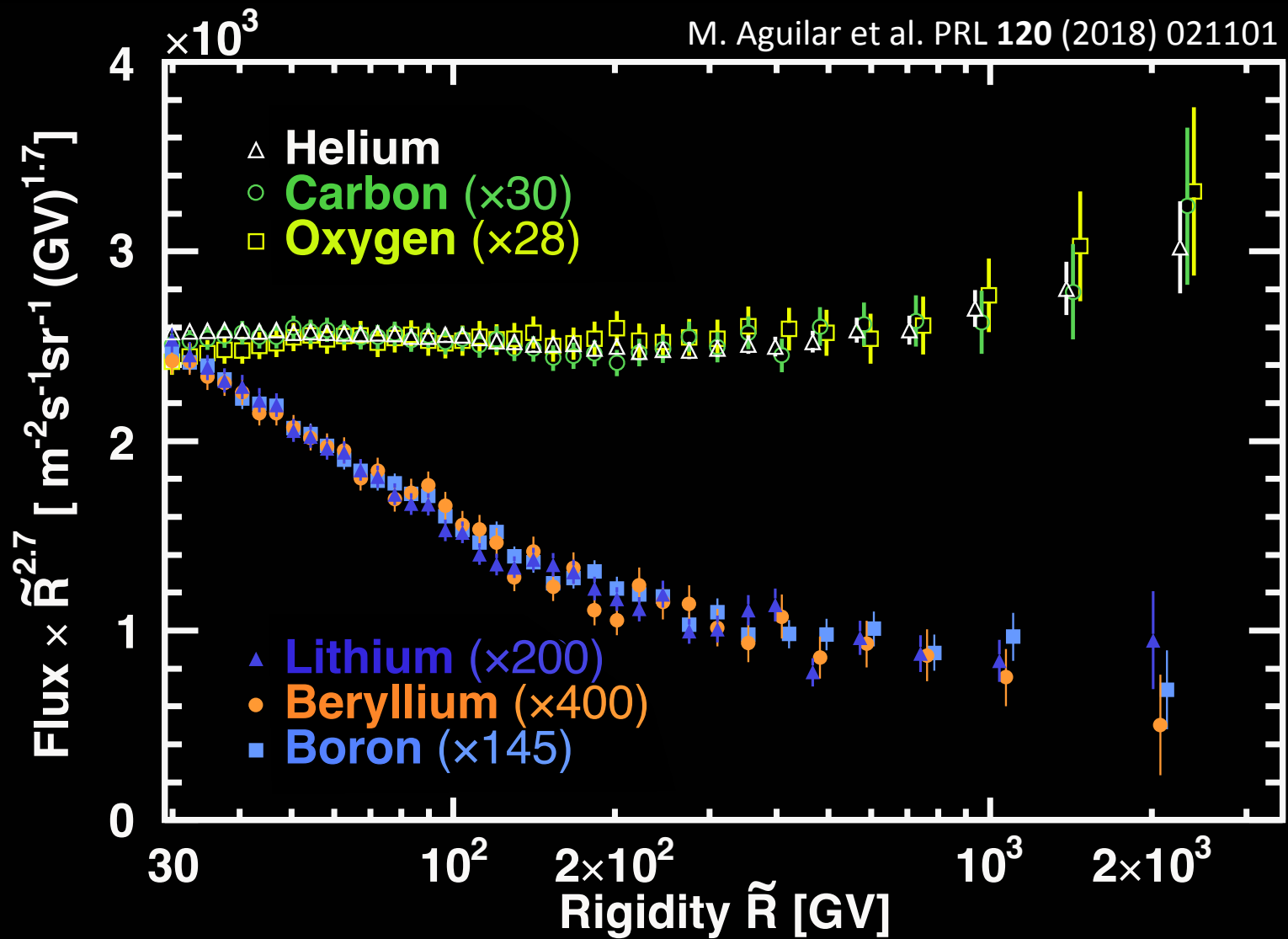


If the hardening in CRs is related to the **injected spectra** at their source, then **similar hardening** is expected both for **secondary** and **primary** cosmic rays.

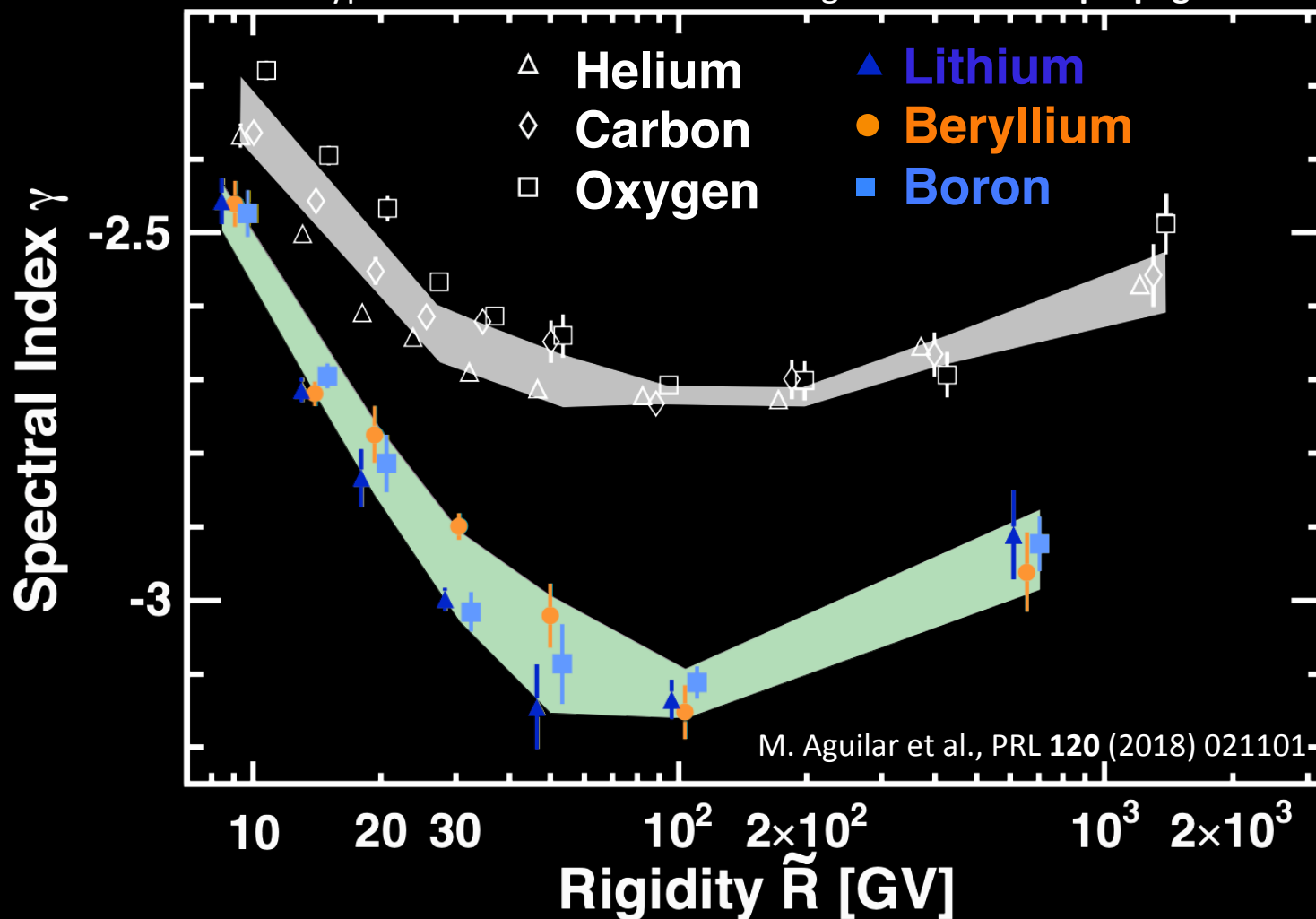


If the hardening is related to **propagation properties** in the Galaxy then a **stronger hardening** is expected for the **secondary** with respect to the **primary** cosmic rays.

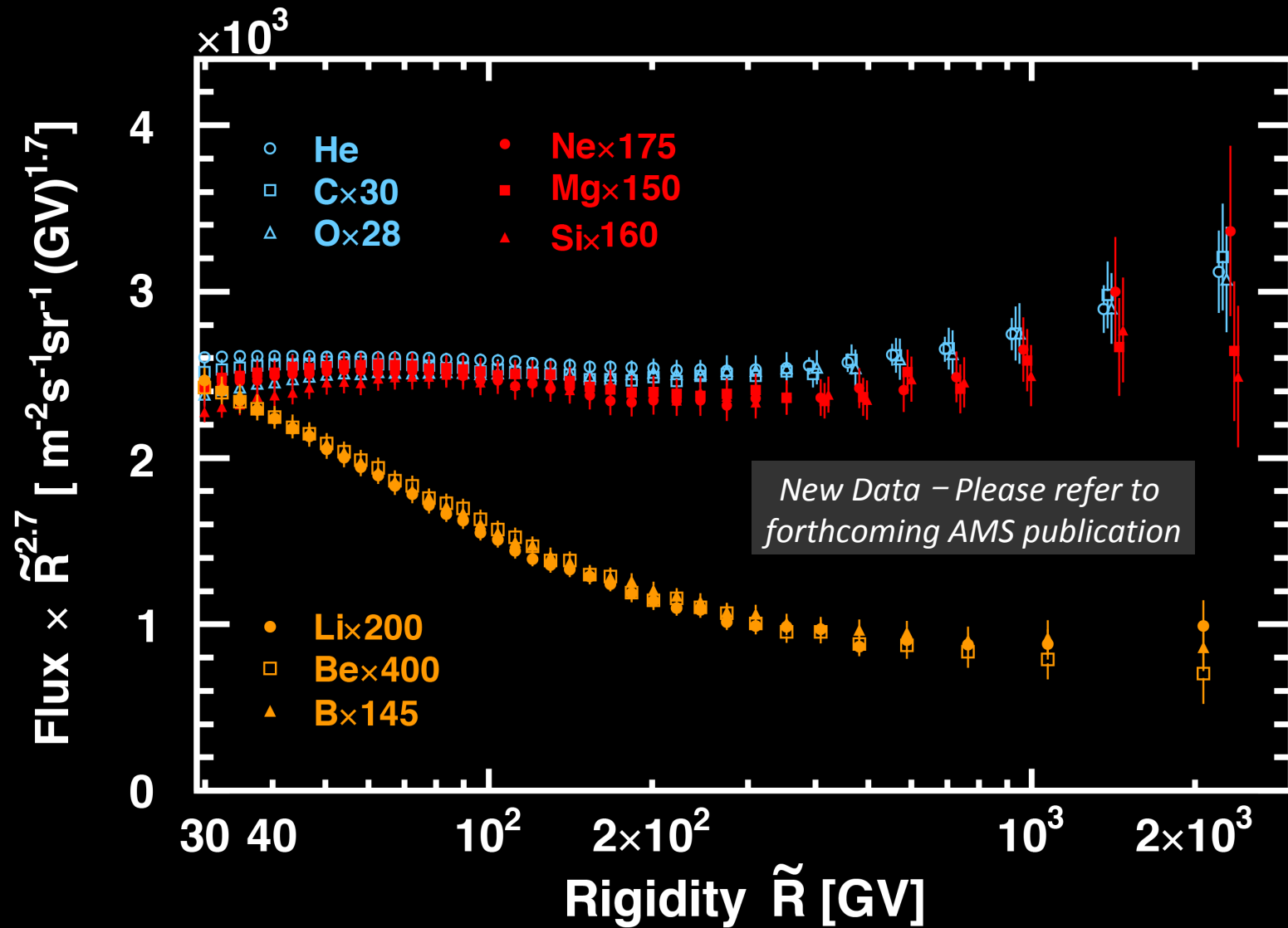
Primary and Secondary Nuclei Fluxes



Deviate from single power law above 200 GV. Secondary hardening is stronger
AMS favors the hypothesis that the flux hardening is an **universal propagation effect**.

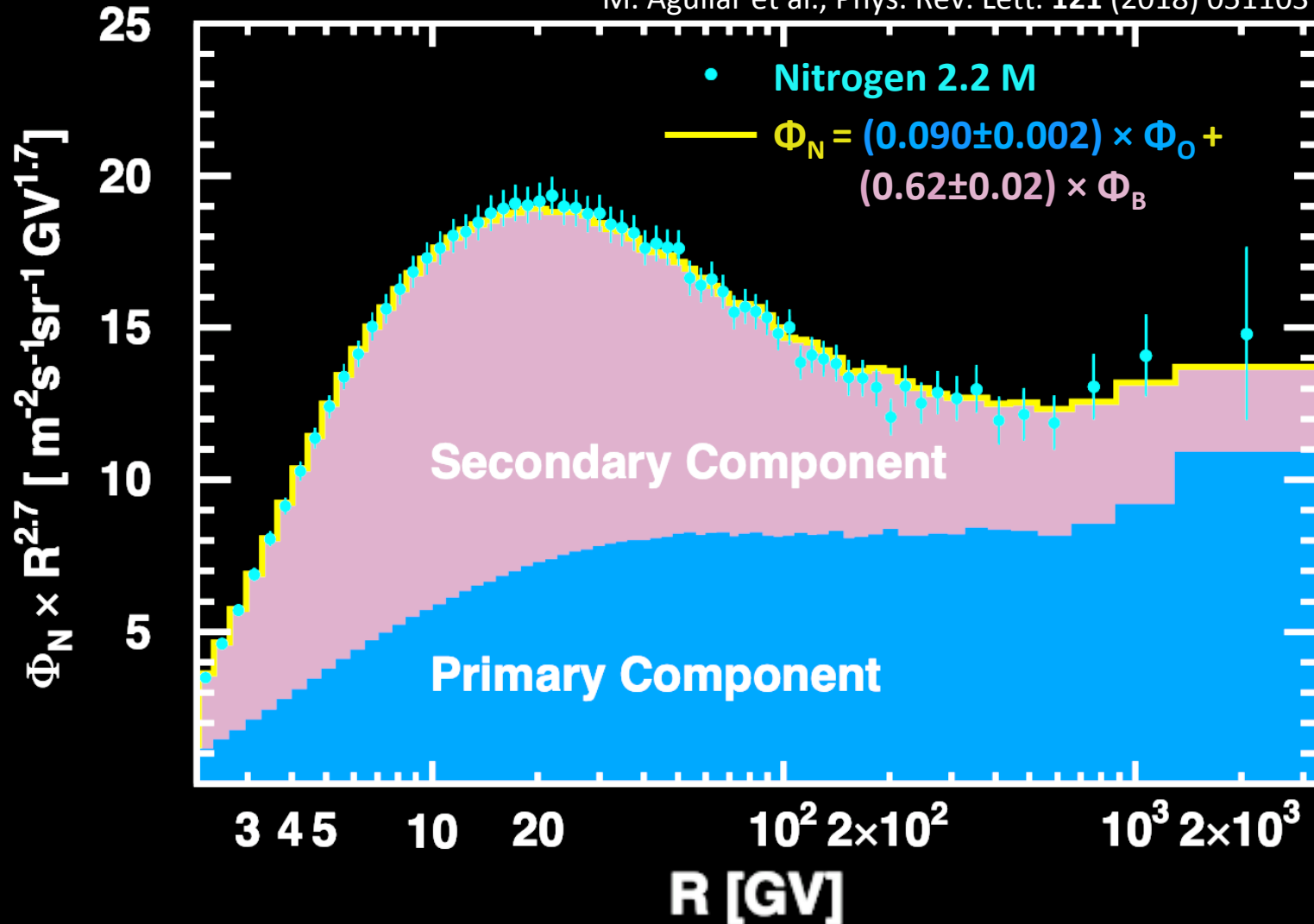


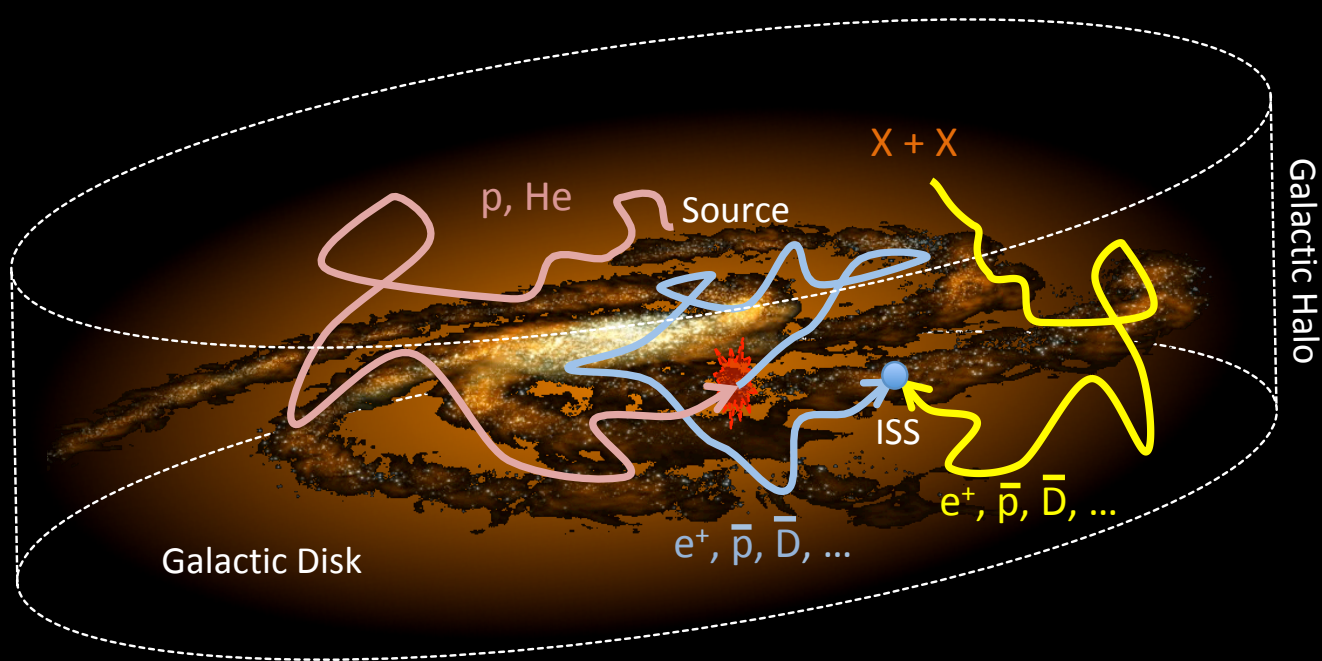
Extending Nuclei Fluxes Towards High-Z



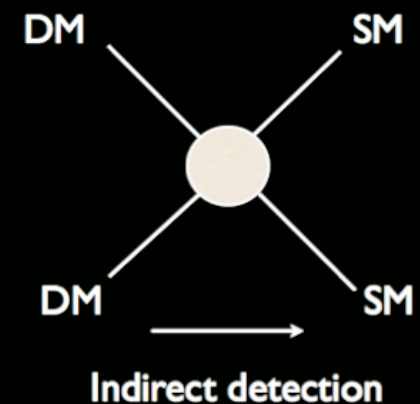
Nitrogen Flux

M. Aguilar et al., Phys. Rev. Lett. **121** (2018) 051103



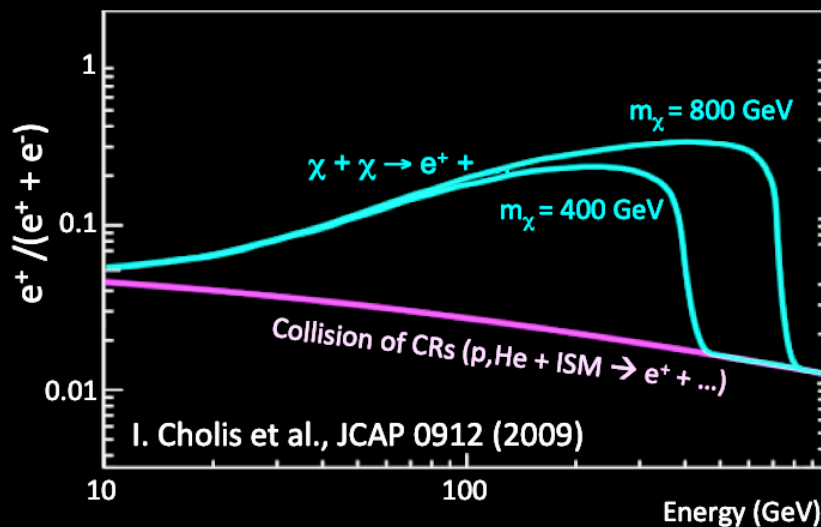


Collisions of dark matter particles (ex. neutralinos) may produce a signal of e^+ , \bar{p} , \bar{D} , ... that can be detected above the background from the collisions of primary CRs on interstellar medium

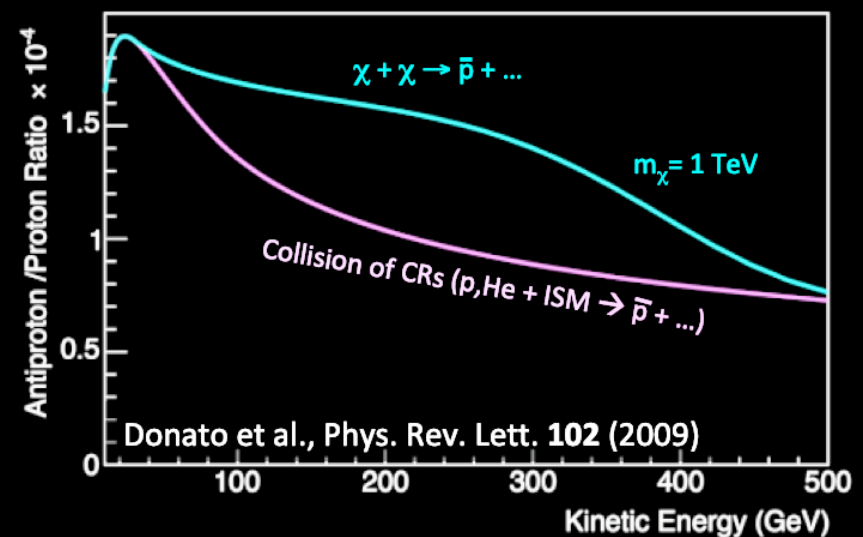


Collisions of Dark Matter particles (ex. neutralinos) may produce a signal of e^+ , \bar{p} , \bar{D} ... detected above the background from the collisions of CRs on interstellar medium (ISM)

positron fraction



\bar{p}/p flux ratio

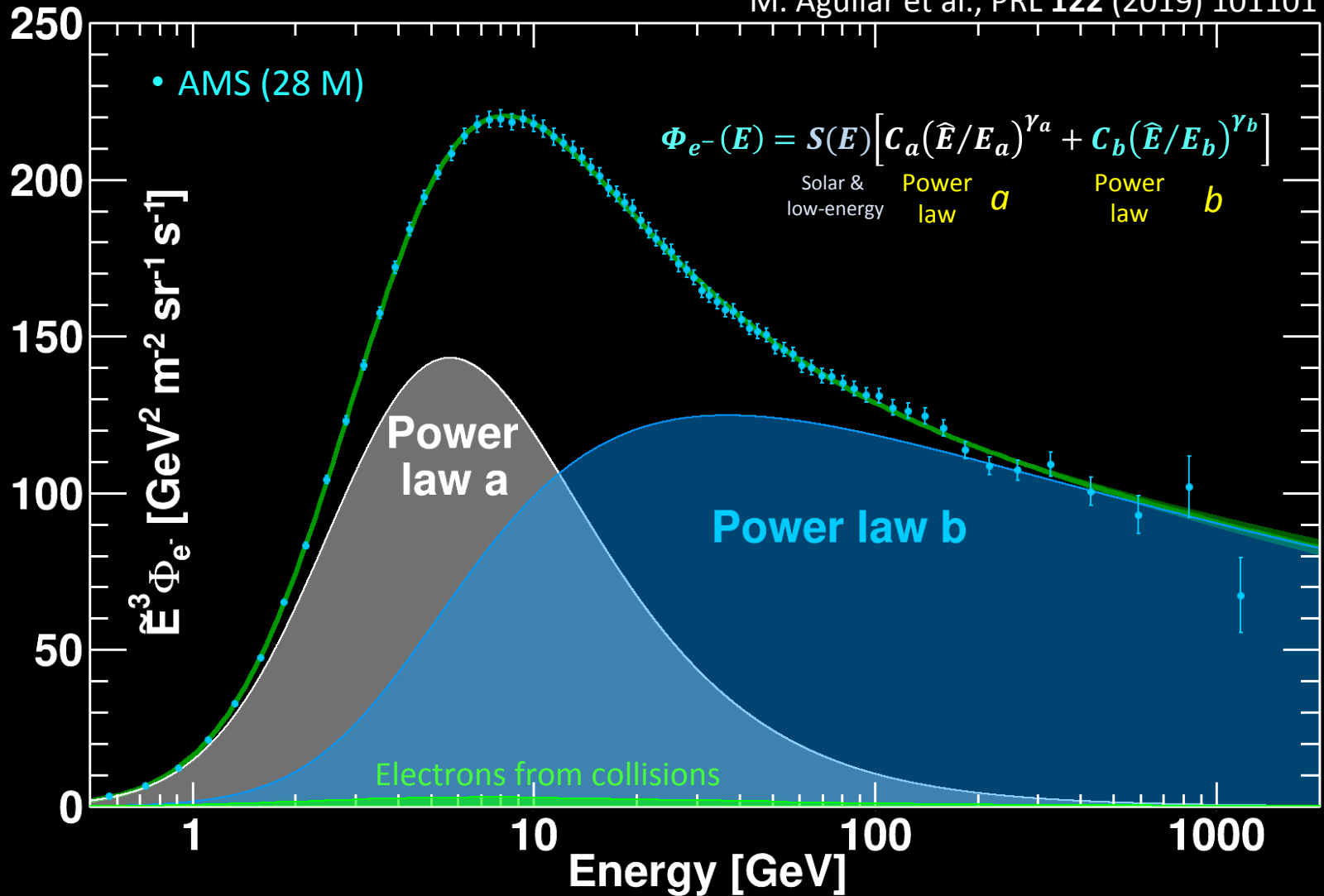


To calculate the secondary production of e^+ and \bar{p} we need

- The cosmic ray fluxes of their "parents" (p, He)
- Production cross-section ($p \rightarrow p + p + p + \dots$)
- Behaviour of their propagation in the Milky Way ($B/C, Be/B, \dots$)

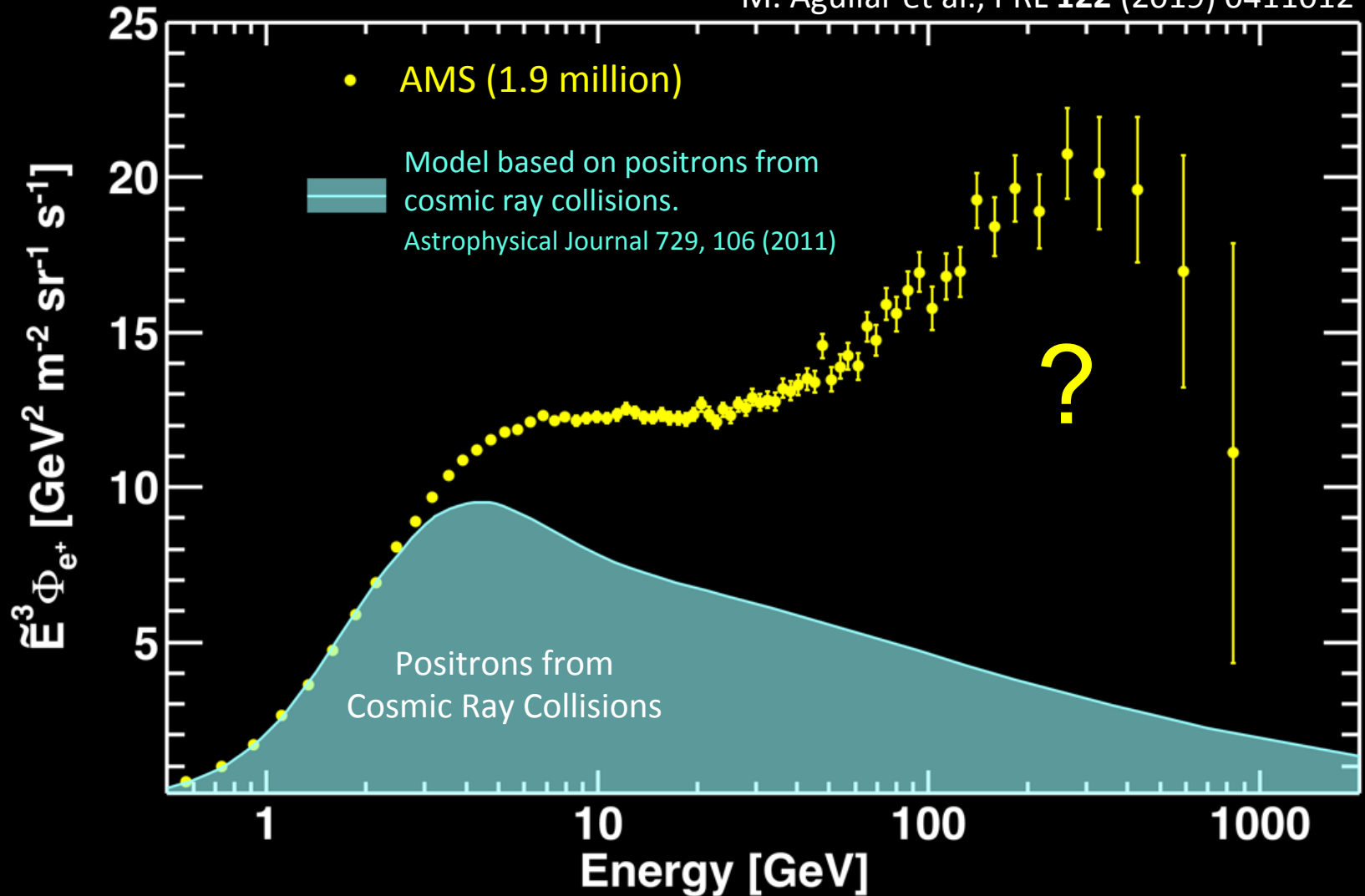
Electron Flux

M. Aguilar et al., PRL **122** (2019) 101101

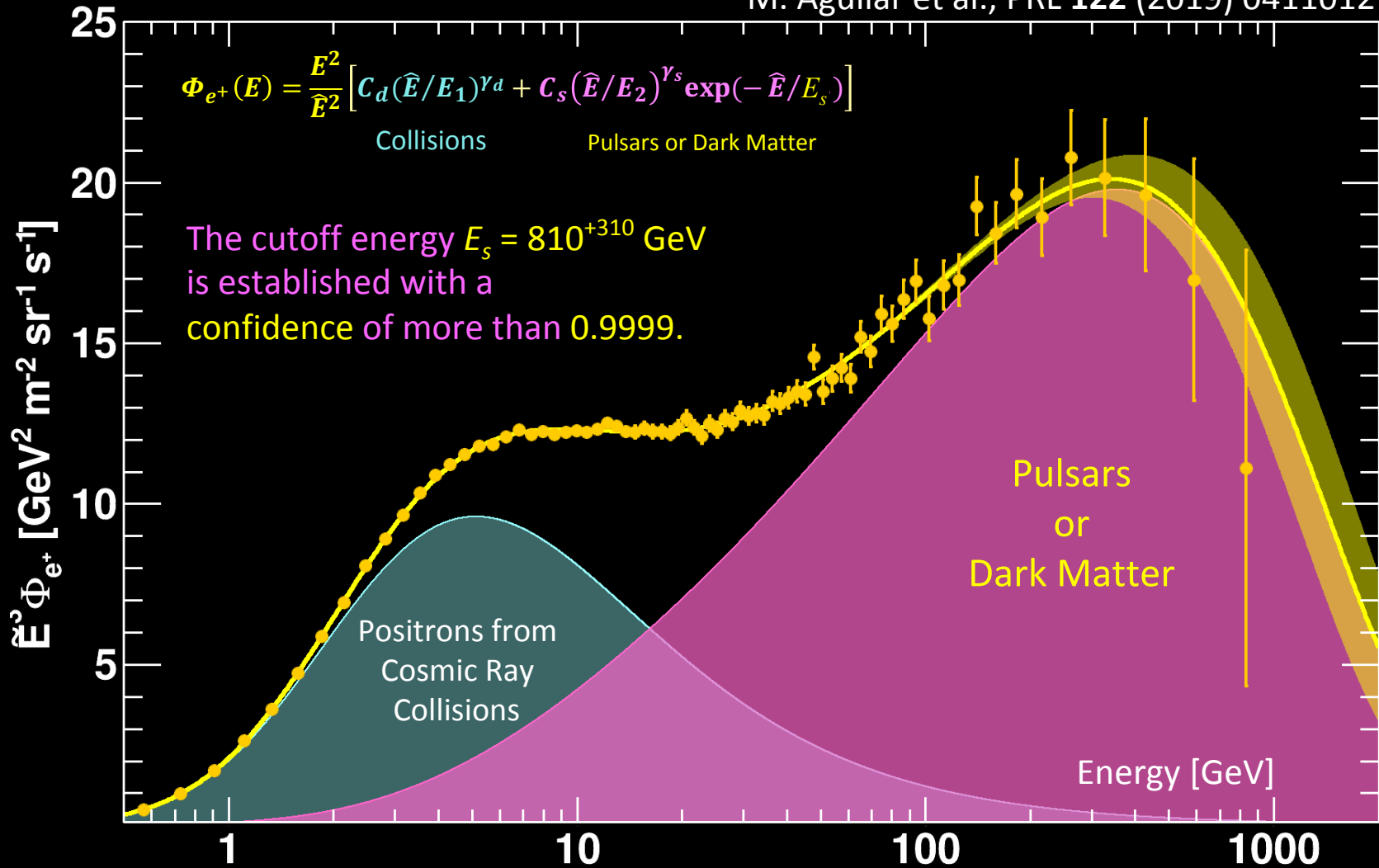


Positron Flux

M. Aguilar et al., PRL **122** (2019) 0411012

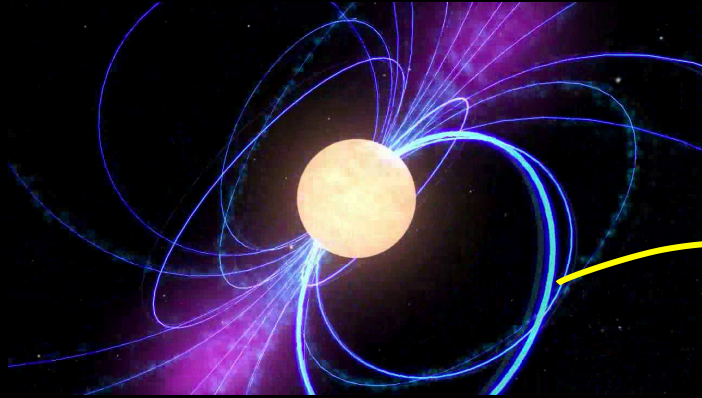


M. Aguilar et al., PRL **122** (2019) 0411012

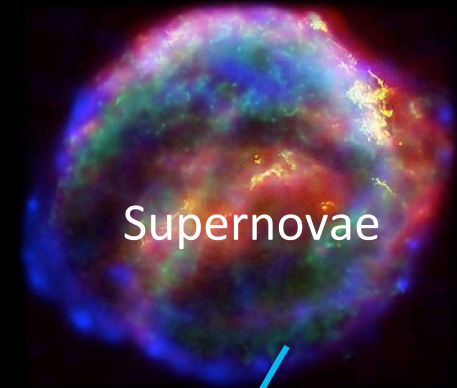


Origin of Positrons

New Astrophysical Sources: Pulsars, ...



Positrons
from Pulsars



Supernovae

Protons,
Helium, ...

Interstellar
Medium

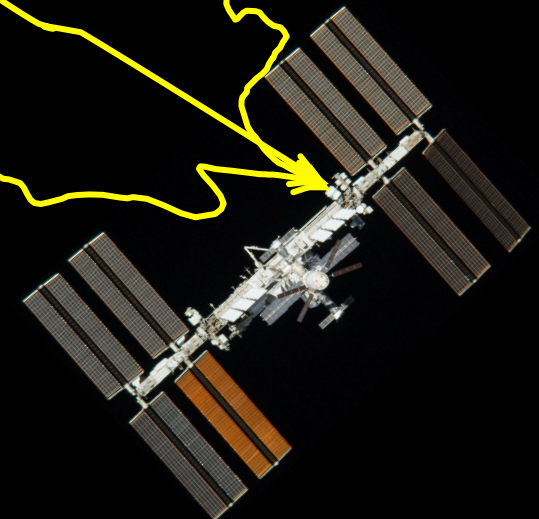
Positrons
from Collisions

Dark Matter

Positrons
from Dark Matter

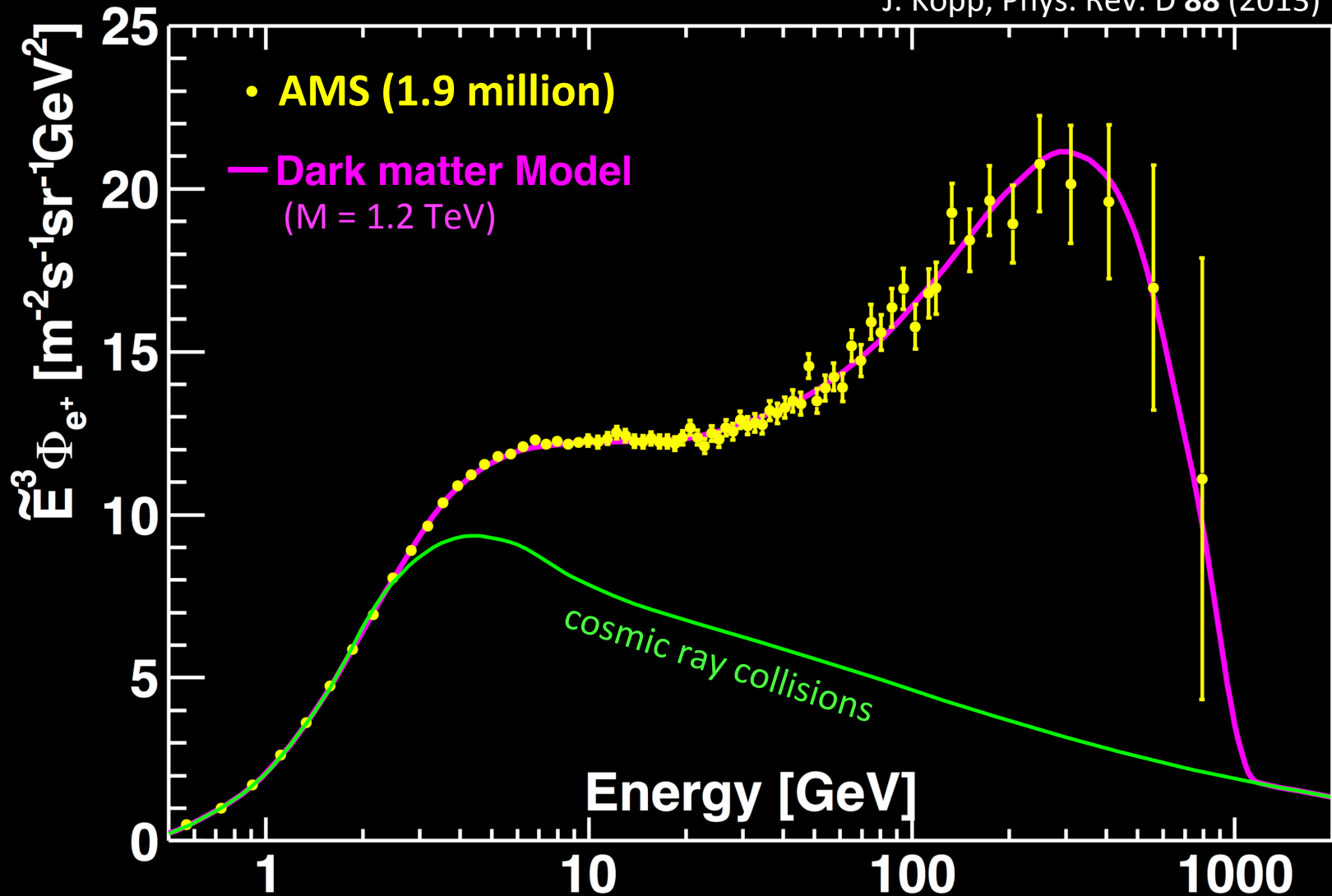
Electrons

Dark Matter



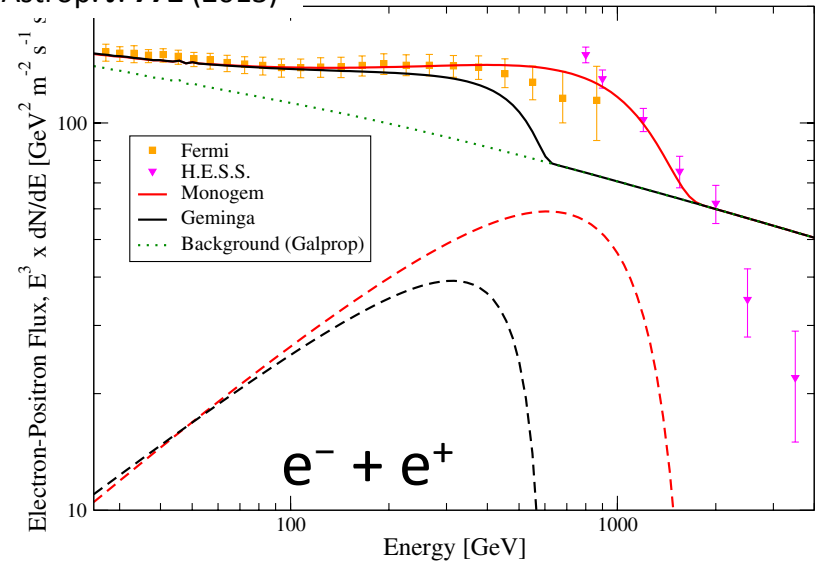
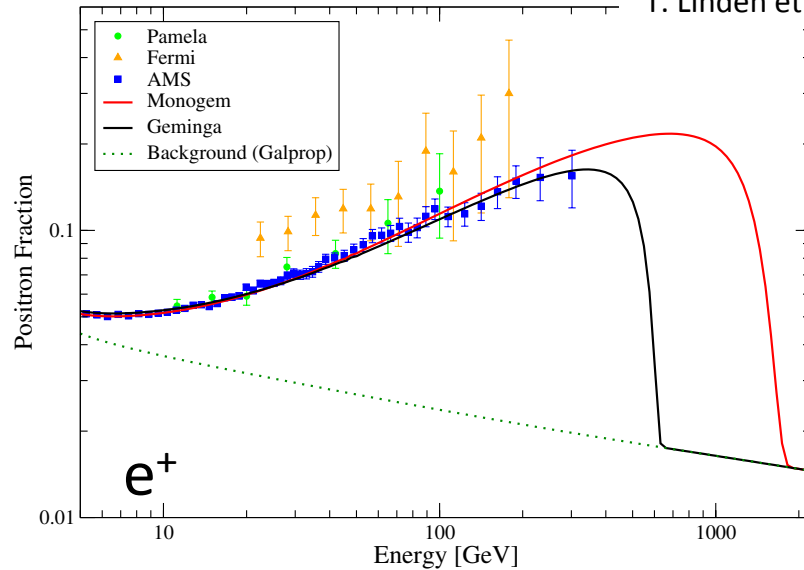
Positrons & Dark Matter

J. Kopp, Phys. Rev. D **88** (2013)

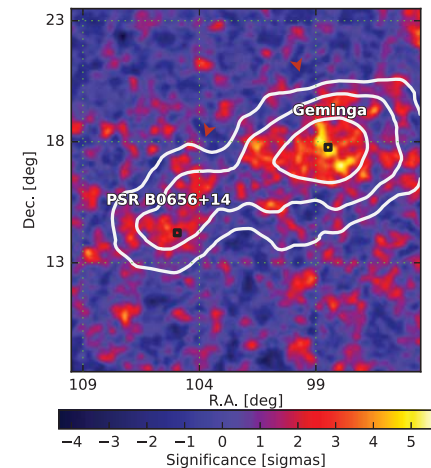
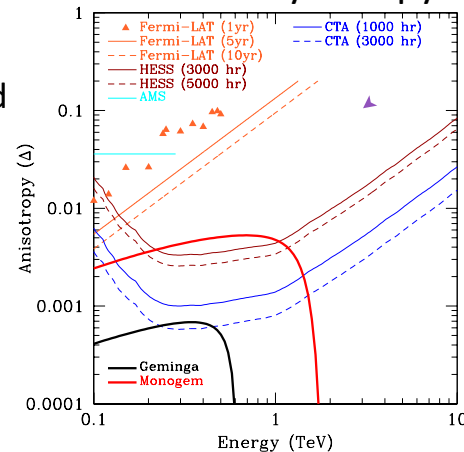


Positrons from Pulsar

T. Linden et al., *Astrop. J.* **772** (2013)



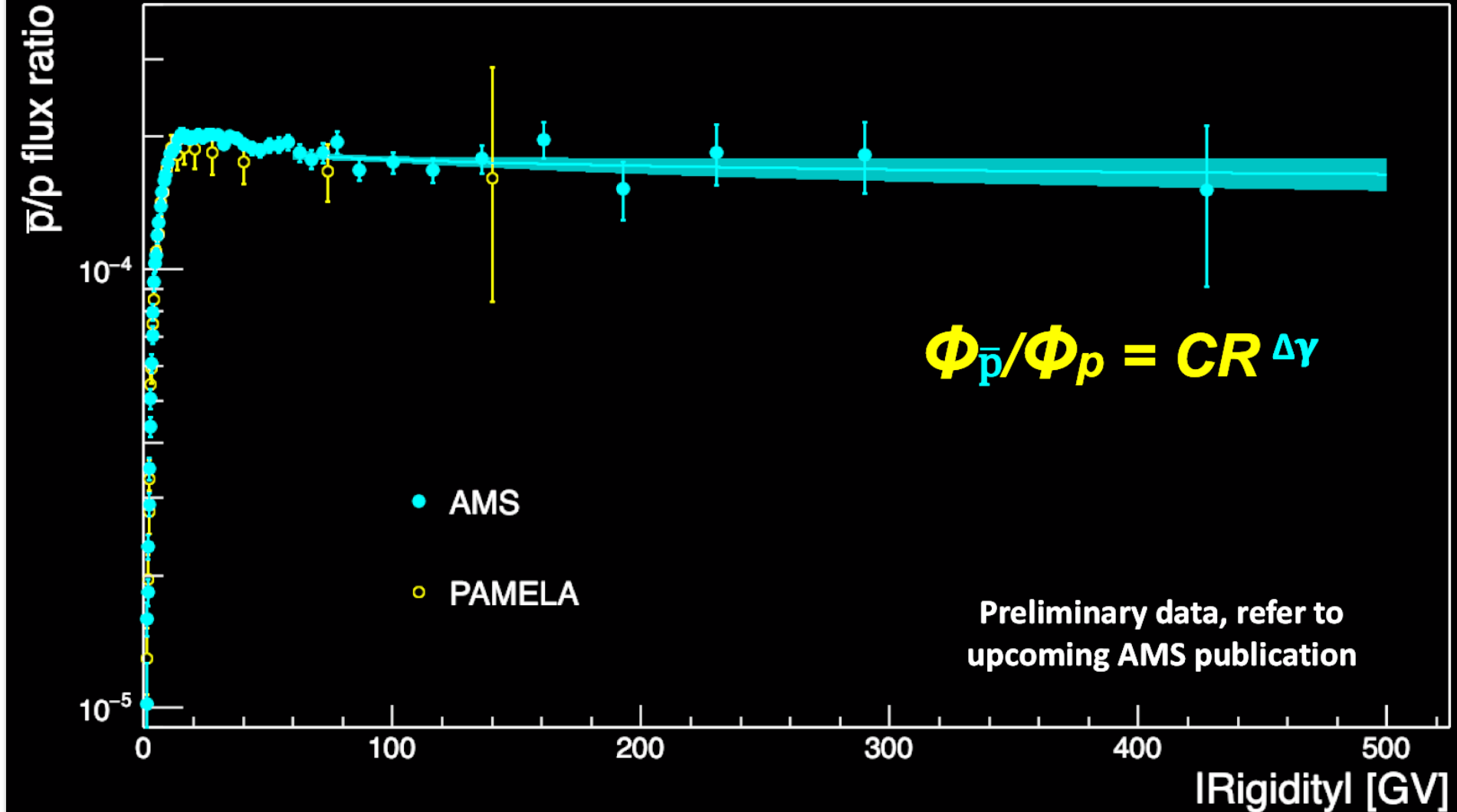
$e^- + e^+$ anisotropy



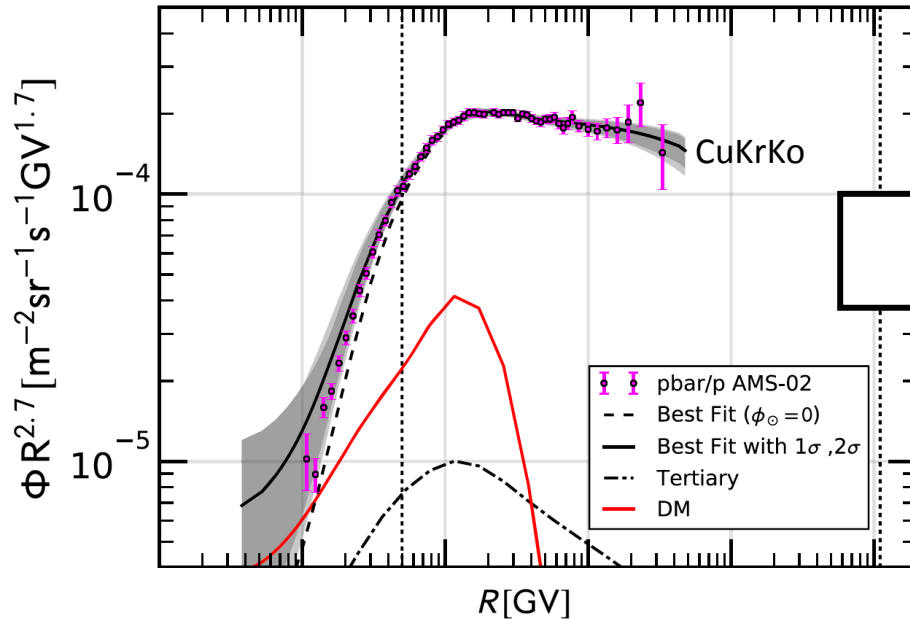
Pulsars spinning produce EM radiation and cosmic rays (pair production).
To distinguish from DM models:
→ **spectral features** of e^+ and of $(e^+ + e^-)$
→ **anisotropy** of e^+ and of $(e^+ + e^-)$
→ **no anti-proton** production

Anti-Proton / Proton Ratio

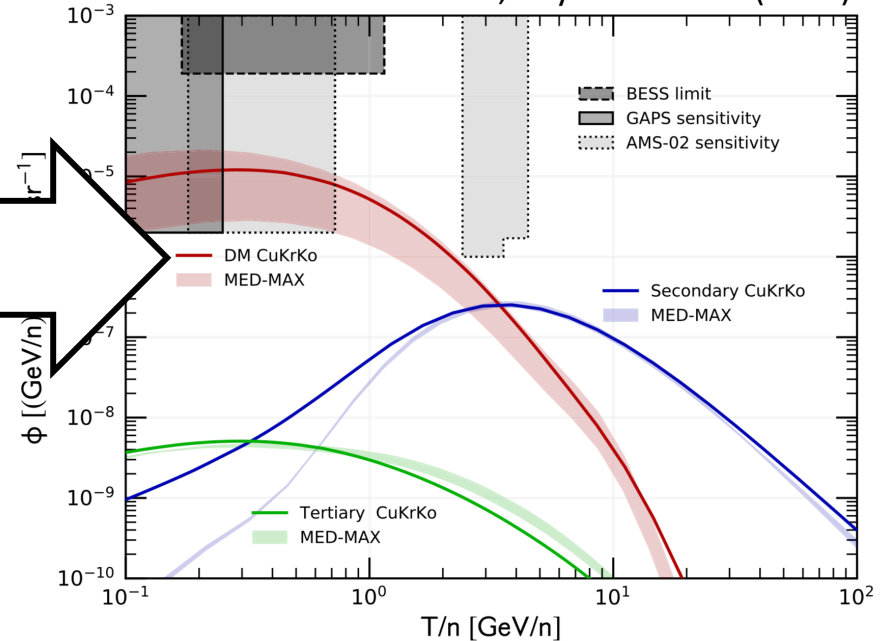
7 years update of M. Aguilar et al., PRL 117 (2016) 0911003



A. Cuoco et al., Phys. Rev. Lett. **118** (2017)

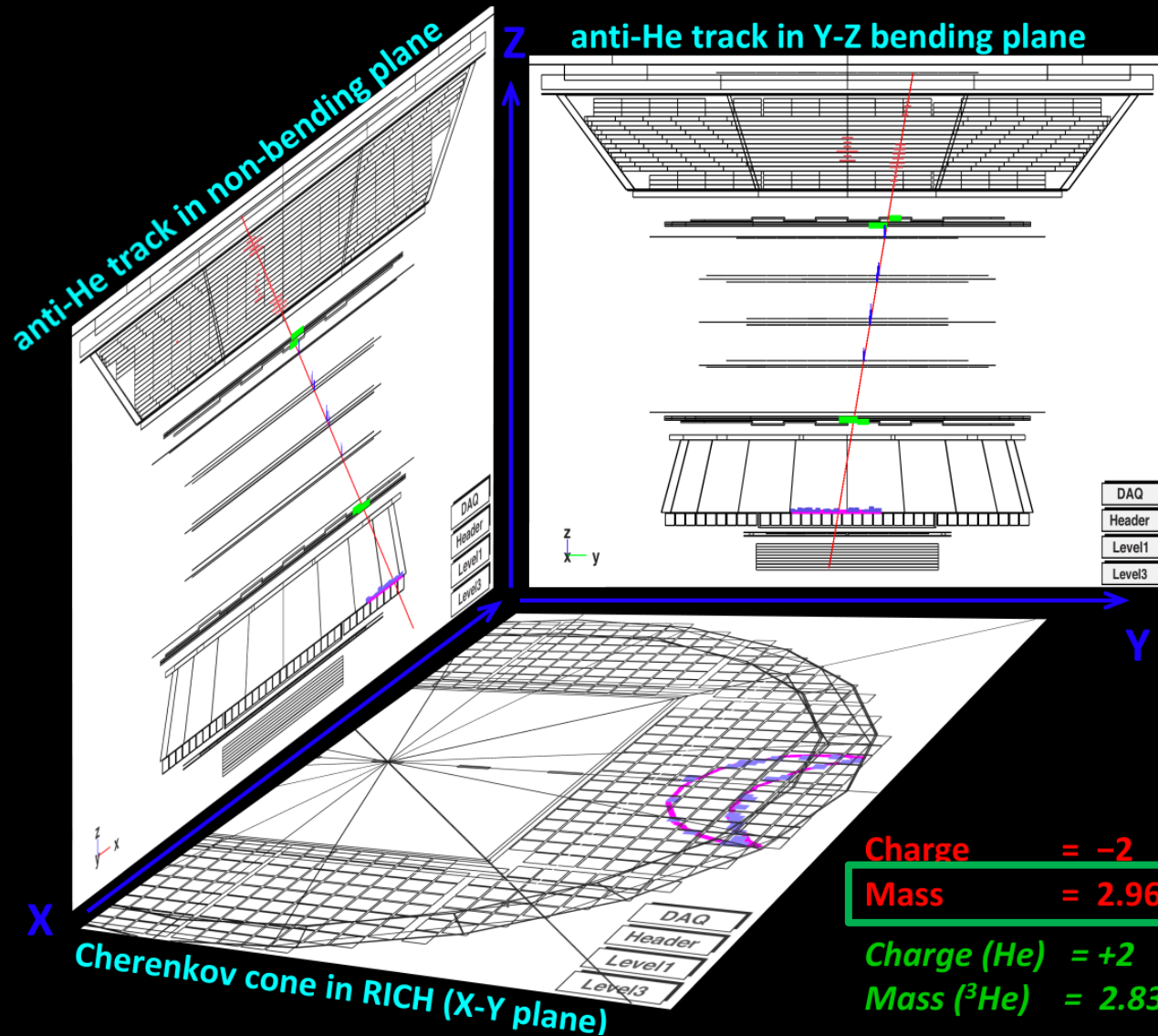


M. Korsmeier et al., Phys. Rev. D **97** (2018)



Several authors reported an allowed **anti-proton** excess at low energy, with different significances, at 10 GV that can be explained a dark matter signal. This signal can give a detectable **anti-deuteron** signal.

Key concepts/detectors



Charge = -2

Mass = $2.96 \pm 0.33 \text{ GeV}/c^2$

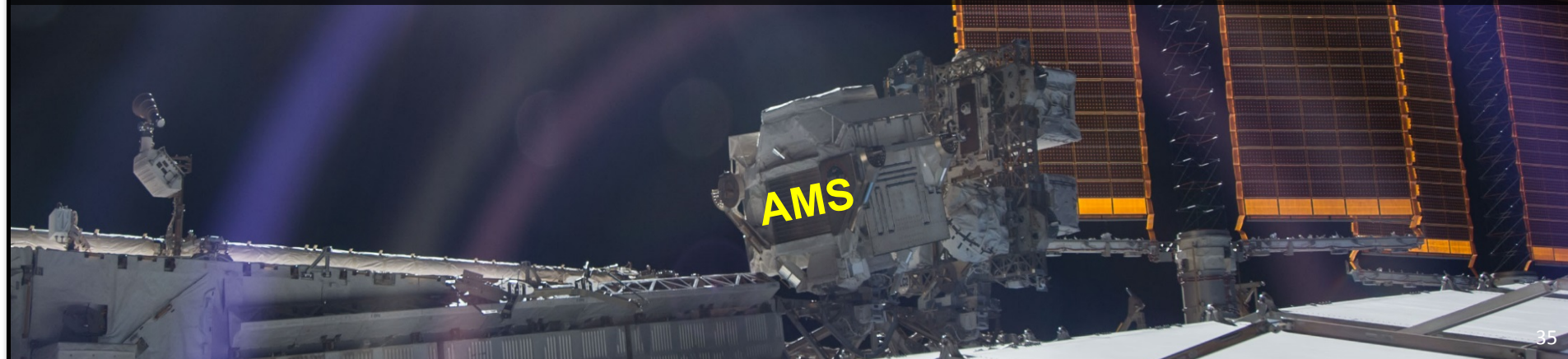
Charge (He) = +2

Mass (^3He) = $2.83 \text{ GeV}/c^2$

Currently, AMS observed 8 anti-helium candidates (mass region from 0-10 GeV/c^2) with rigidity < 50 GV with respect to a sample of 700 million helium events selected.

The rate in AMS of antihelium candidates is less than 1 in 100 million helium.

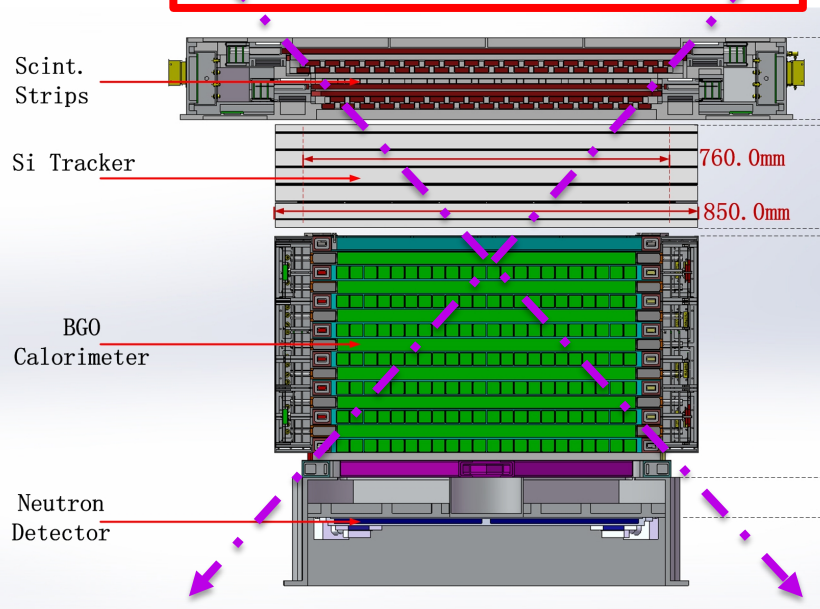
At this extremely low rate, more data (**through the lifetime of the ISS**) is required to further check the origin of these events.



Future/proposed 4π experiments

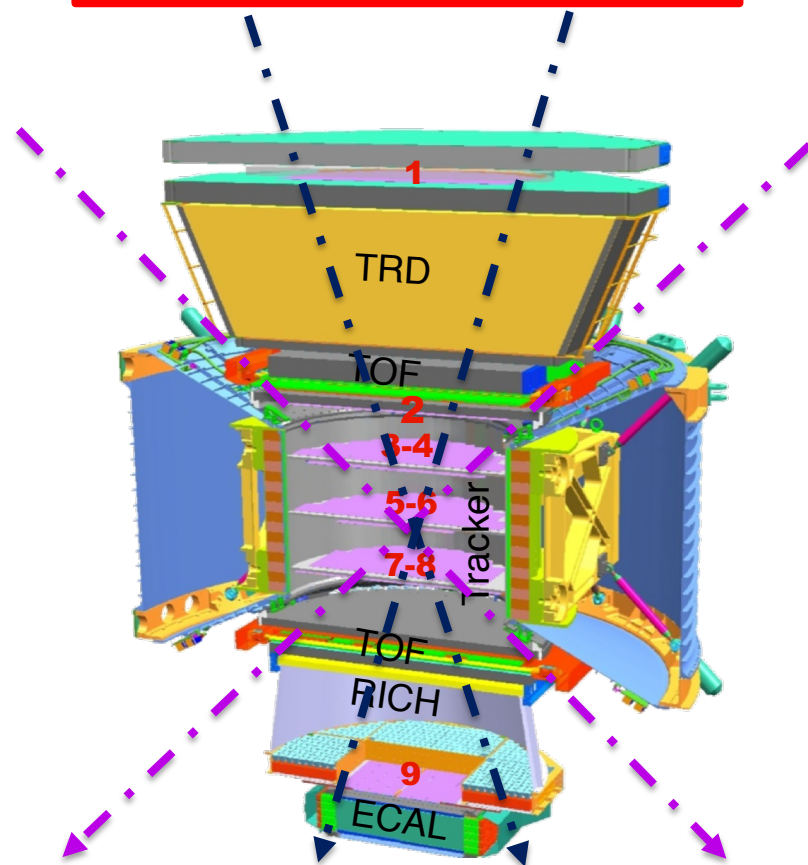
- HERD
- ALADInO
- AMS-100

DAMPE Field of View ~ 1 sr
 \rightarrow Acc ~ 0.3 m² sr



All the current and past detectors are designed as 'telescopes': they're sensitive only to particles impinging from "the top"
 limited FoV \rightarrow small acceptance

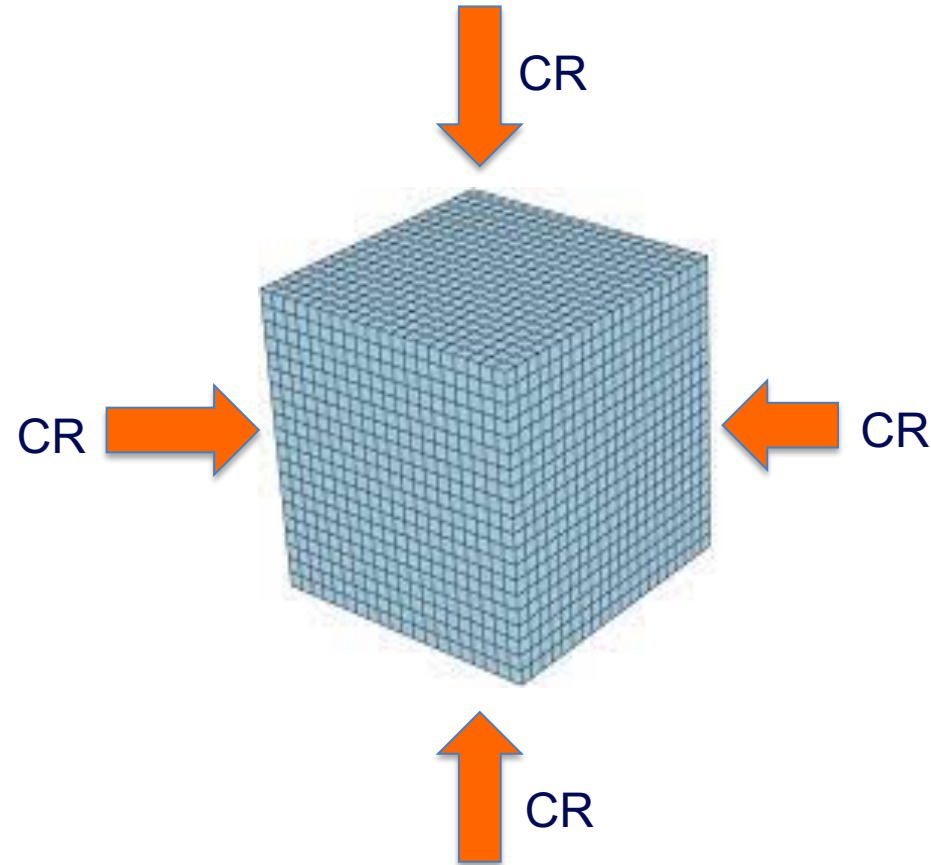
AMS Inner ~ 0.5 m² sr
 AMS Full Span ~ 0.05 m² sr



New paradigm - CaloCube

- Exploit the CR "isotropy" to maximize the effective geometrical factor, by using all the surface of the detector (aiming to reach $\Omega = 4\pi$)
- The calorimeter should be highly isotropic and homogeneous:
 - the needed depth of the calorimeter must be guaranteed for all the sides (i.e. cube, sphere, ...)
 - the segmentation of the calorimeter should be isotropic

→ this is in general doable just with an homogeneous calorimeter

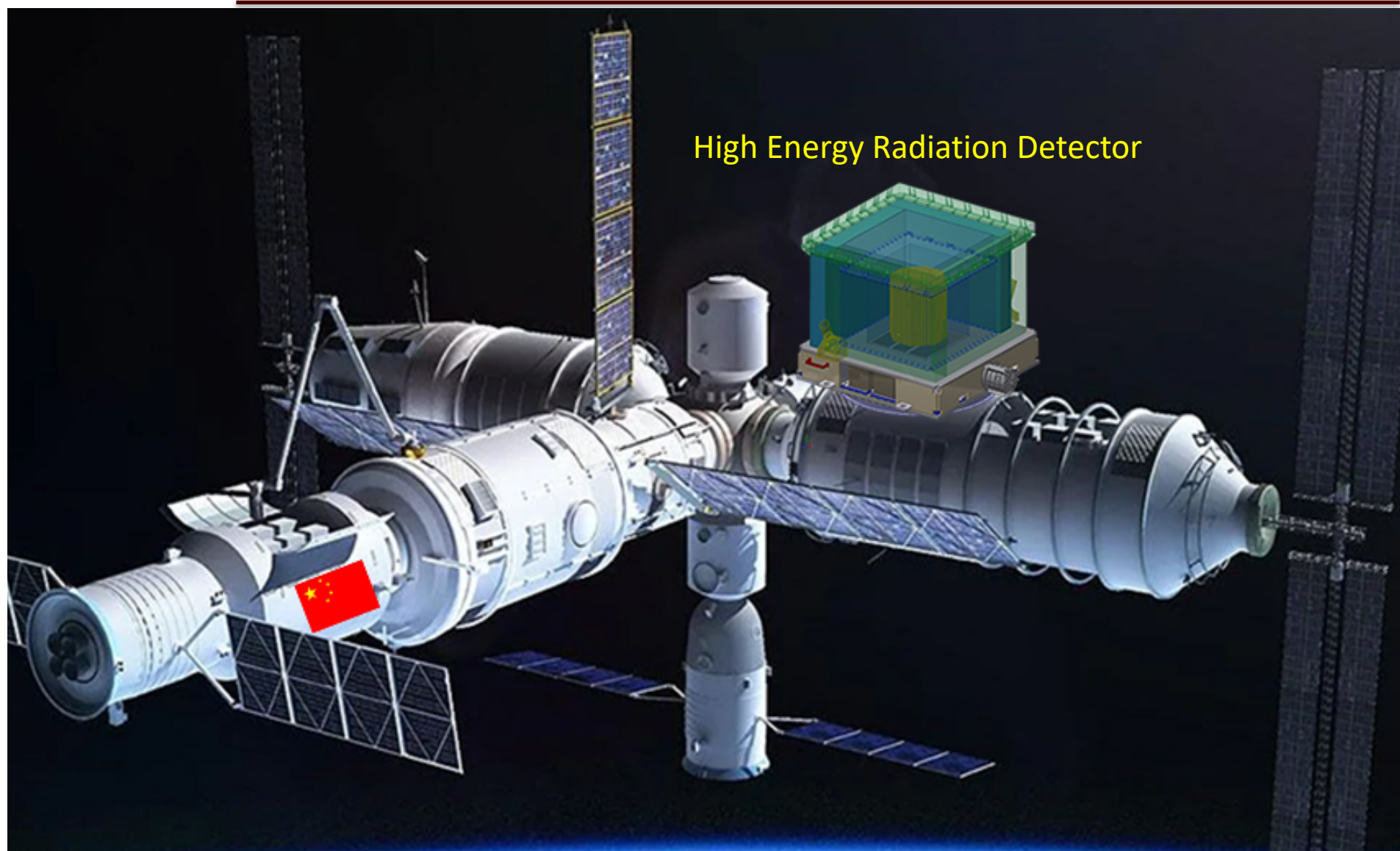


CaloCube is an INFN R&D initiated in Florence (Adriani et al.), almost always inspiring the next generation of large space cosmic rays detectors

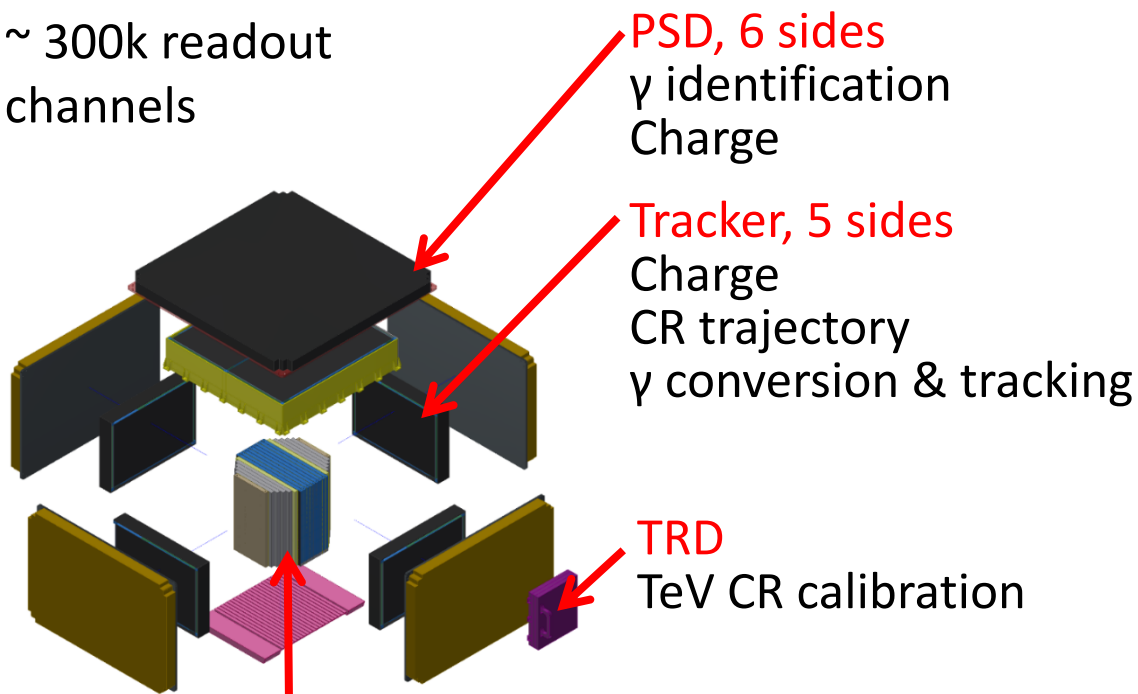
- HERD on the Chinese Space Station (CSS)
- ALADInO (in L2)
- AMS-100 (in L2)

HERD on the CSS

High Energy Radiation Detector



~ 300k readout channels

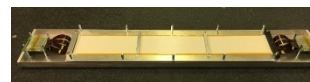


CALO: 3-D
 Energy
 e/p separation

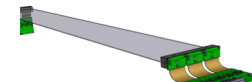
~7500 LYSO crystals ($55 X_0$, $3 \lambda_I$)
 Trigger sub-system
 Dual readout with IsCMOS & PD/SiPM

PS + SiPM

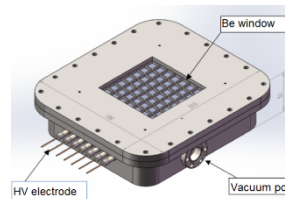
@INFN Bari & IHEP



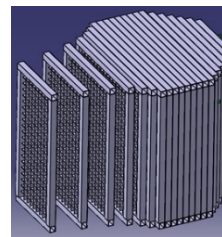
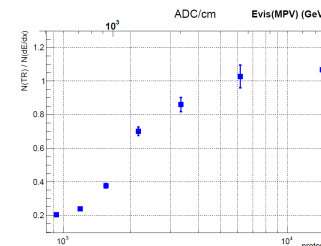
Silicon Track
 @INFN Perugia



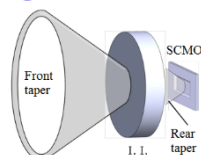
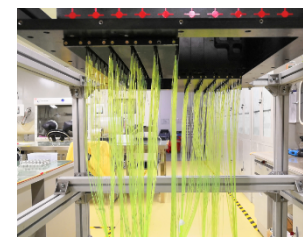
Fiber Tracker
 @Univ. of Geneva



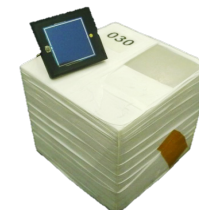
@Guangxi Univ.



@IHEP



@XIOPM

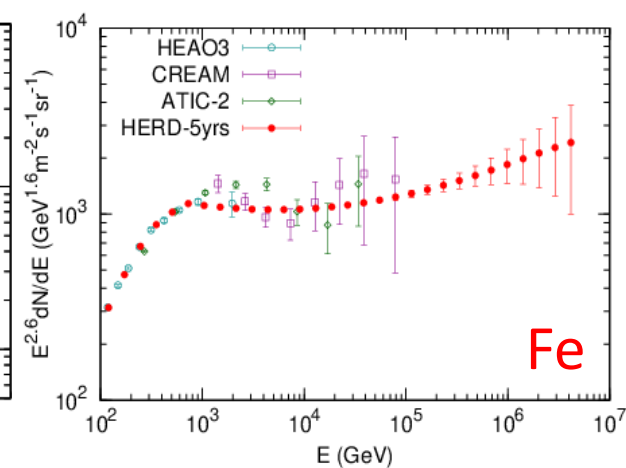
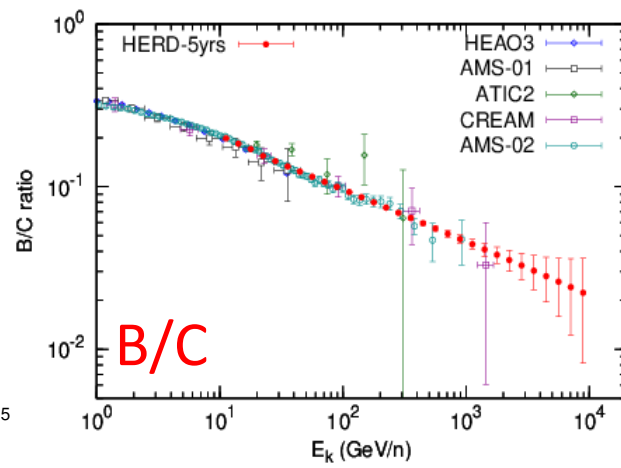
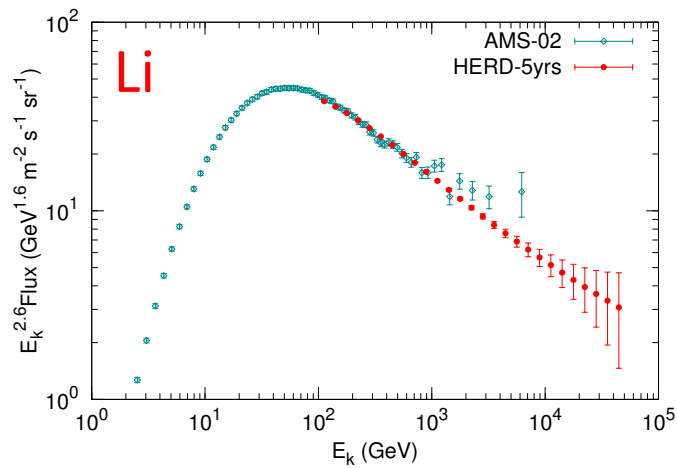
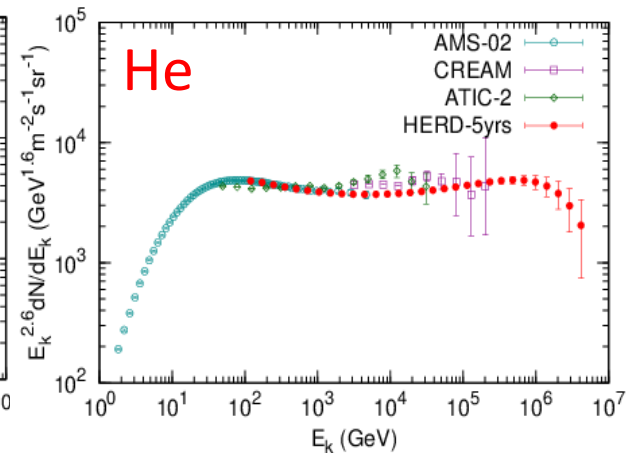
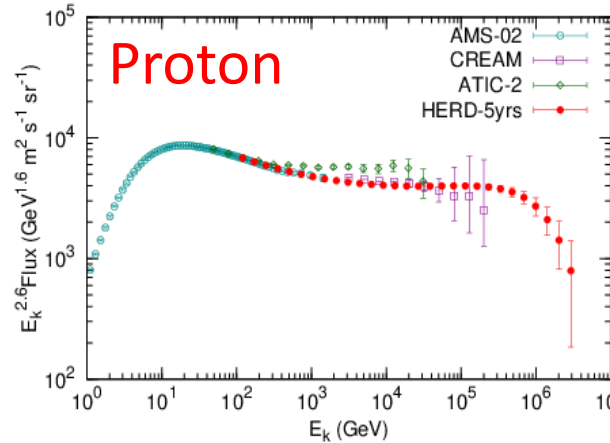
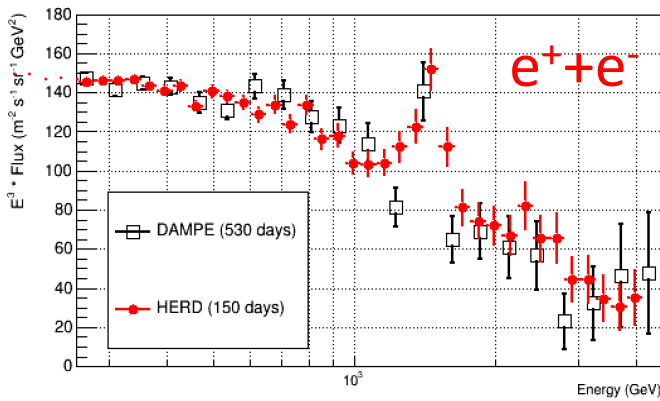


@INFN Florence

HERD performances

Item	Value
Energy range (e/ γ)	10 GeV - 100 TeV (e); 0.5 GeV-100 TeV (γ)
Energy range (nuclei)	30 GeV - 3 PeV
Angle resolution	0.1 deg.@10 GeV
Charge resolution	0.1-0.15 c.u
Energy resolution (e)	1-1.5%@200 GeV
Energy resolution (p)	20-30%@100 GeV - PeV
e/p separation	$\sim 10^{-6}$
G.F. (e)	$>3 \text{ m}^2\text{sr}@200 \text{ GeV}$
G.F. (p)	$>2 \text{ m}^2\text{sr}@100 \text{ TeV}$
Field of View	$\sim 6 \text{ sr}$
Envelope (L*W*H)	$\sim 2300*2300*2000 \text{ mm}^3$
Weight	$\sim 4000 \text{ kg}$
Power Consumption	$\sim 1400 \text{ W}$

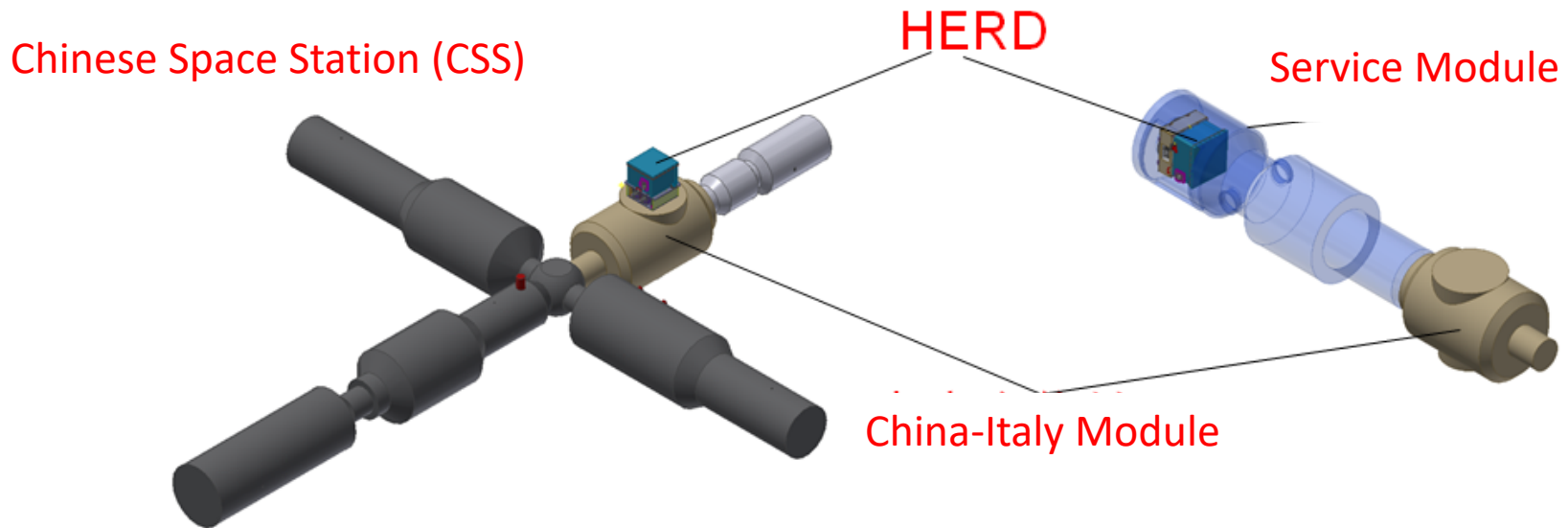
HERD capabilities



- The HERD consortium includes 130+ scientists from China, Italy, Switzerland, Spain, Germany, Denmark, Sweden, Russia, ... (new collaborators are welcome!)
 - most of the members have been collaborating on previous high energy experiments, both on hardware development and data analysis
- 7 HERD international workshops have been organized in China and Europe since 2012. Last one in China in December 2019.
- 3 CERN beam tests on HERD prototypes have been successfully implemented by Chinese and European colleagues.

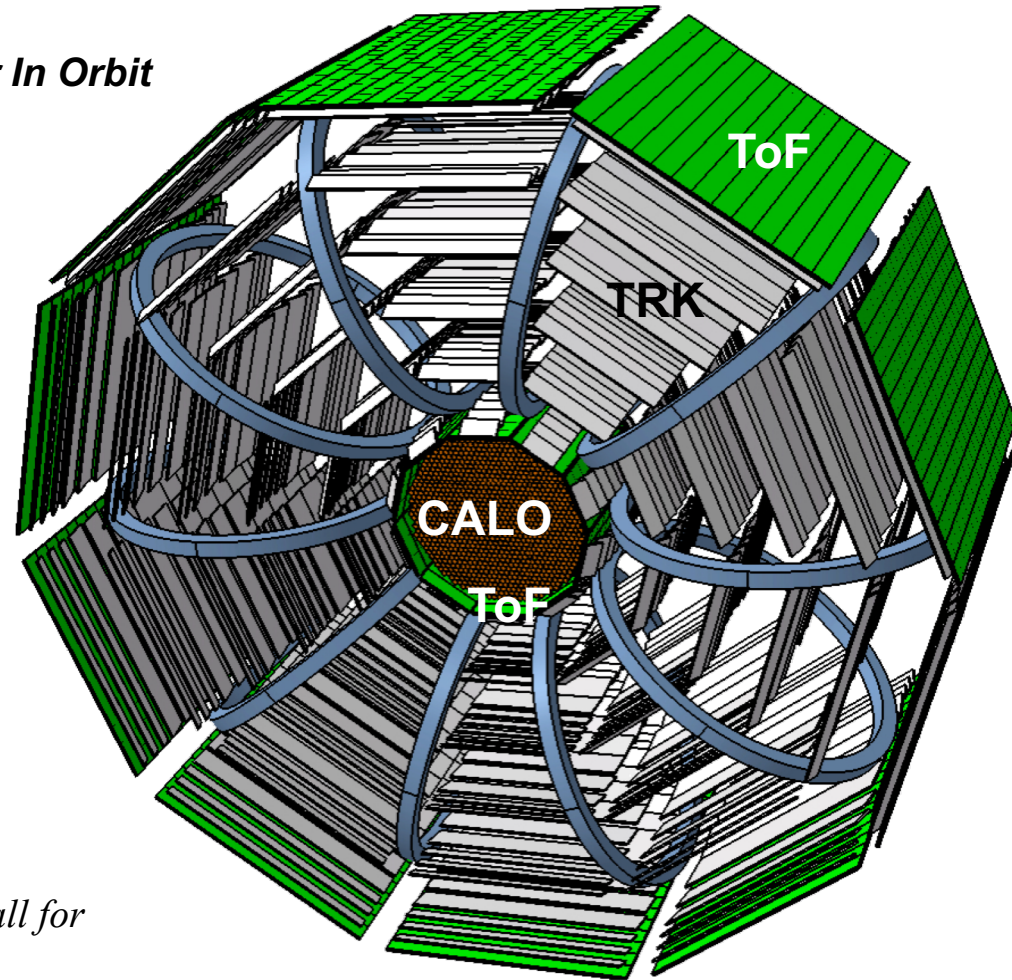


- Mission concept
 - Launched with the China-Italy Module and installed on the Module.
 - Periodic calibration is performed every 3-6 months.
 - Several devices are replaced or upgraded every 3-4 years.
 - Telemetry is achieved with the help of relay satellites.
- The HERD proposal was reviewed positively in May 2018 at ASI.
- HERD is written into the joint declaration between China & Italy during the visit of President Xi Jinping in March 2019.



High Precision Particle Astrophysics as a New Window on the Universe

*with an Antimatter Large Acceptance Detector In Orbit
(ALADInO)*



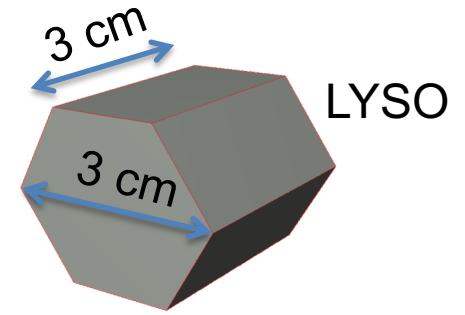
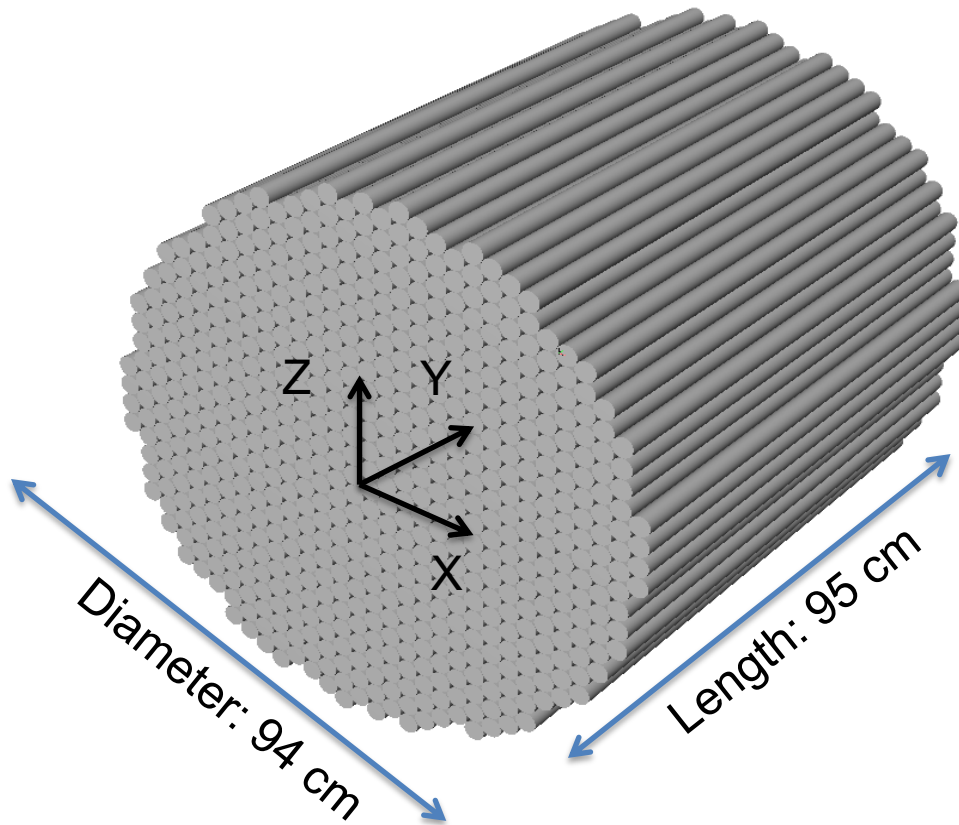
*A White Paper submitted in response to ESA's Call for
the VOYAGE 2050 long-term plan*

<https://www.cosmos.esa.int/web/voyage-2050/white-papers>

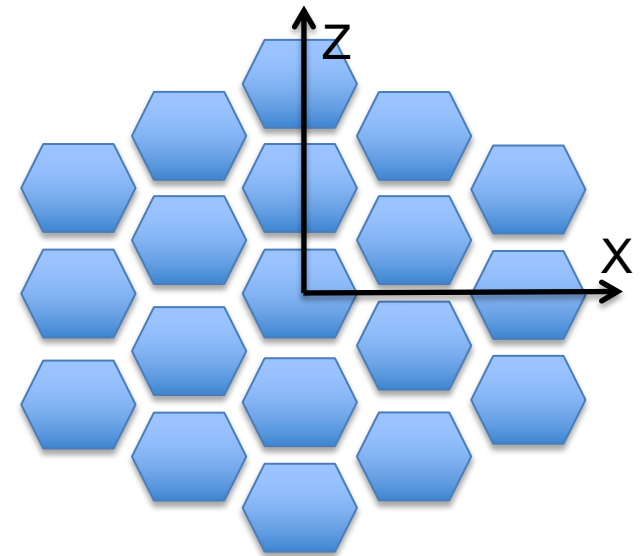
https://www.cosmos.esa.int/documents/1866264/3219248/BattistonR_ALADINO_PROPOSAL_20190805_v1.pdf

ALADInO calorimeter

Weight~(2300+300) kg
N. crystals: ~20.000



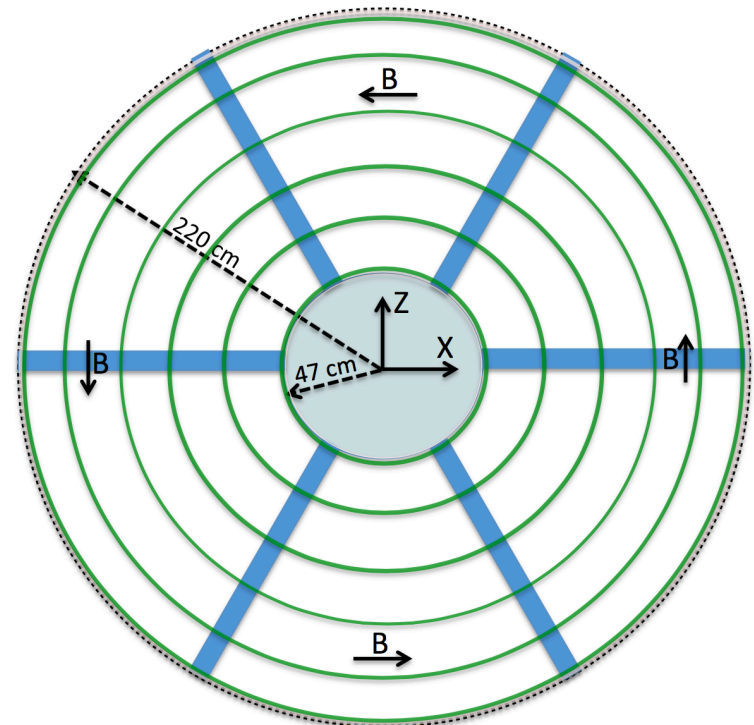
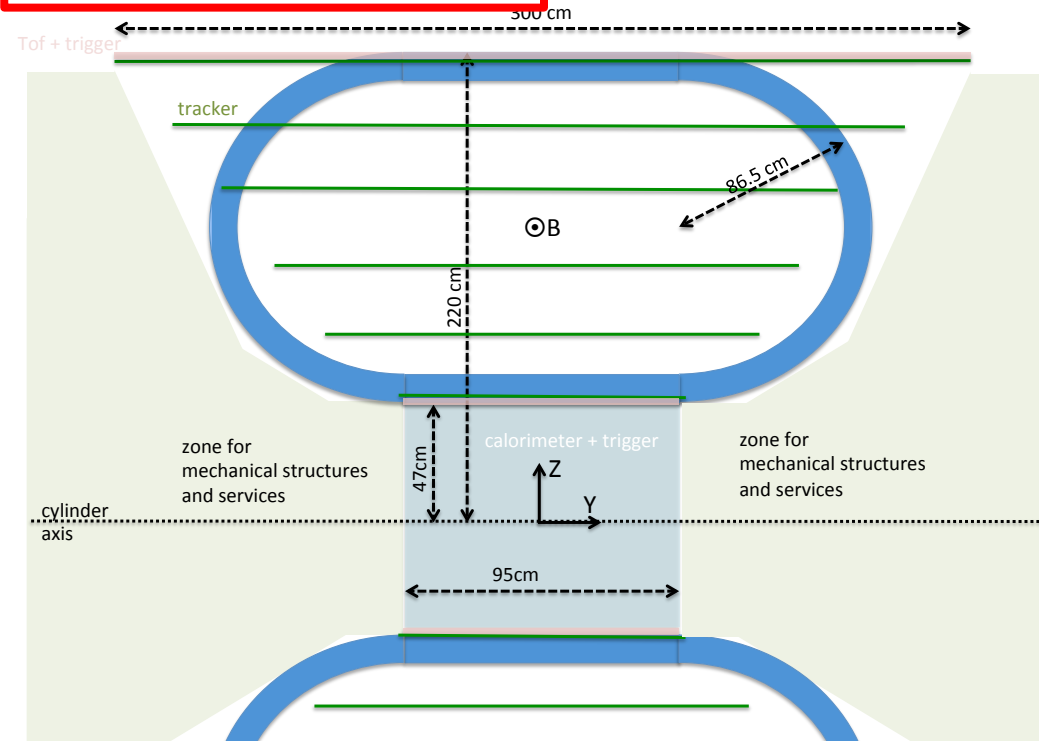
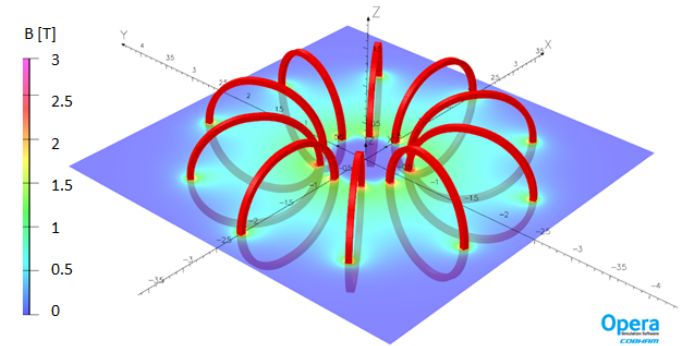
Basic crystal:
hexagonal base
prisma



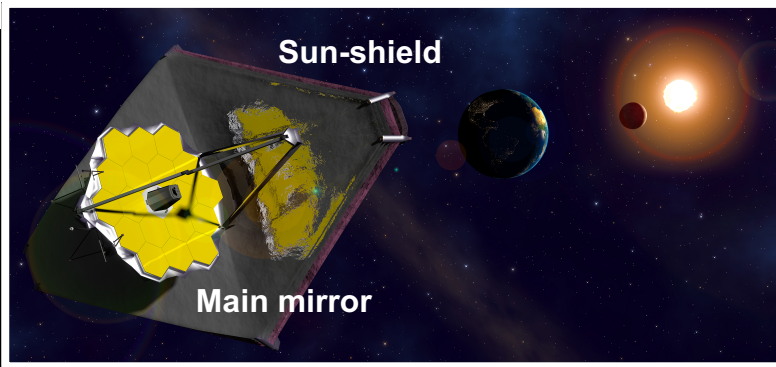
ALADInO magnet

Benefit from the R&D of high temperature superconducting magnets (MgB_2 , YBCO and in particular REBCO) for space applications ($T \approx 15 \div 40^\circ \text{ K}$)

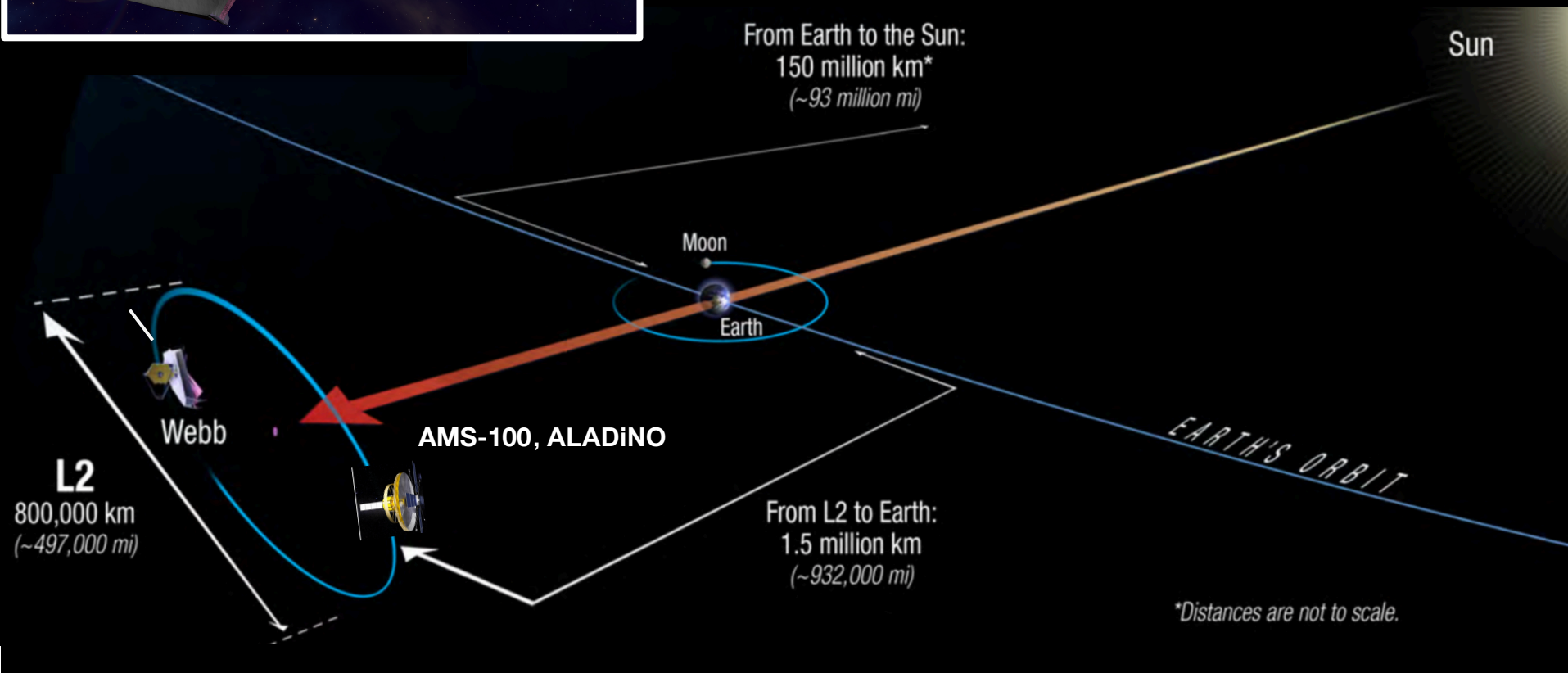
Field 0.8 T
Bending power $> 1.1 \text{ T m}$
Weight $\sim 1000 \text{ Kg}$



James Webb Telescope - L2

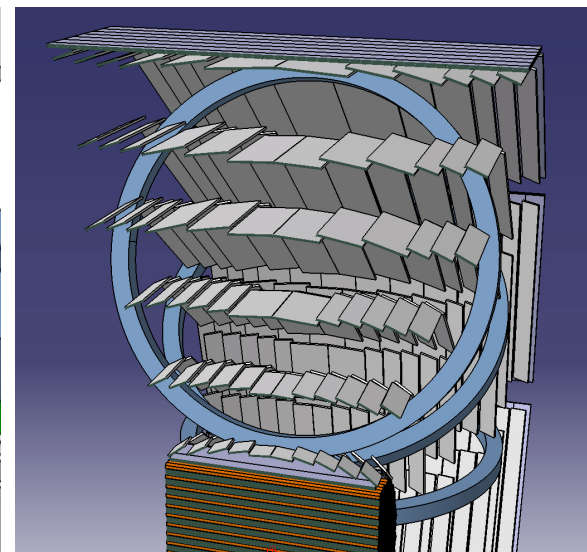
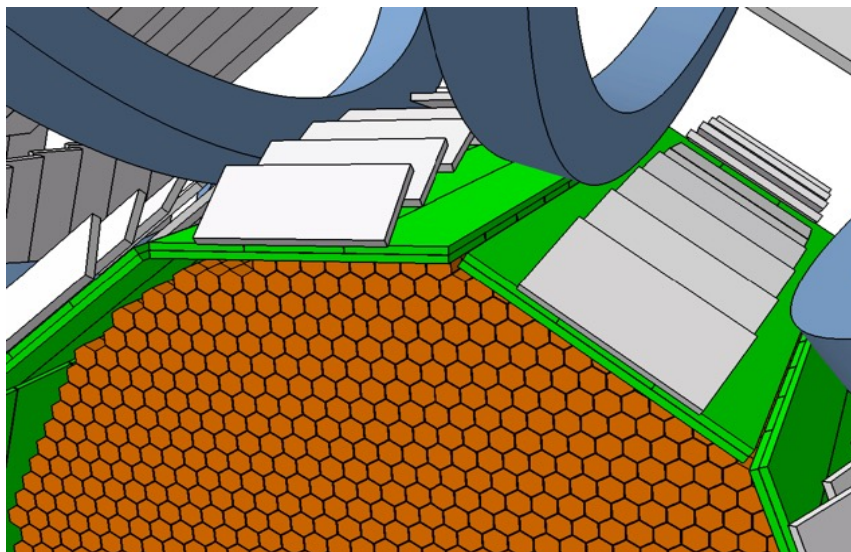


The best place where to operate a cryogenic superconducting magnet is the Lagrange Point 2, like the Webb space telescope

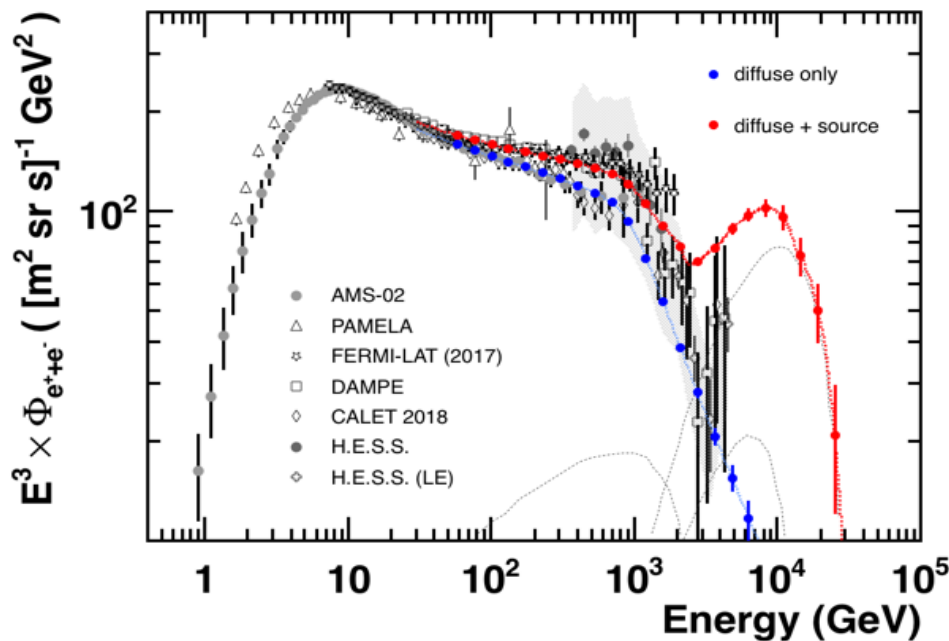


ALADInO performances

Calorimeter acceptance	$\sim 9 \text{ m}^2 \text{ sr}$
Spectrometer acceptance	$> 10 \text{ m}^2 \text{ sr}$ ($\sim 3 \text{ m}^2 \text{ sr}$ w/i CALO)
Spectrometer Maximum Detectable Rigidity (MDR)	$> 20 \text{ TV}$
Calorimeter depth	$61 X_0, 3.5 \lambda_I$
Calorimeter energy resolution	$25\% \div 35\%$ (for nuclei) 2% (for electrons and positrons)
Calorimeter e/p rejection power	$> 10^5$
Time of Flight measurement resolution	$\sim 100 \text{ ps}$
High energy γ -ray acceptance (Calorimeter)	$\sim 9 \text{ m}^2 \text{ sr}$
Low energy γ -ray acceptance (Tracker)	$\sim 0.5 \text{ m}^2 \text{ sr}$
γ -ray Point Spread Function	$< 0.5 \text{ deg}$



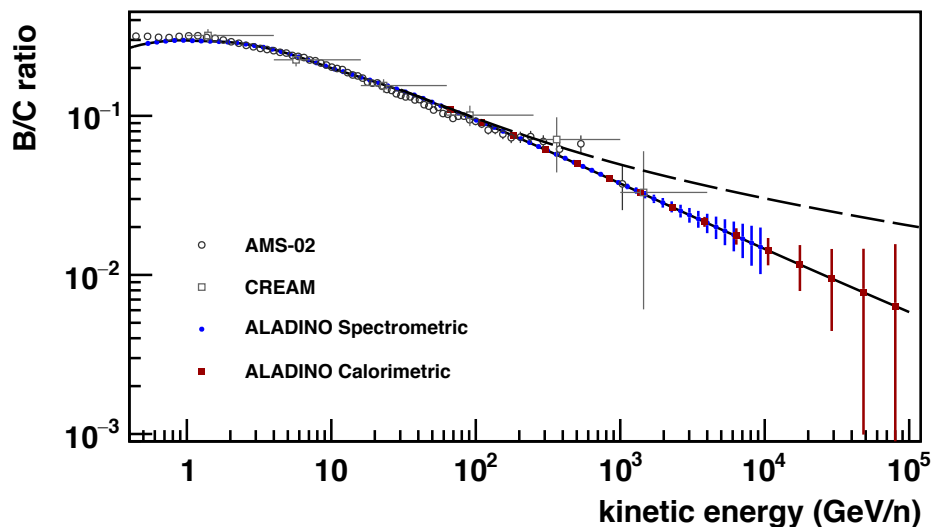
Weight: ~ 6 Tons
Power: ~ 4 kW
channels: 2.5 M



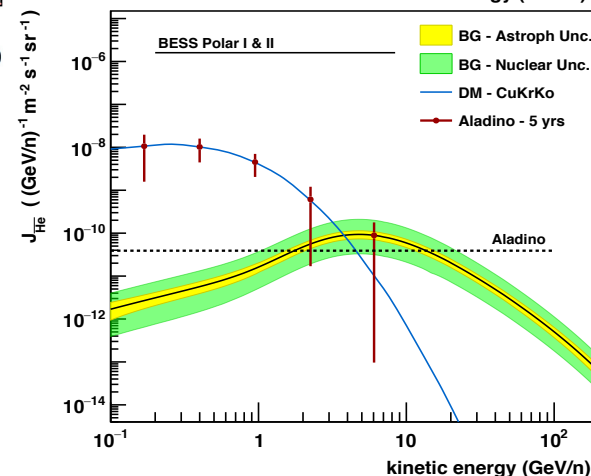
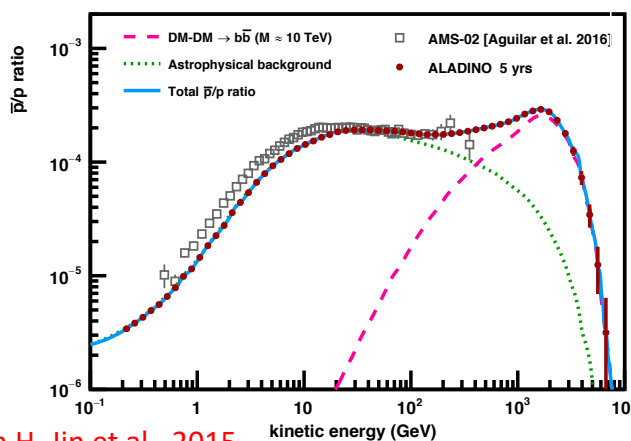
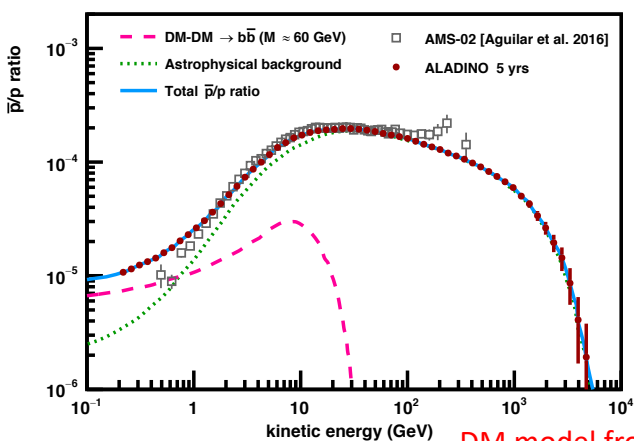
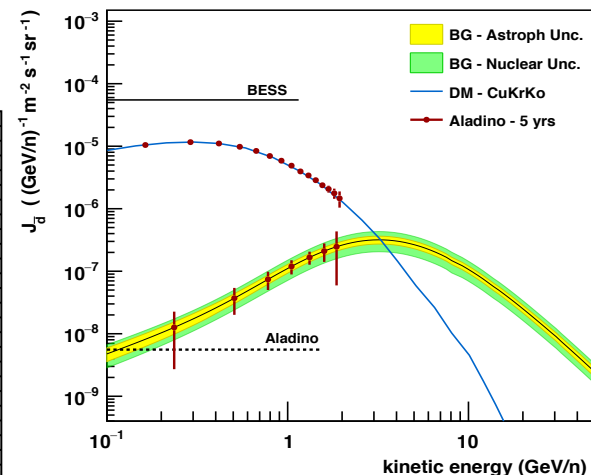
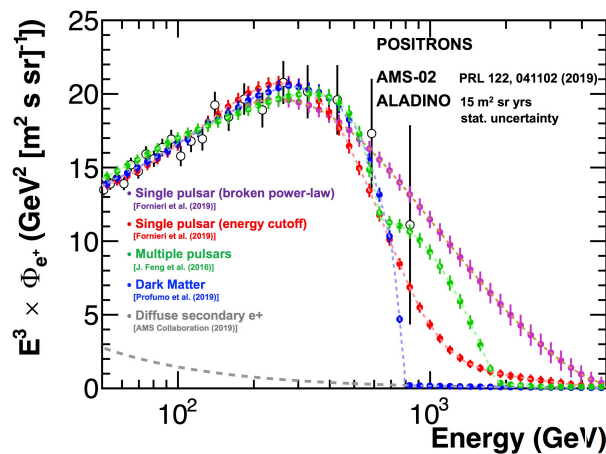
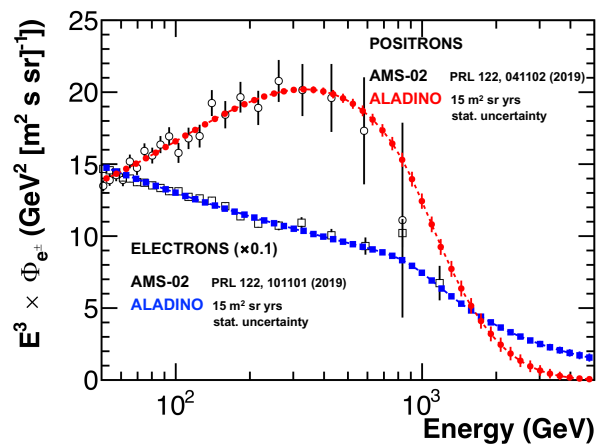
The possibility to crosscheck the Energy in the Calorimeter with the Rigidity (i.e. Momentum) in the Spectrometer will permit lower systematics

The calorimeter is slightly bigger than the HERD one but is very similar:

- similar statistical errors similar energy reach
- similar energy resolution, both for electromagnetic particles and nuclei



ALADInO capabilities

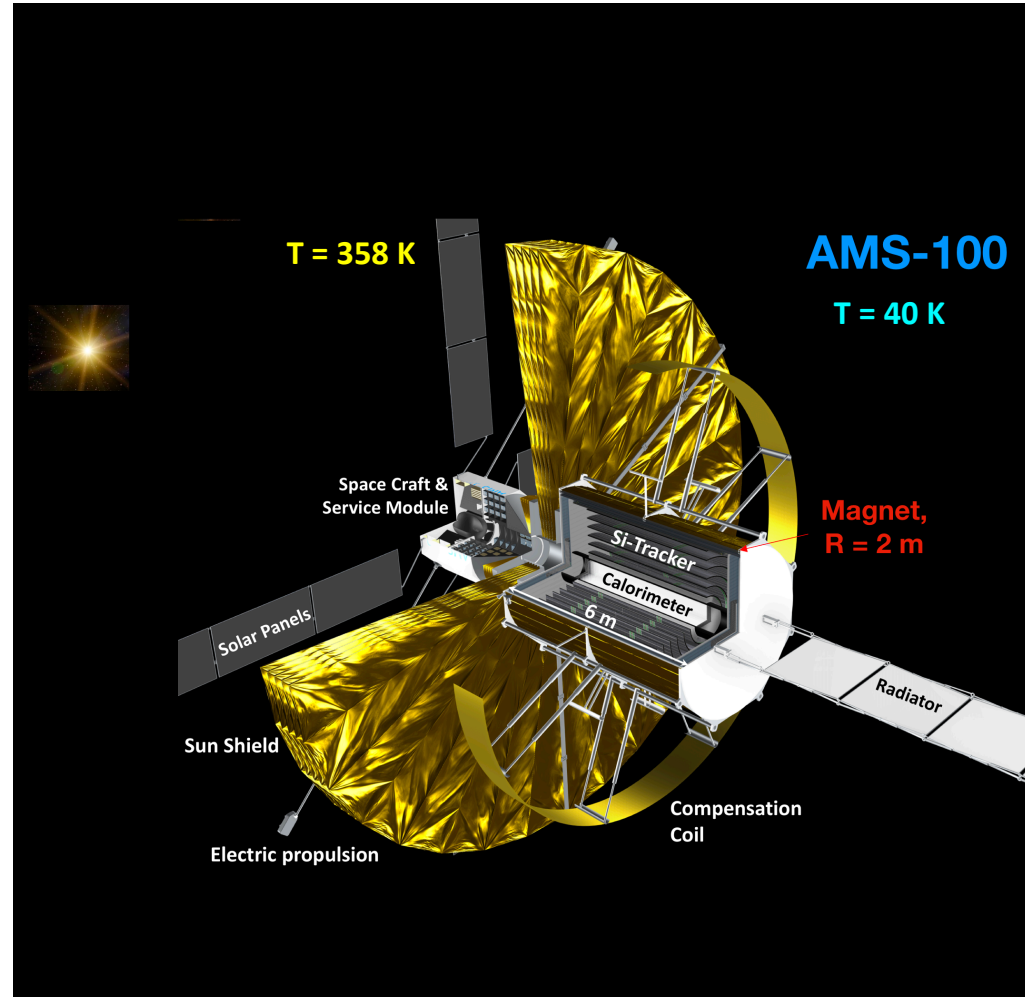
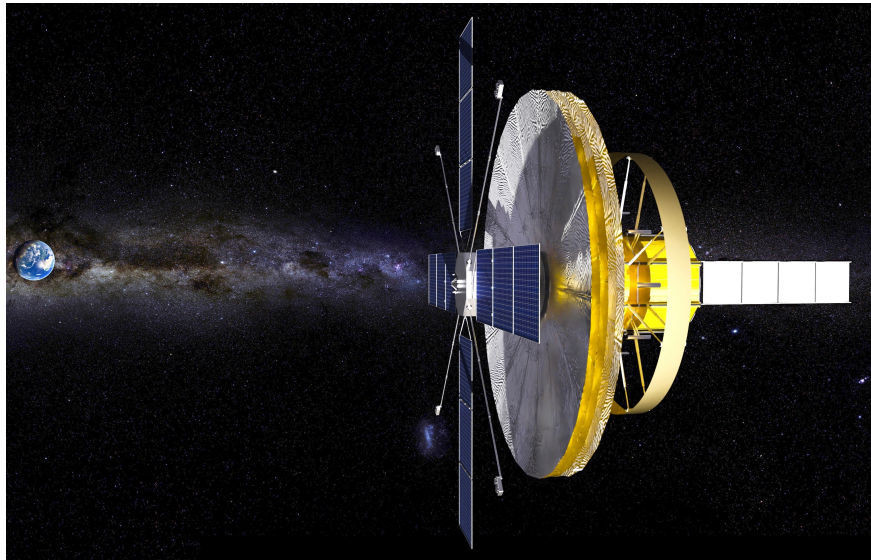


DM model from H. Jin et al., 2015, Phys. Rev. D 92; no. 5, 055027

DM model from A. Cuoco et al. 2017, Phys. Rev. Lett. 118, 191102, M. Korsmeier et al., 2018, Phys. Rev. D 97 n.10, 103011
BKG model from N. Tomassetti and A. Oliva, 2017, ApJ Lett. 844

AMS-100

The Next Generation Magnetic Spectrometer in Space –
An International Science Platform for Physics and Astrophysics at
Lagrange Point 2



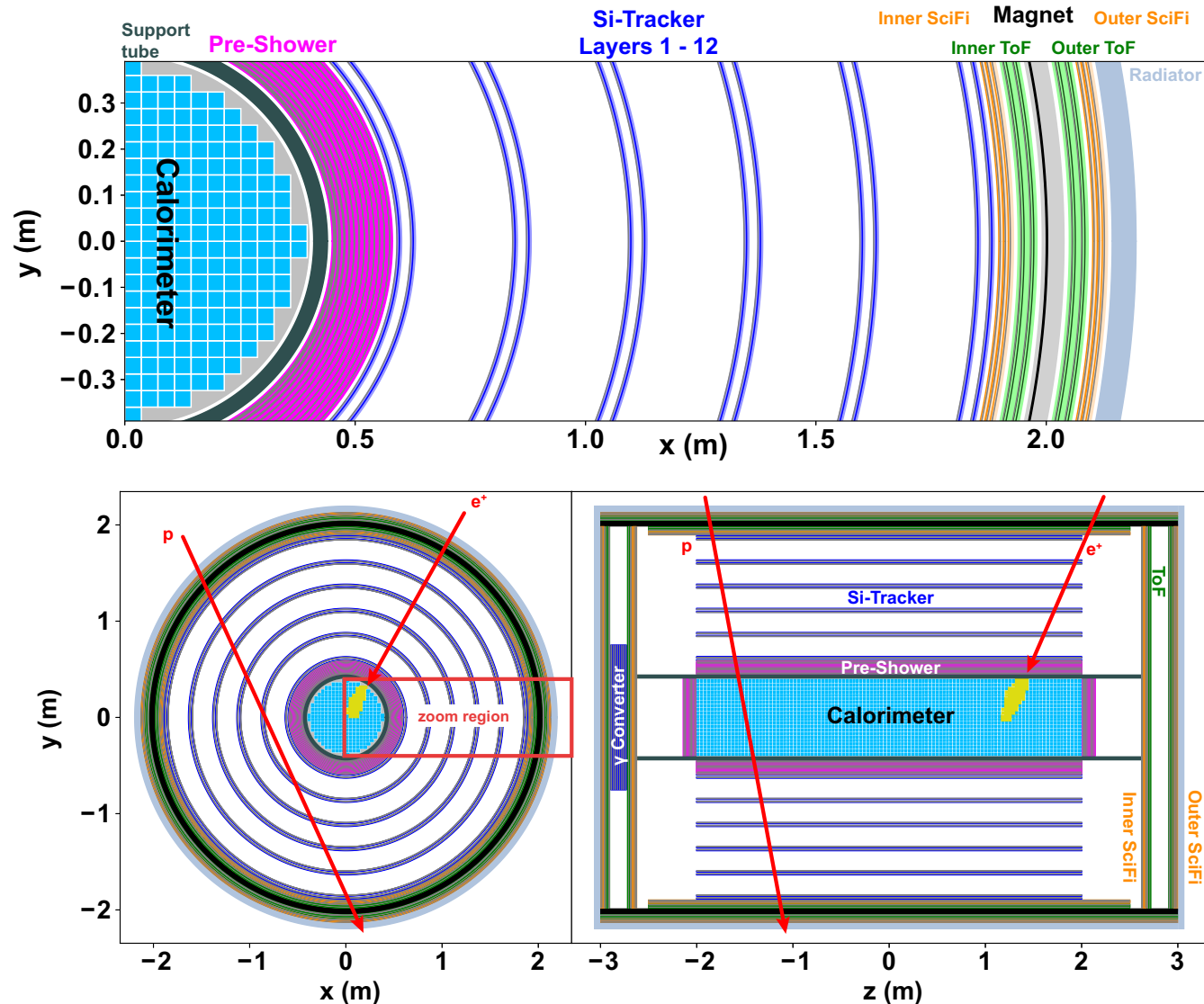
<https://www.cosmos.esa.int/web/voyage-2050/white-papers>

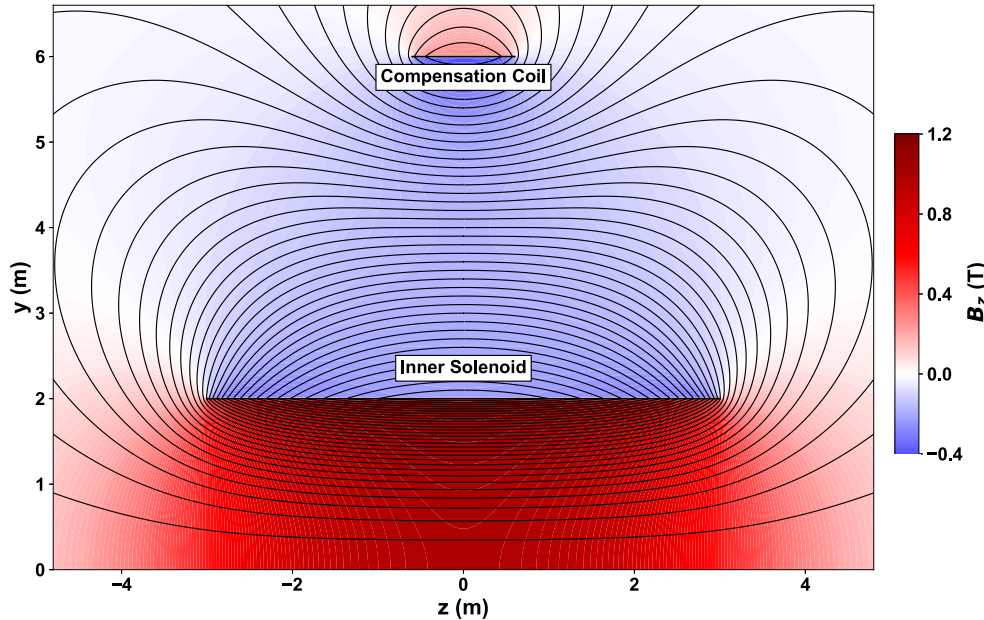
https://www.cosmos.esa.int/documents/1866264/3219248/SchaelS_AMS100_Voyage2050.pdf

arXiv:1907.04168v1 [astro-ph.IM] 9 Jul 2019

AMS-100 detector

- The Calorimeter is essentially based on the HERD design
- A Pre-Shower (silicon detectors + tungsten) is foreseen to provide an angular resolution for γ -rays similar to the Fermi-LAT one
- An additional external γ -ray converter on the end-cap is foreseen to increase the γ -ray acceptance

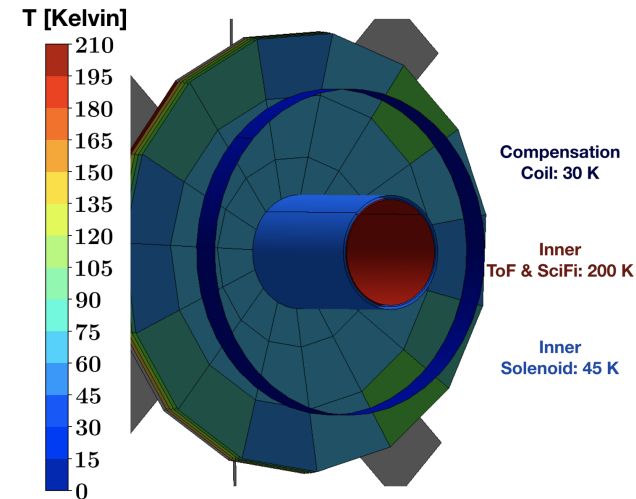
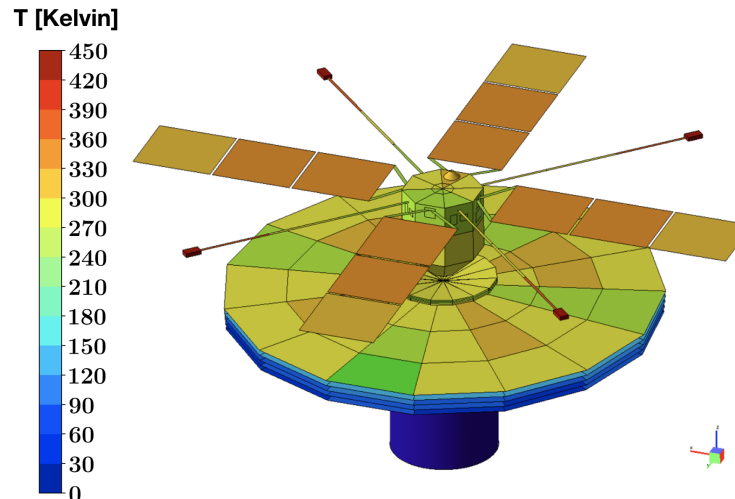




An (expandable) compensation coil balances the magnetic moment of the solenoid and allows the attitude control of the instrument within the heliospheric magnetic field

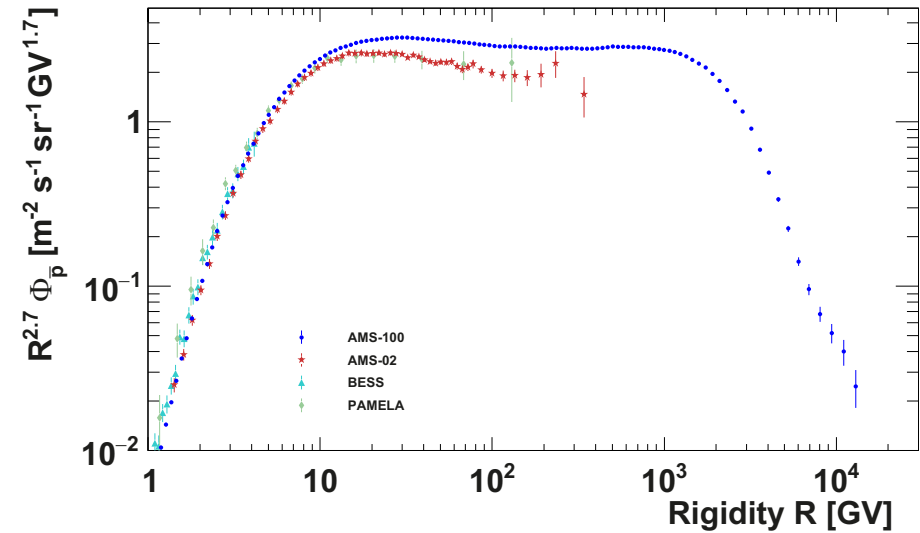
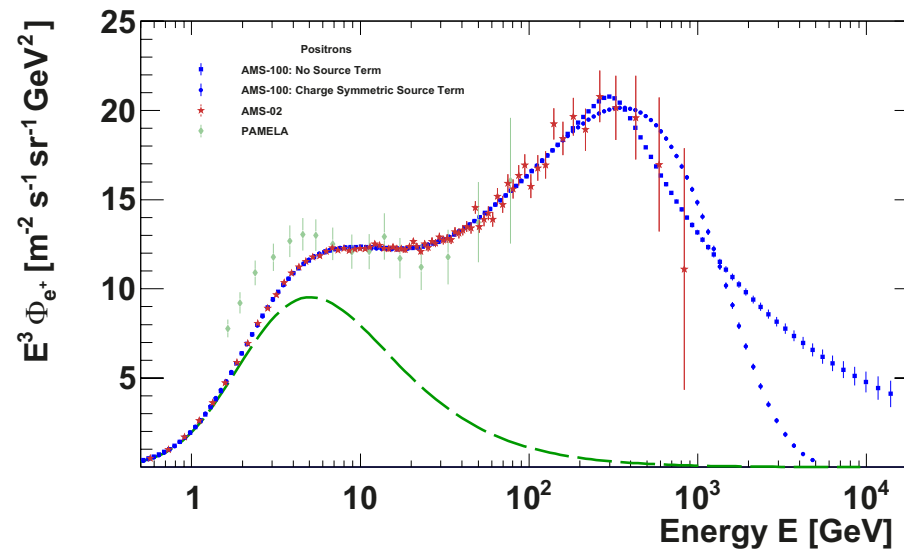
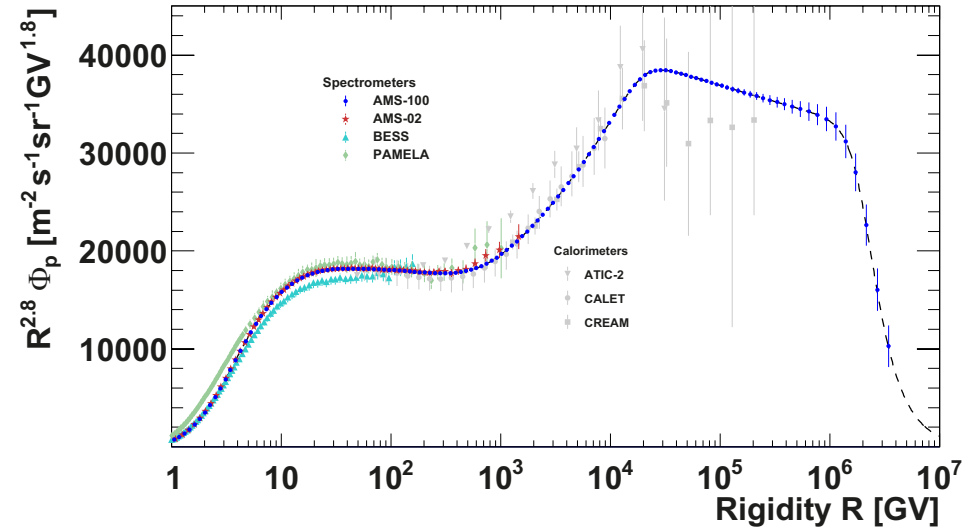
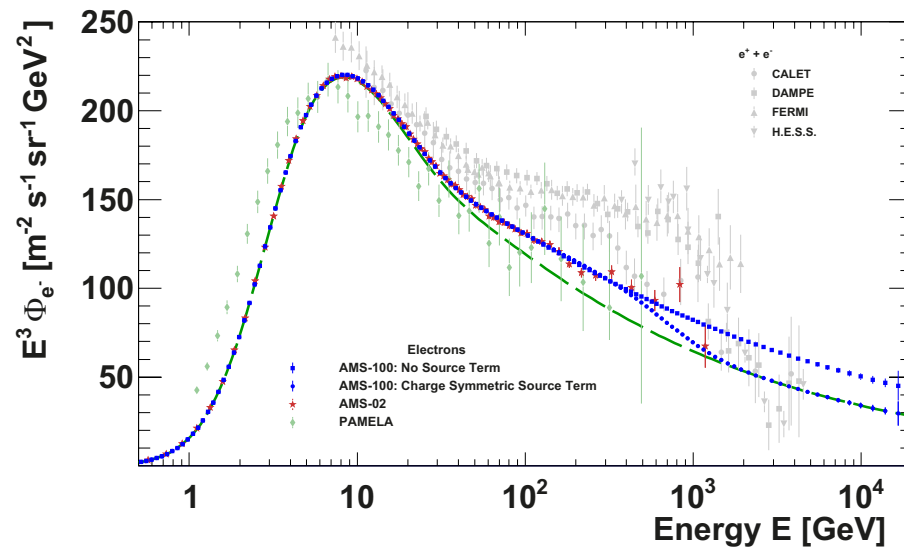
The High Temperature Superconducting magnetic system is based on REBCO tapes operated at 50-60° K

The sunshield is a key component of AMS-100, allowing the HTS magnet to operate without cryogenics. It has a radius of 9 m and is designed similar to the concept developed for the James Webb Space Telescope

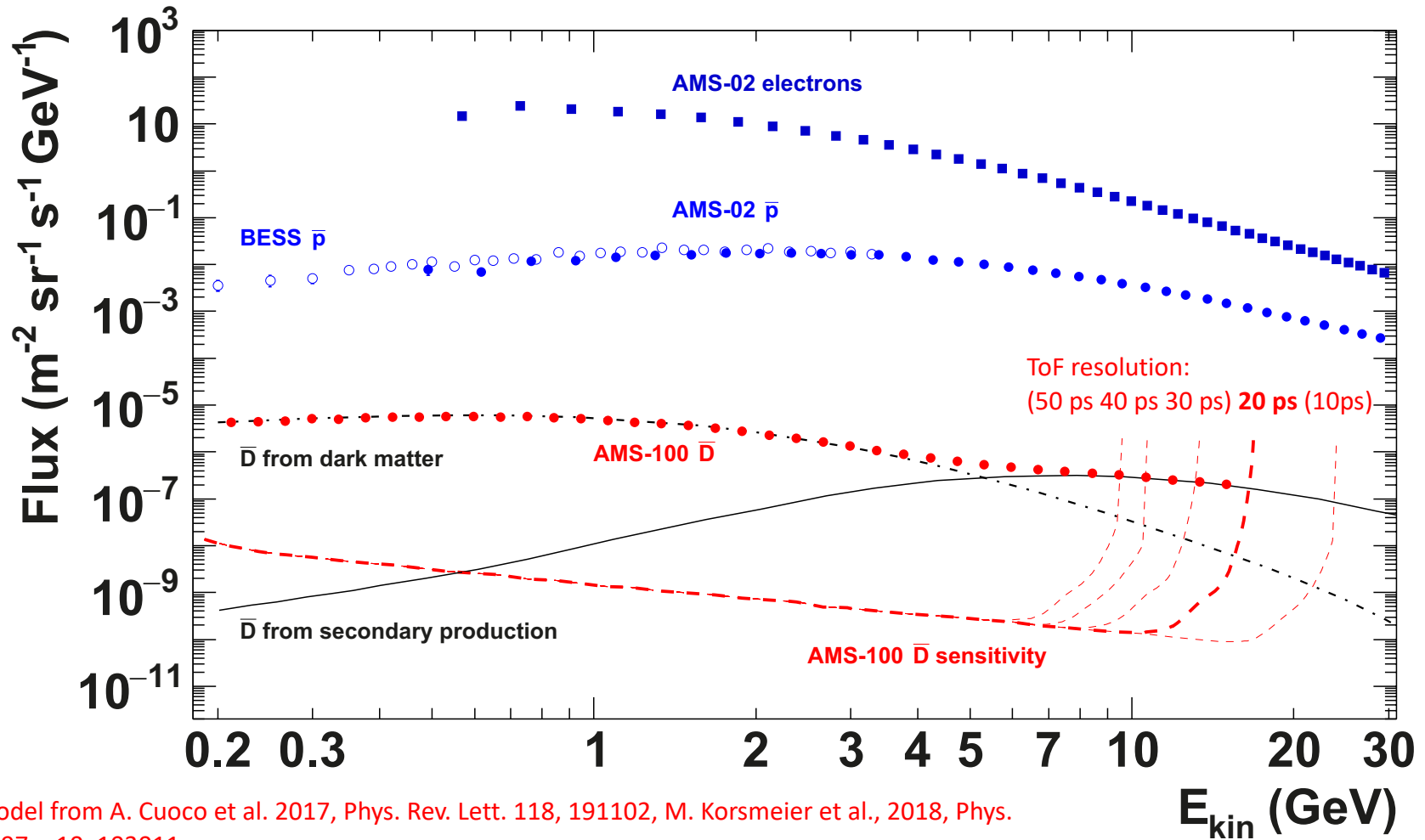


AMS-100 performances

Quantity	Value
Acceptance	100 m ² sr
MDR	100 TV for $ Z = 1$
Material budget	0.12 X_0
of main solenoid	0.012 λ_I
Calorimeter depth	70 X_0 , 4 λ_I
Energy reach	10 ¹⁶ eV for nucleons 10 TeV for e^+ , \bar{p} 8 GeV/n for \bar{D}
Angular resolution	4'' for photons at 1 TeV 0''4 for photons at 10 TeV
Spatial resolution (SciFi)	40 μ m
Spatial resolution (Si-Tracker)	5 μ m
Time resolution of single ToF bar	20 ps
Incoming particle rate	2 MHz
High-level trigger rate	few kHz
Downlink data rate	~28 Mbps
Instrument weight	43 t
Number of readout channels	8 million
Power consumption	15 kW
Mission flight time	10 years



AMS-100 capabilities



DM model from A. Cuoco et al. 2017, Phys. Rev. Lett. 118, 191102, M. Korsmeier et al., 2018, Phys.

Rev. D 97 n.10, 103011

BKG model from S.-J. Lin, X.-J. Bi, and P.-F. Yin, arXiv e-prints (2018). arXiv:1801.00997.

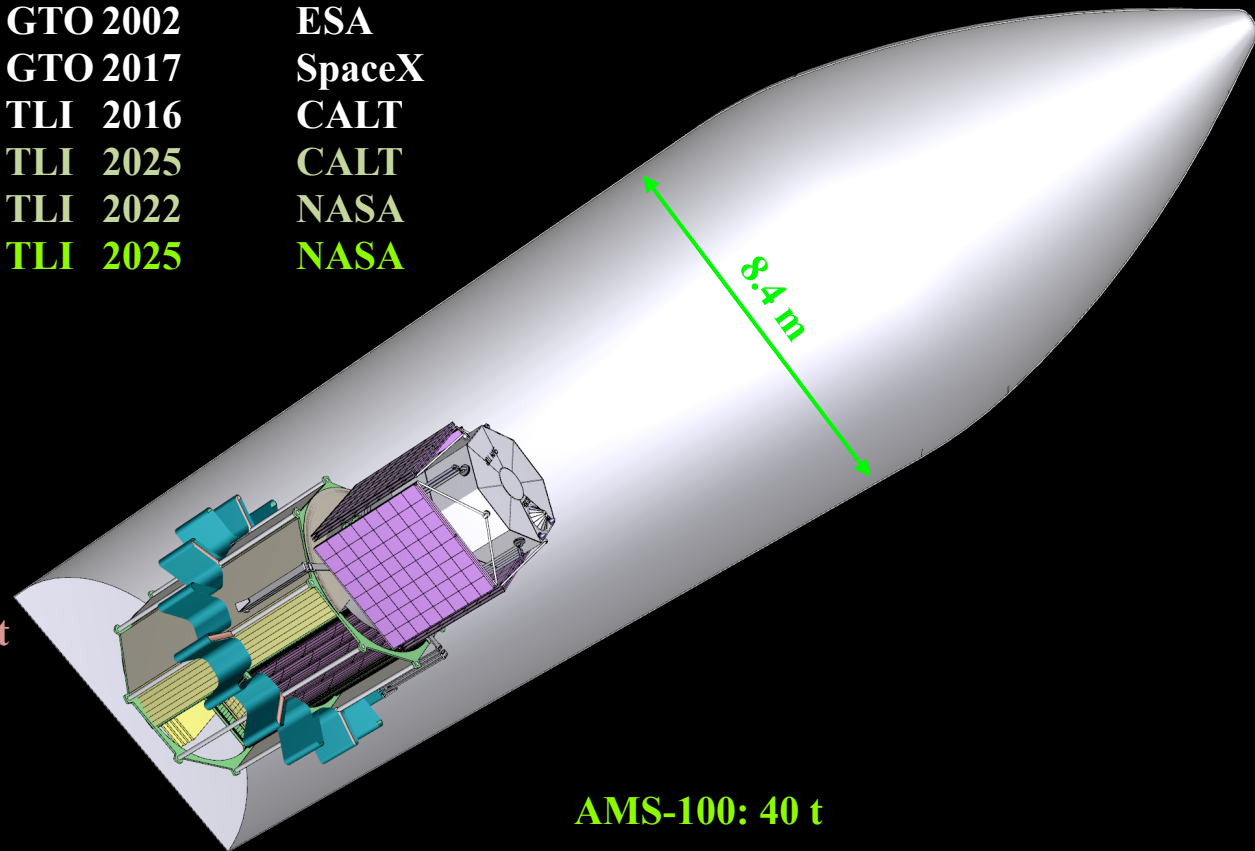
For anti-He the expected sensitivity is $\sim 10^{-11}$

Current and upcoming rockets

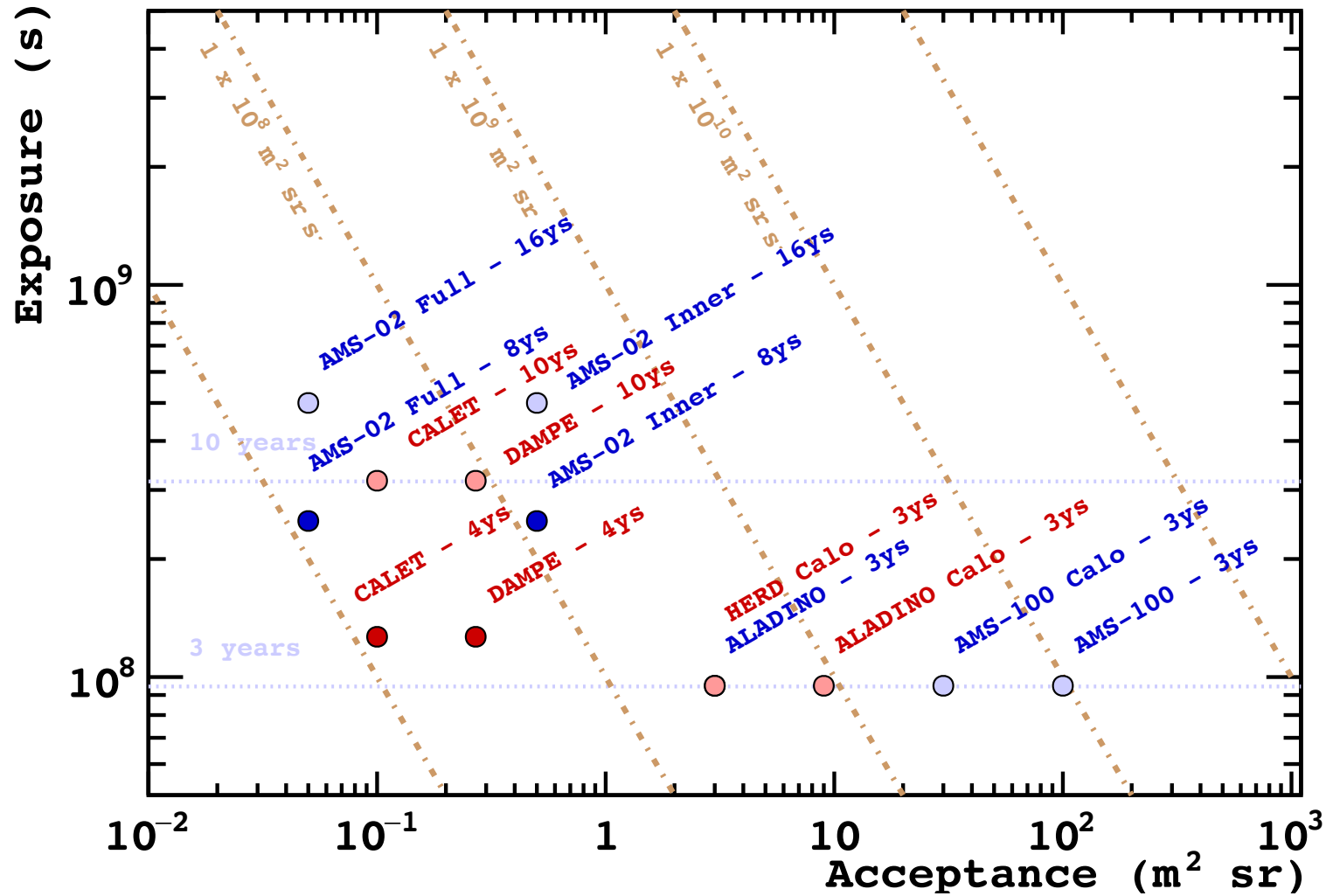
Name	LEO [kg]	other [kg]	First flight	
Ariane 5	21,000	10,730 GTO	2002	ESA
Falcon Heavy	63,800	26,700 GTO	2017	SpaceX
Long March 5	25,000	8,000 TLI	2016	CALT
Long March 9	130,000	50,000 TLI	2025	CALT
SLS Block 1B	105,000	39,100 TLI	2022	NASA
SLS Block 2	130,000	45,000 TLI	2025	NASA

Operational
Under development

LEO: Low Earth orbit
GTO: Geostationary transfer orbit
TLI: Trans-lunar injection



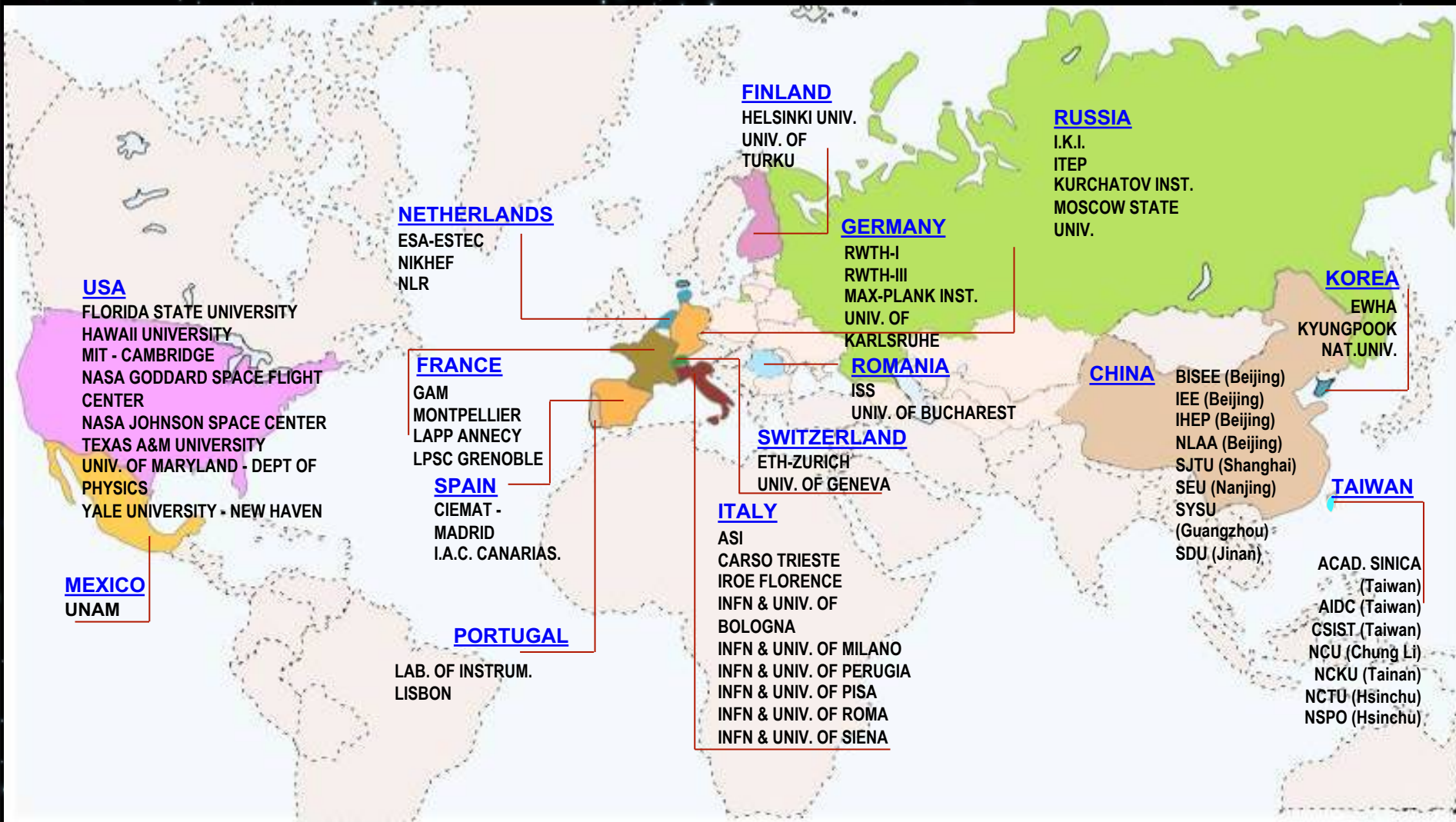
Item	HERD	ALADINO	AMS-100
Electromagnetic calorimeter depth	$55 X_0$	$61 X_0$	$70 X_0$
Hadronic calorimeter depth	$3 \lambda_1$	$3.5 \lambda_1$	$4 \lambda_1$
MDR	-	20 TV	100 TV
Acceptance (spectrometer)	-	$\sim 10 \text{ m}^2 \text{ sr}$	$\sim 100 \text{ m}^2 \text{ sr}$
Acceptance (spectrometer + calorimeter)	-	$\sim 3 \text{ m}^2 \text{ sr}$	$\sim 30 \text{ m}^2 \text{ sr}$
Acceptance (calorimeter)	$\sim 3 \text{ m}^2 \text{ sr}$	$\sim 9 \text{ m}^2 \text{ sr}$	$\sim 30 \text{ m}^2 \text{ sr}$
# of channels	300 k	2.5 M	8 M
Weight	$\sim 4000 \text{ kg}$	$\sim 6000 \text{ kg}$	$\sim 40000 \text{ kg}$
Power Consumption	$\sim 1400 \text{ W}$	$\sim 4000 \text{ W}$	$\sim 15000 \text{ W}$

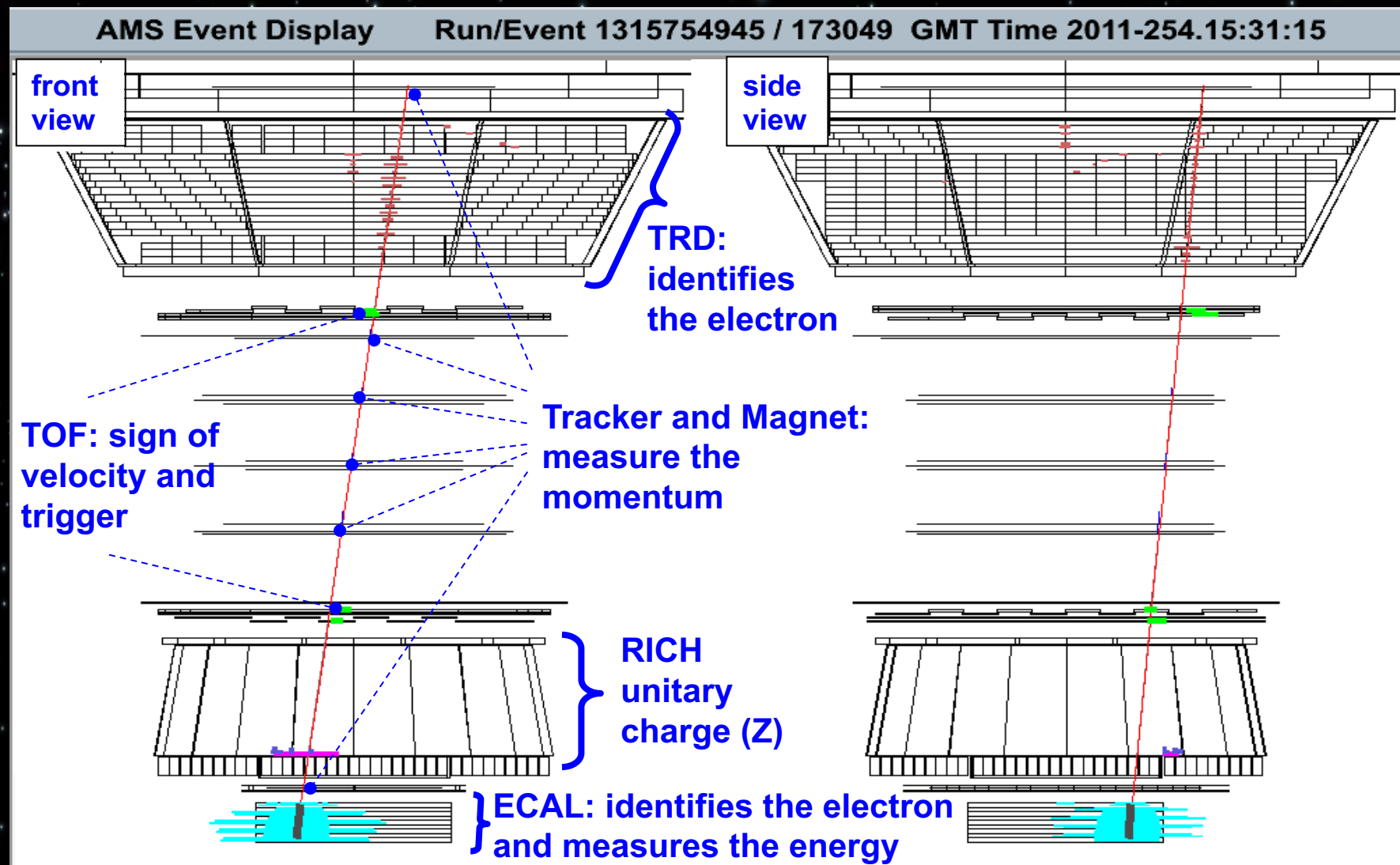


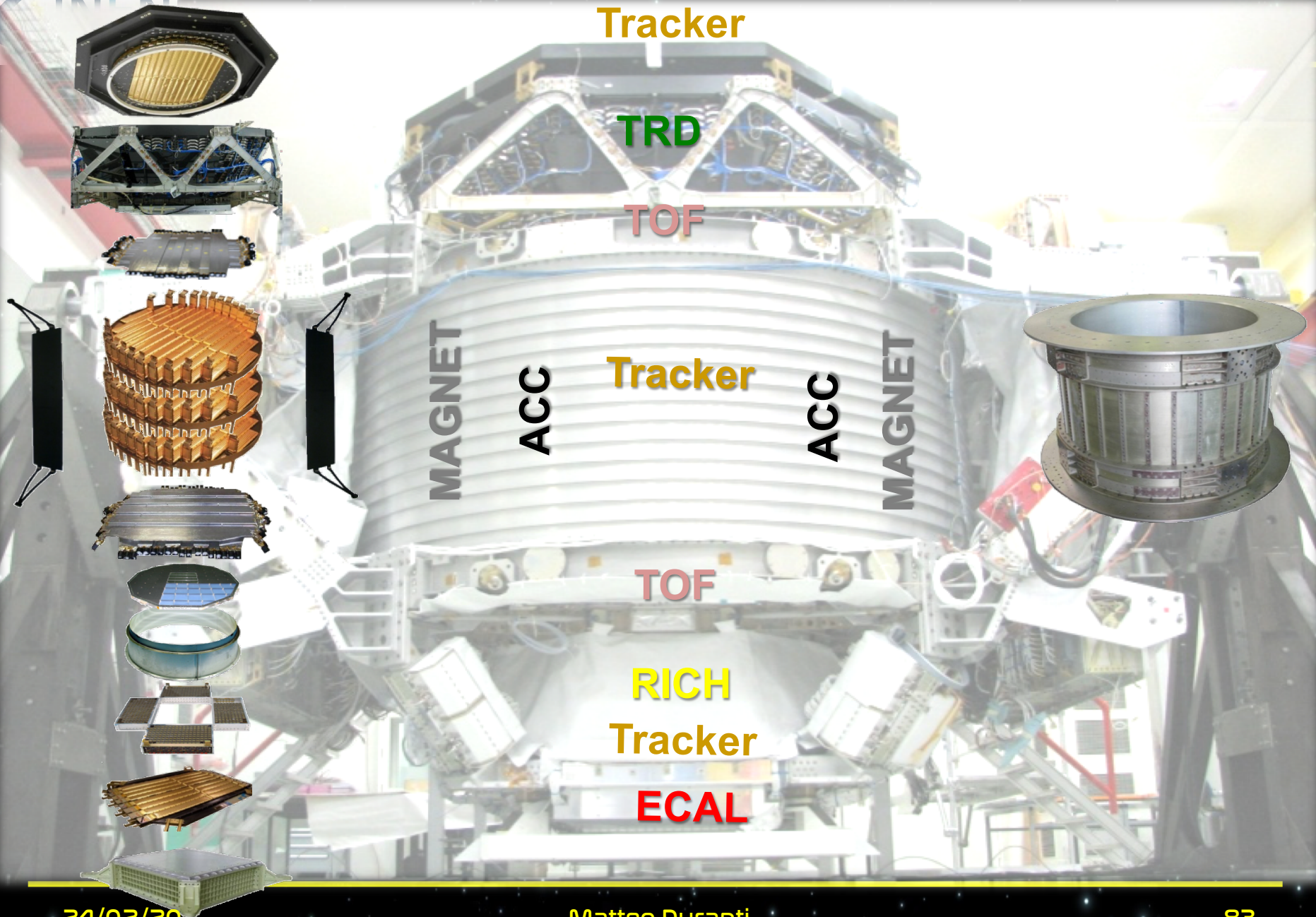
Stay tuned...

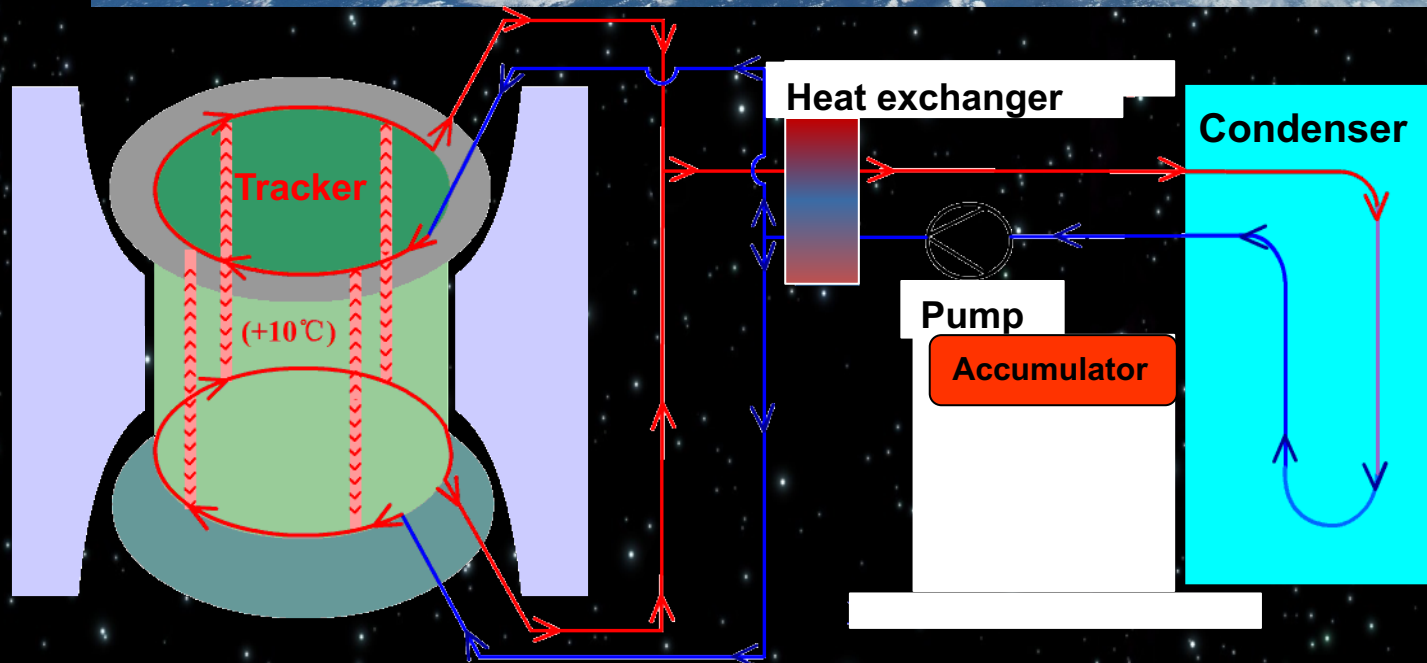
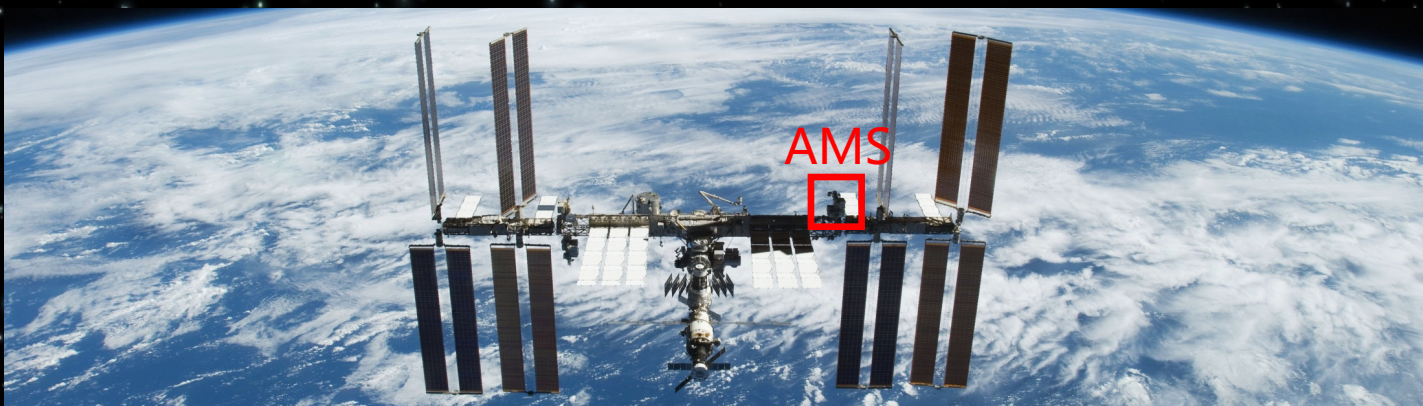


Backup









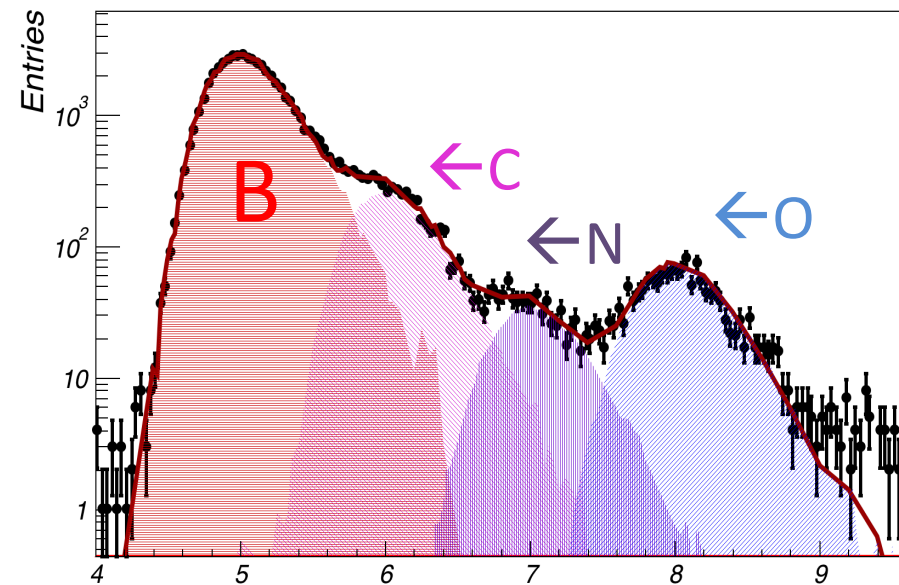
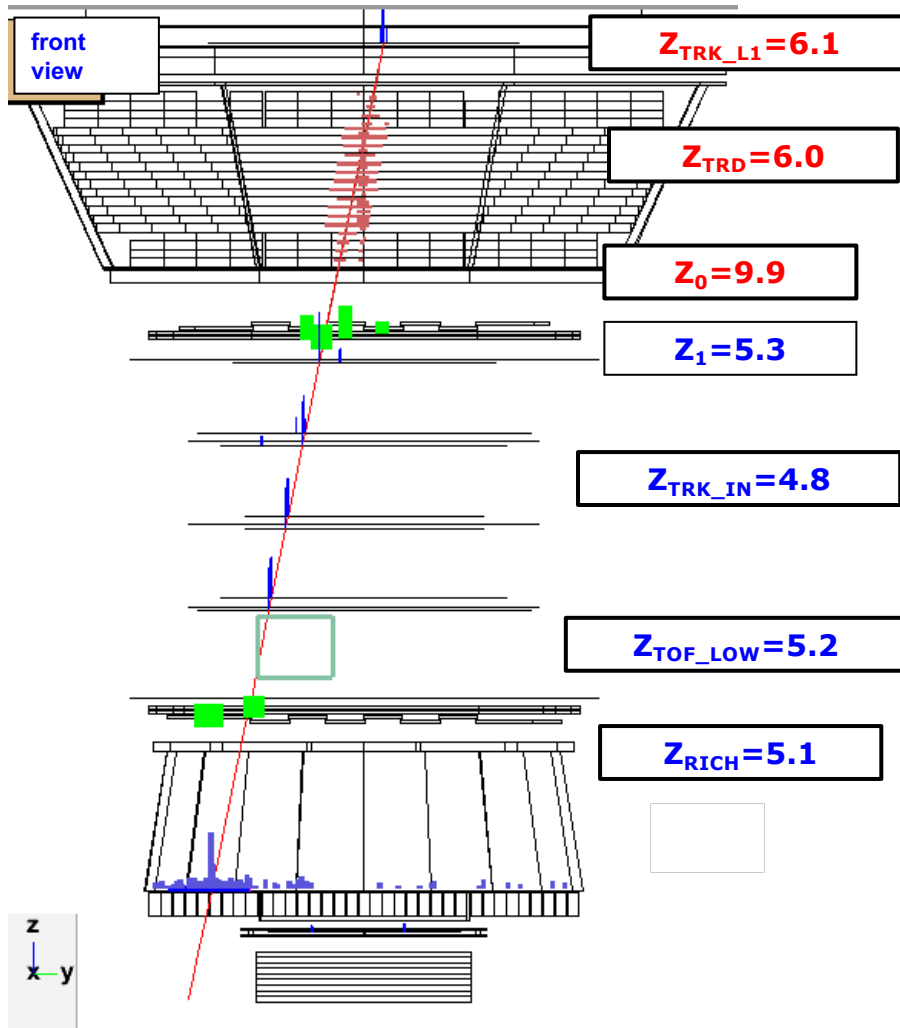
Red line: CO₂ gas/liquid two phase

Blue line: CO₂ liquid phase

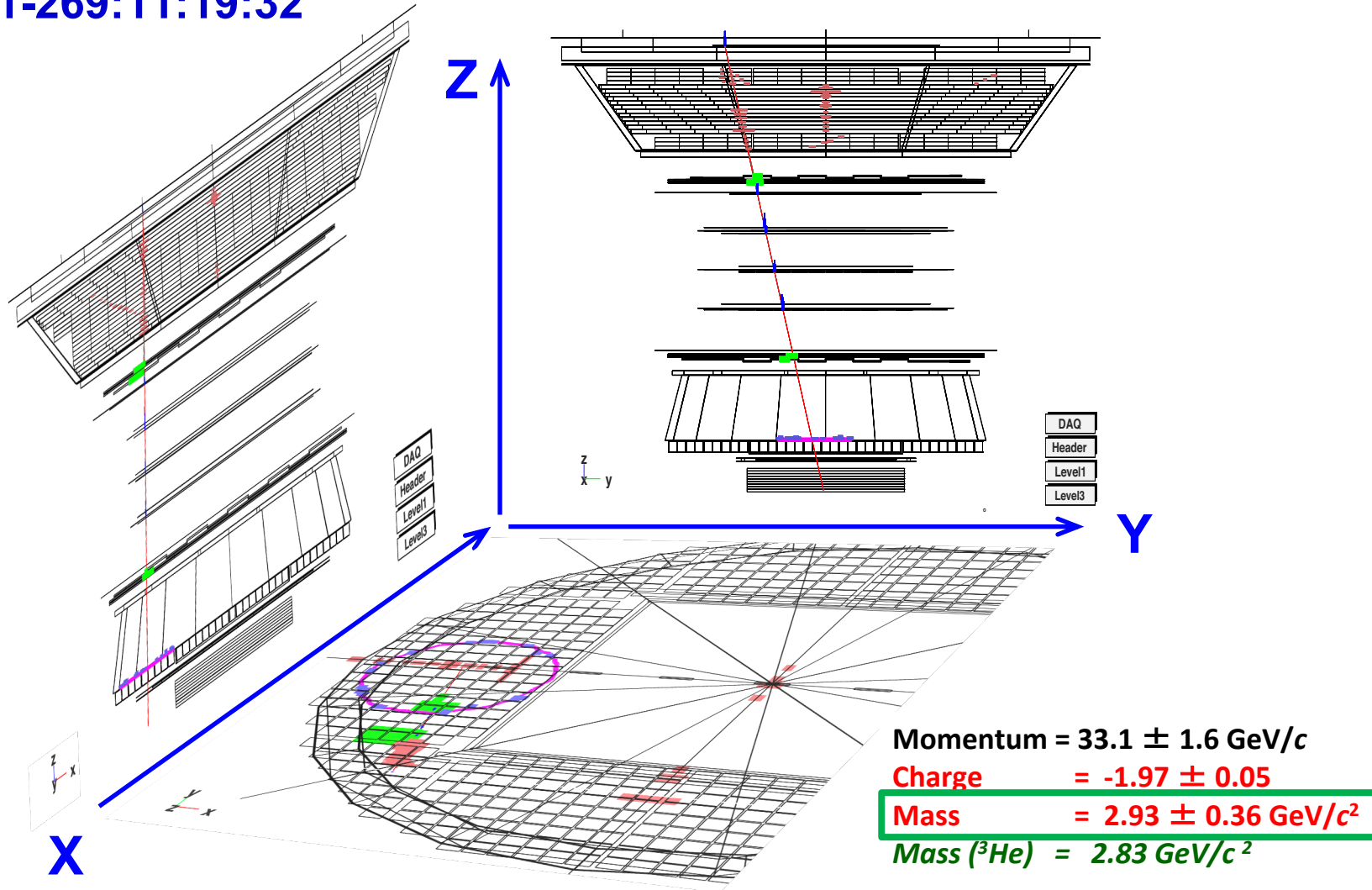
#SpacewalkforAMS



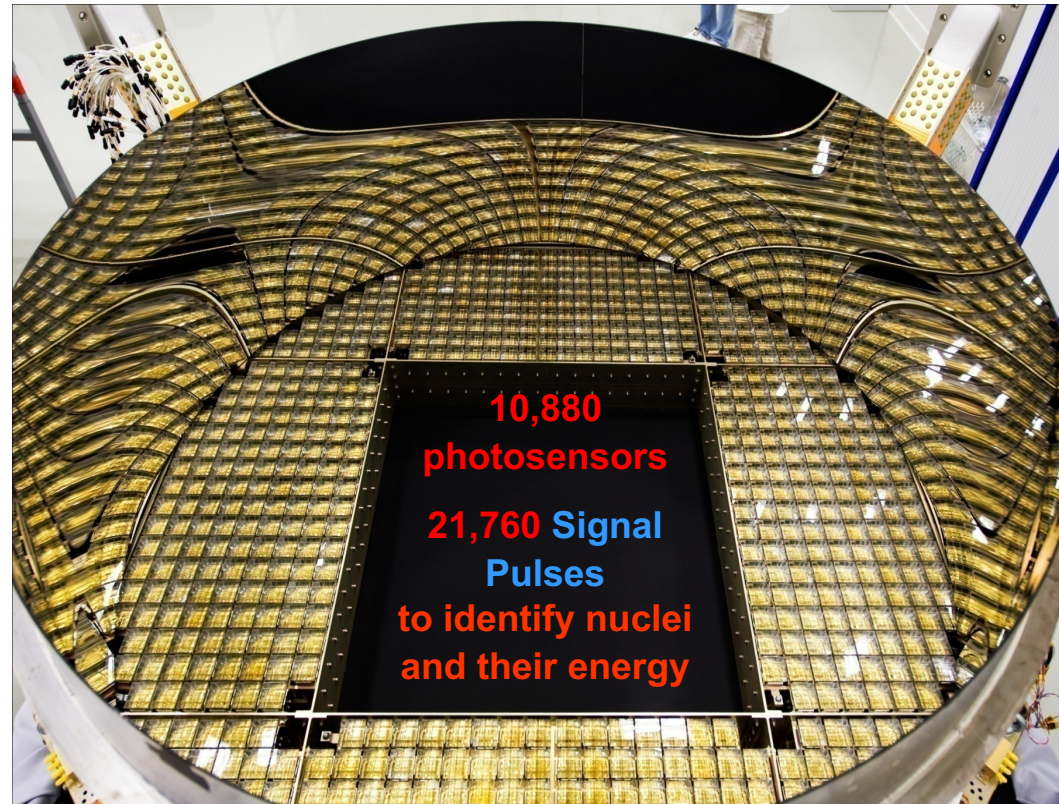
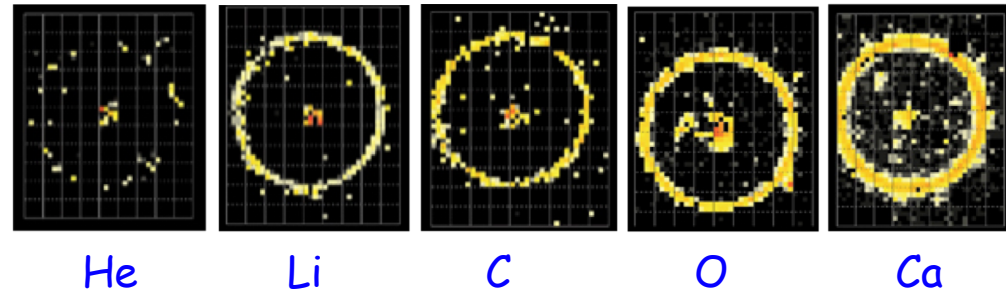
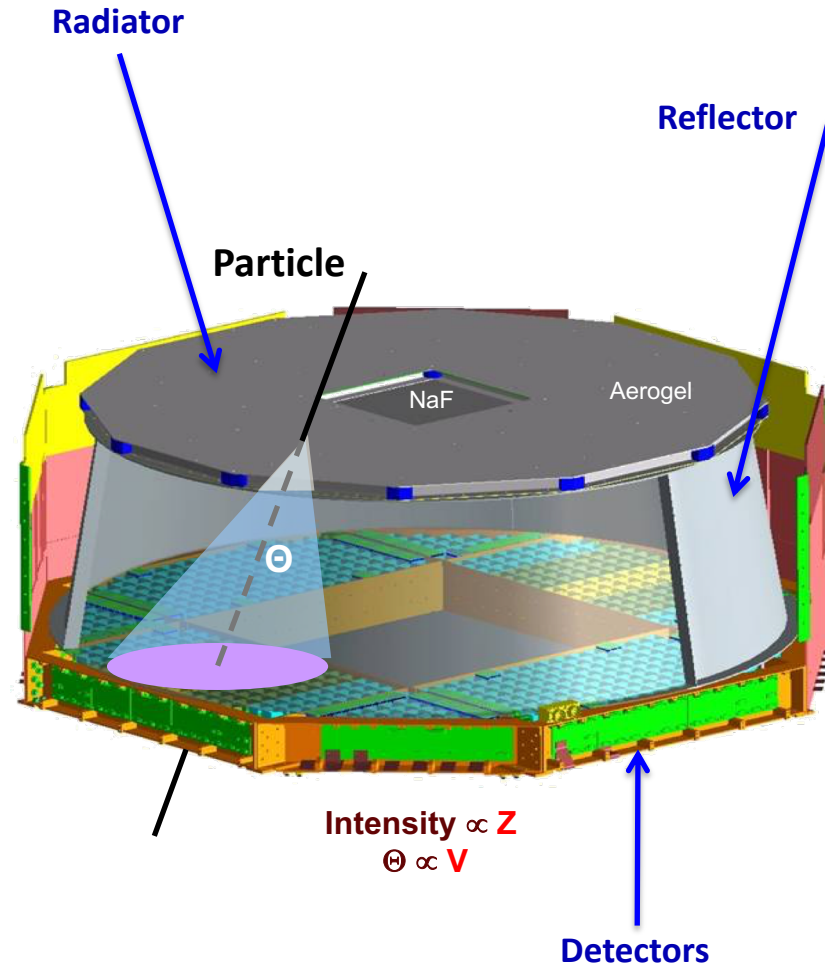
15 November 2019



Date: 2011-269:11:19:32

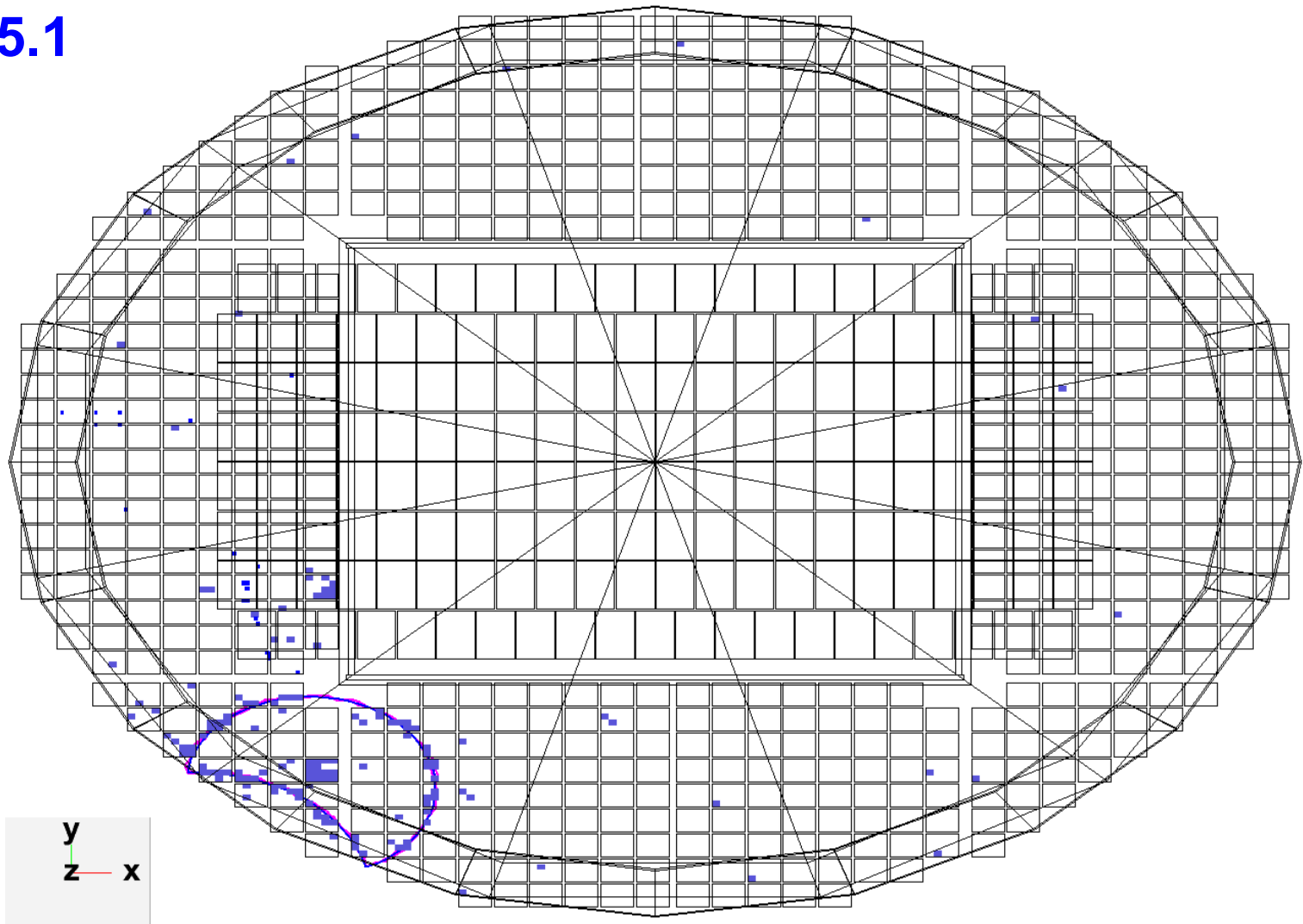


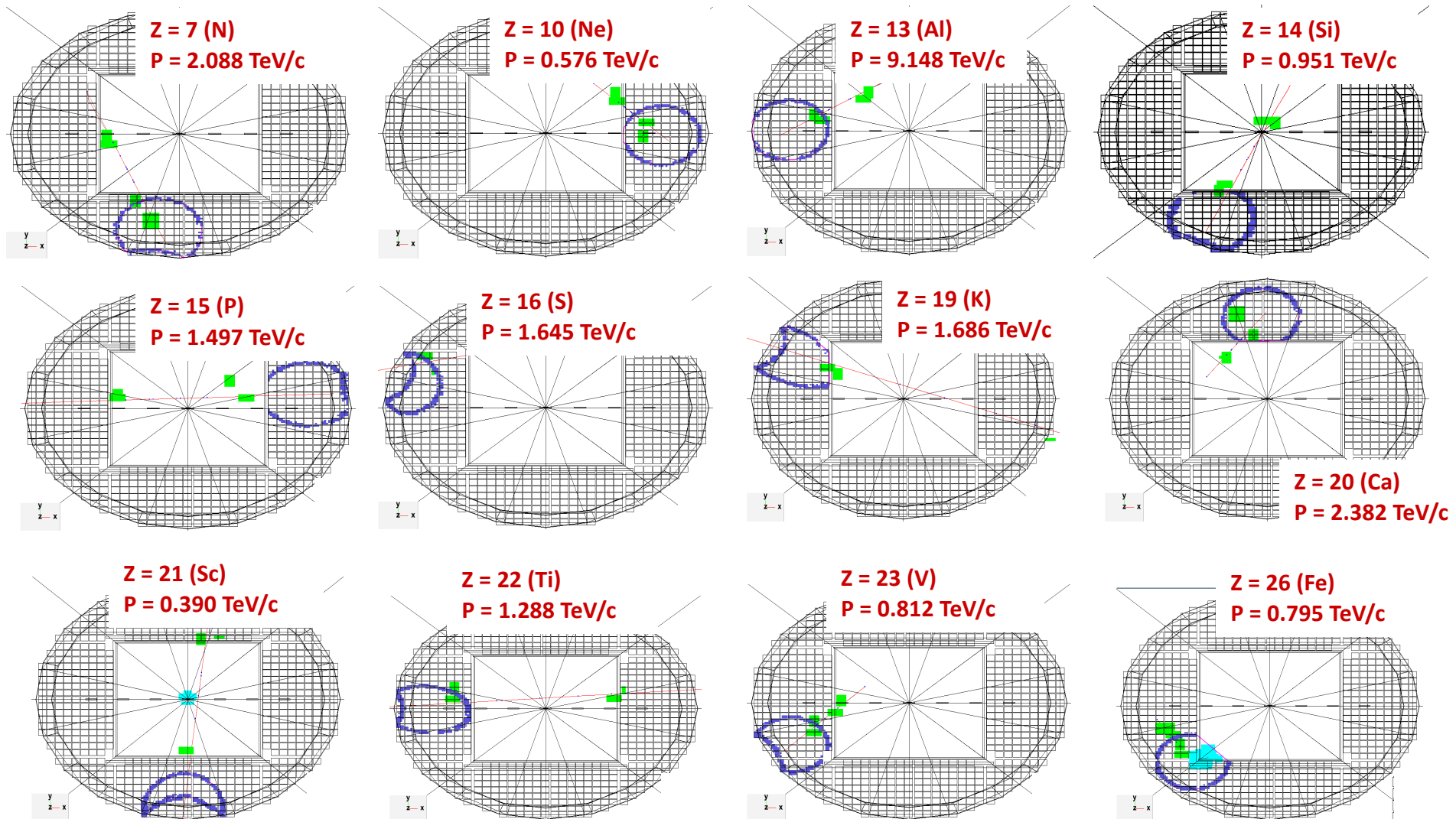
AMS-02 Ring Imaging Cherenkov



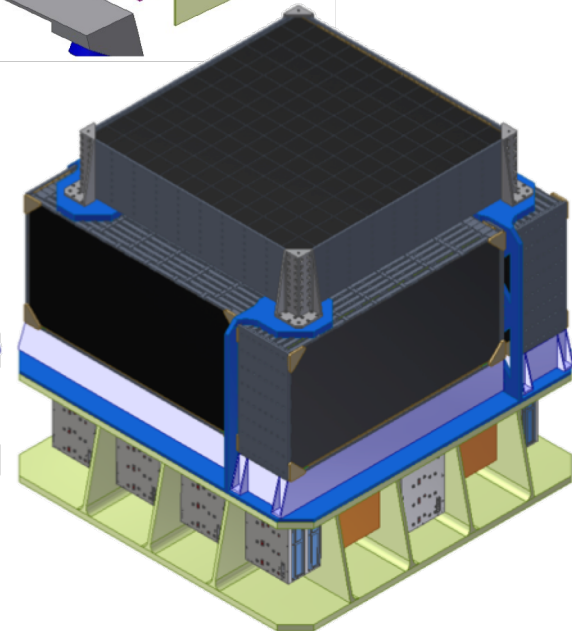
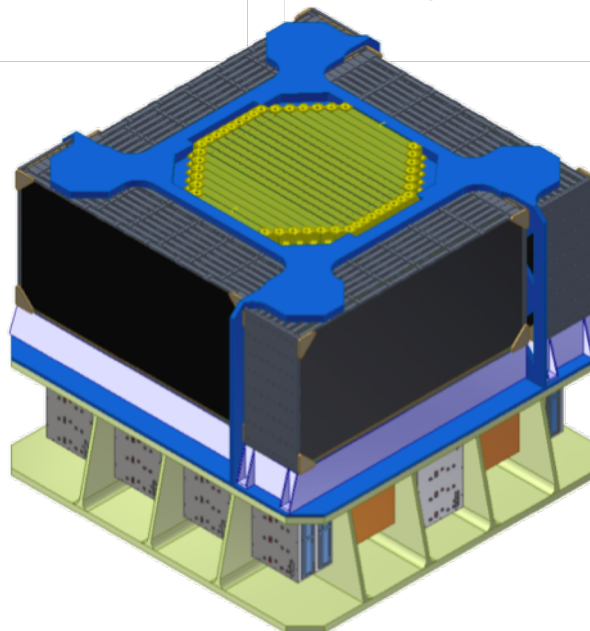
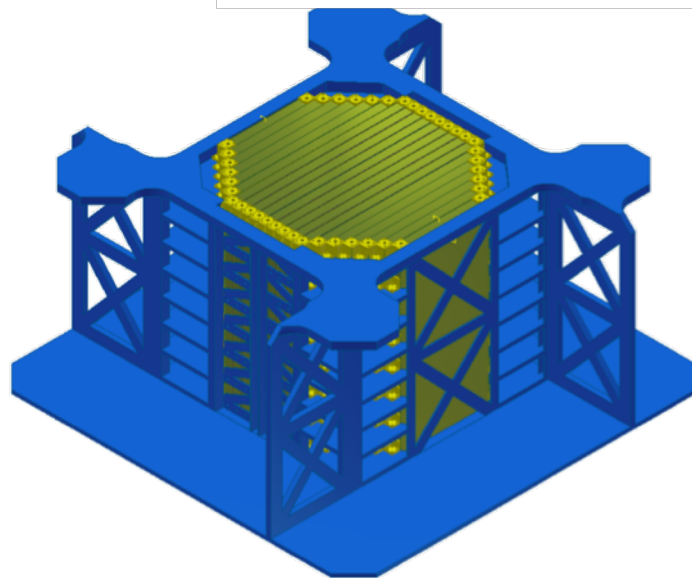
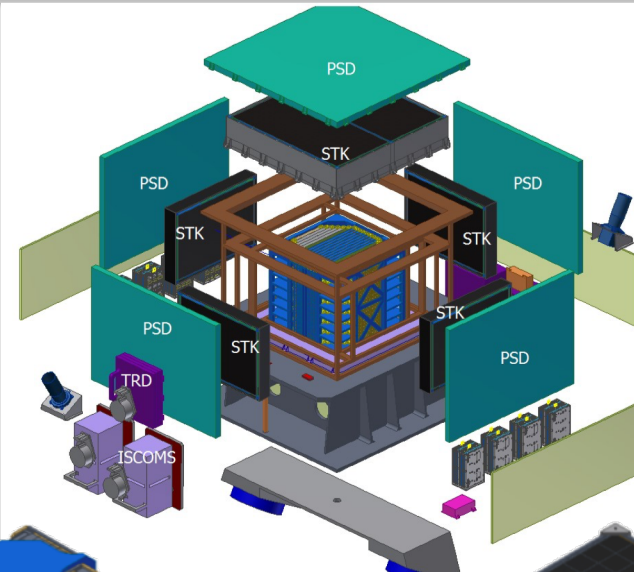
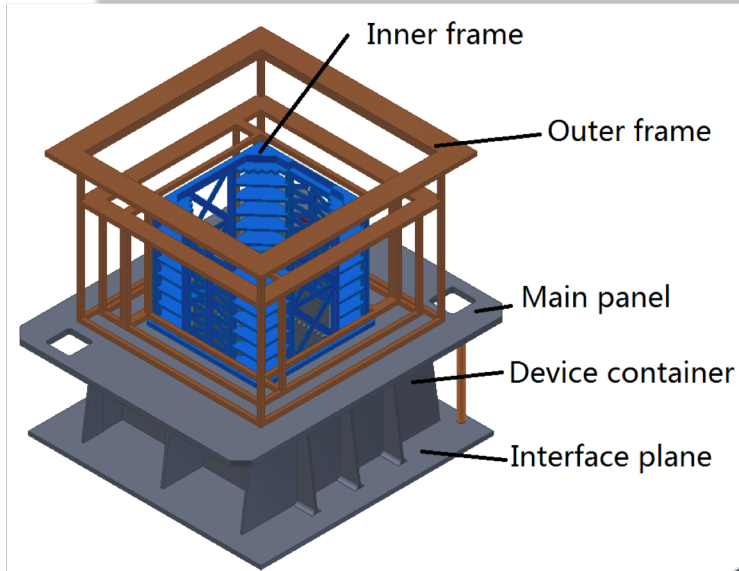
An AMS-02 RICH ion ring

$Z_{\text{RICH}} = 5.1$





HERD detector



Benefit from the R&D of high temperature superconducting magnets (MgB_2 , YBCO and in particular REBCO) for space applications ($T \approx 15 \div 40^\circ \text{ K}$)

Field 0.8 T

Bending power $> 1.1 \text{ T m}$

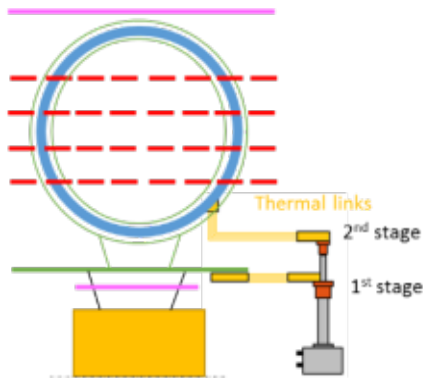
Weight $\sim 1000 \text{ Kg}$

The natural place for this kind of detector is the L2 Lagrangian point.

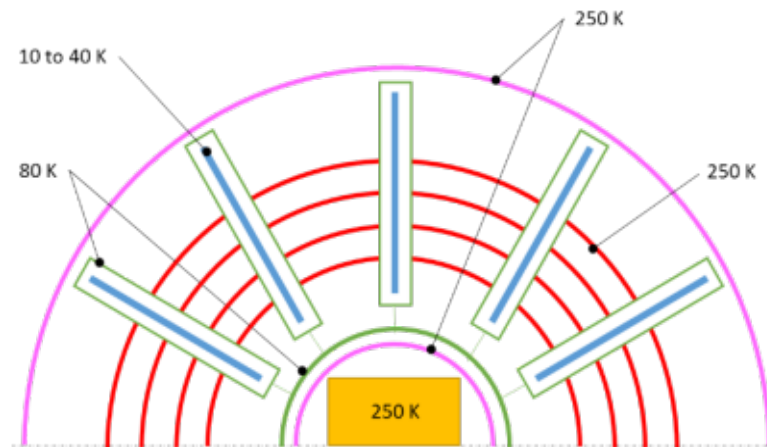
Most of the heat flux from the Sun is passively intercepted by an umbrella-type shield made of V-groove layers (inspired by the one of the J. Webb telescope).

The temperature on the dark side of the sunshield can be estimated $\sim 60 \text{ K}$.

Side cross-sectional view



Top cross-sectional view



ToF + trigger
Mechanical support

Superconducting coil
Calorimeter + trigger

Tracker

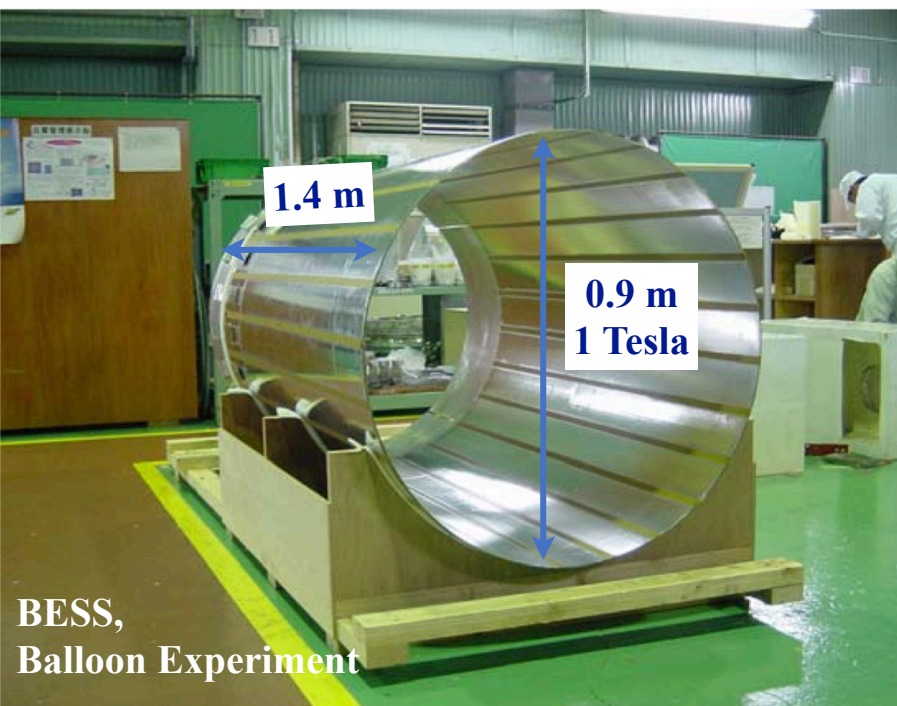
Cryogenic shield + thermal link

Cryocooler

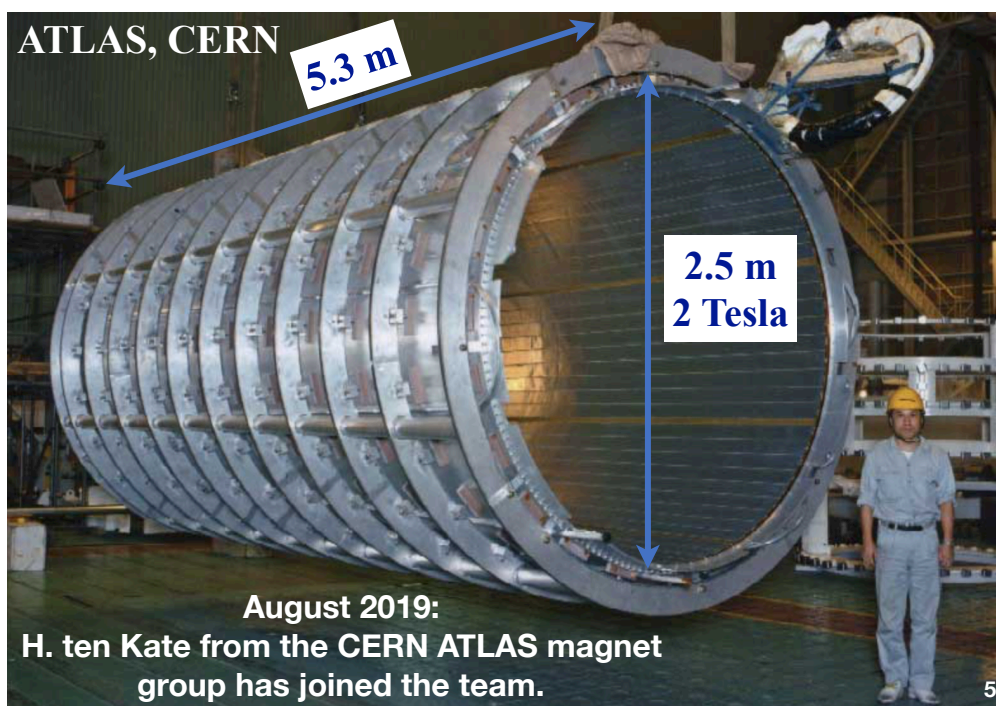
AMS-100 magnet

Example of Thin Solenoids using Low Temperature Superconductors (Nb-Ti) at $T = 4$ Kelvin

The coil weighs 43 kg and has a radial thickness of 3.4 mm and was built at KEK, Japan.

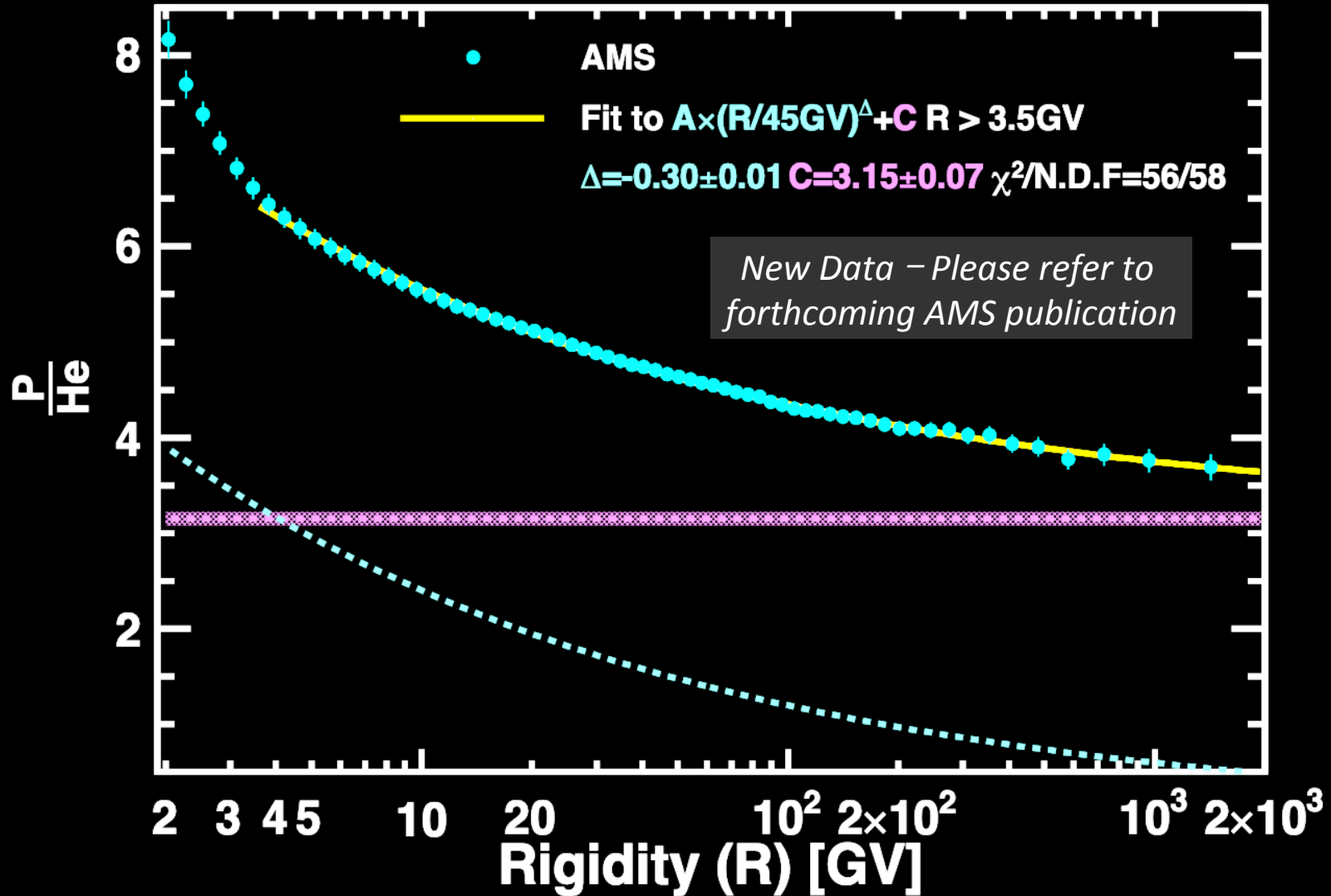


The coil weighs 5.5 tons and has a radial thickness of 4.5 cm and was built at Toshiba, Japan.

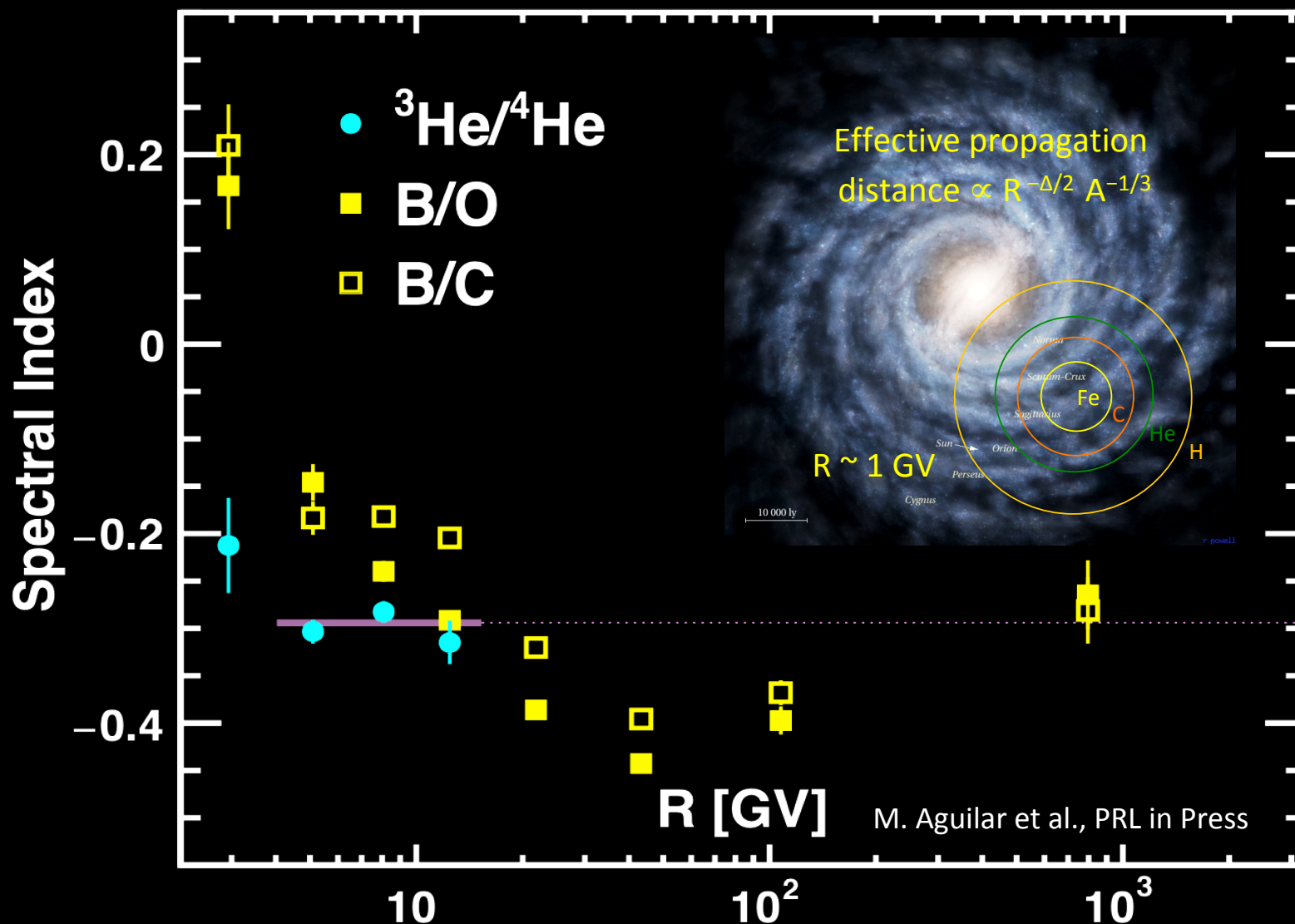


p/He Flux Ratio

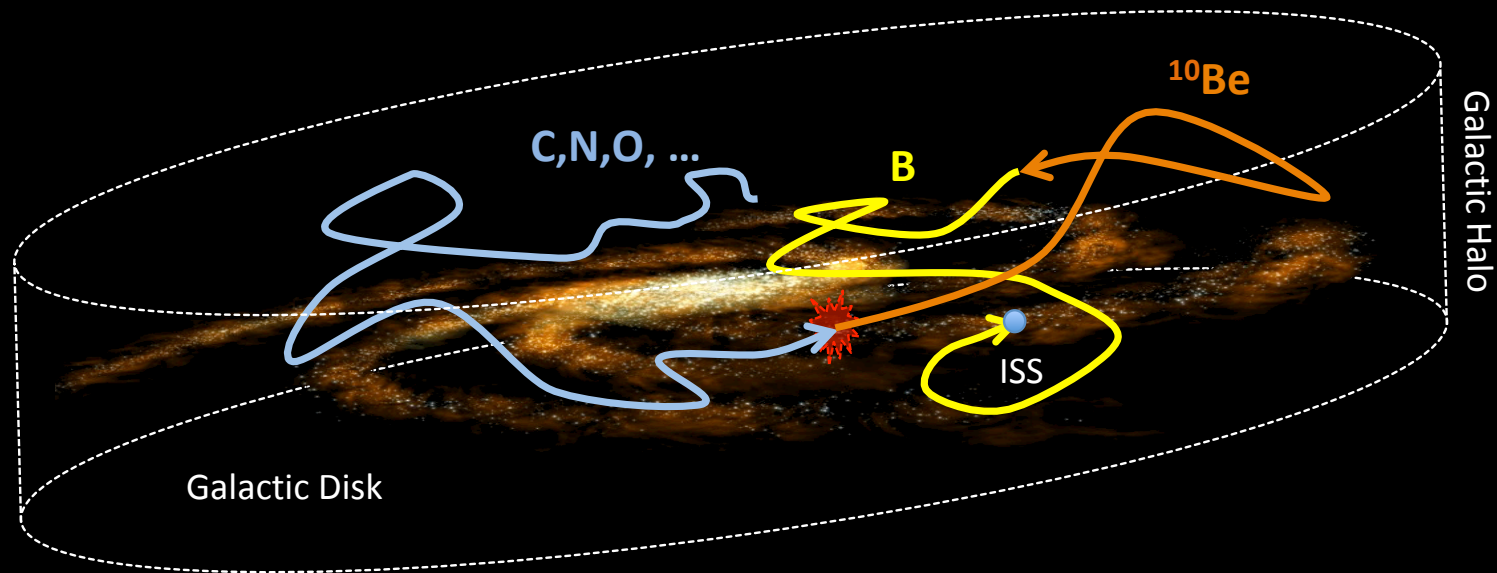
7 years update of M. Aguilar et al., PRL **115** (2015) 211101



Probing Non-Homogeneous Diffusion: AMS $^3\text{He}/^4\text{He}$ Ratio

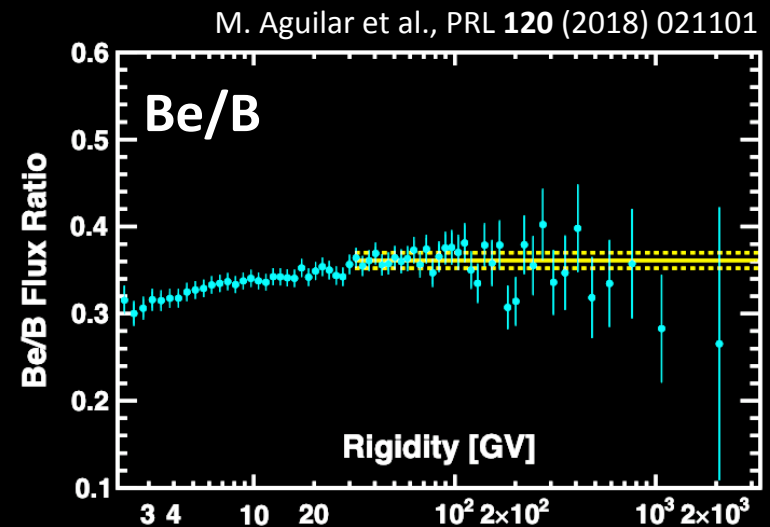


Cosmic Ray Clock: AMS Be/B Flux Ratio



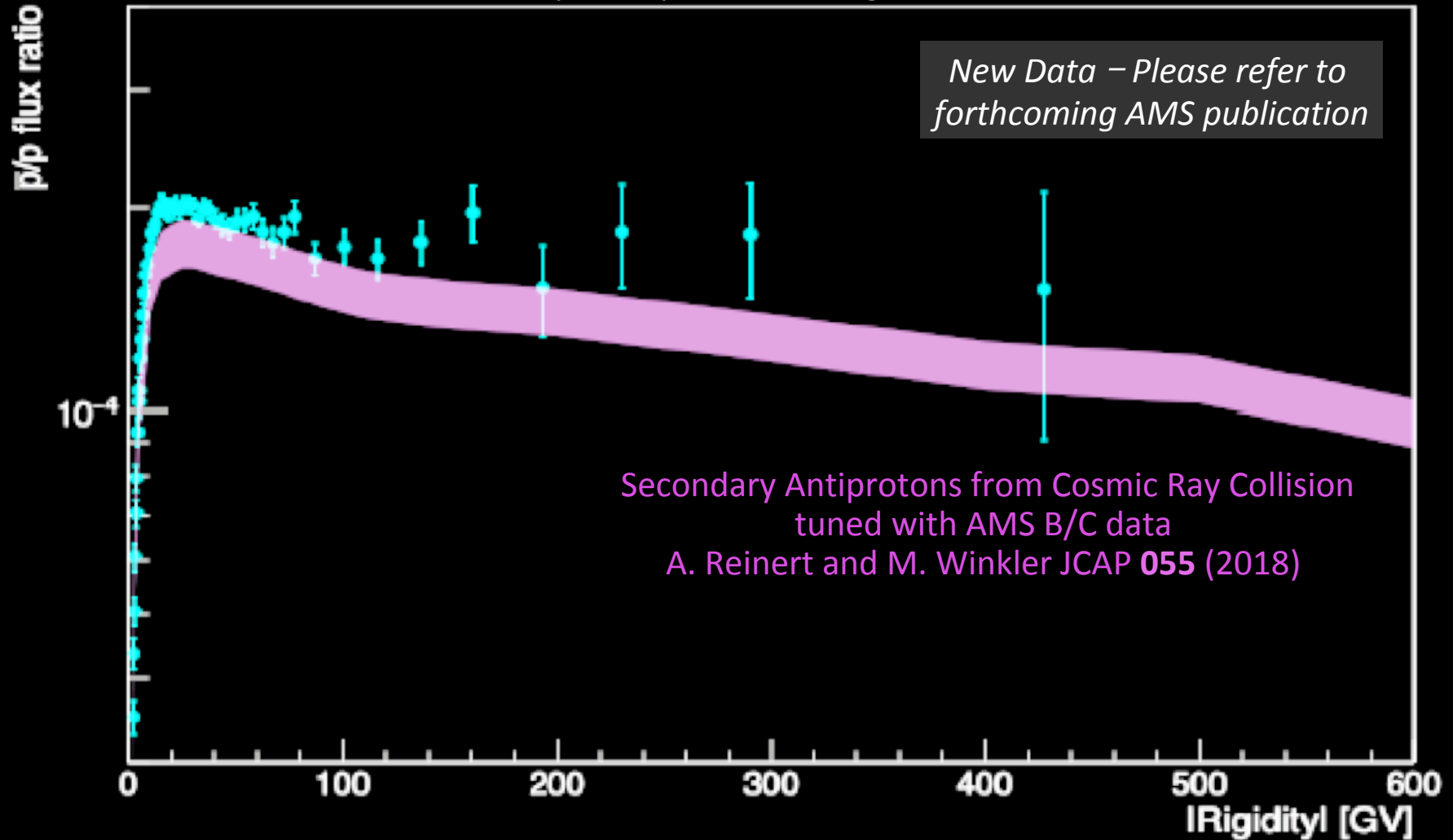
The secondary ^{10}Be beta-decays with $t_{1/2} = 1.4 \text{ My}$ through $^{10}\text{Be} \rightarrow ^{10}\text{B} + e^- + \bar{\nu}$.

The Be/B ratio rigidity dependence is related to the **cosmic rays confinement time** (or the galactic halo size in diffusion models).

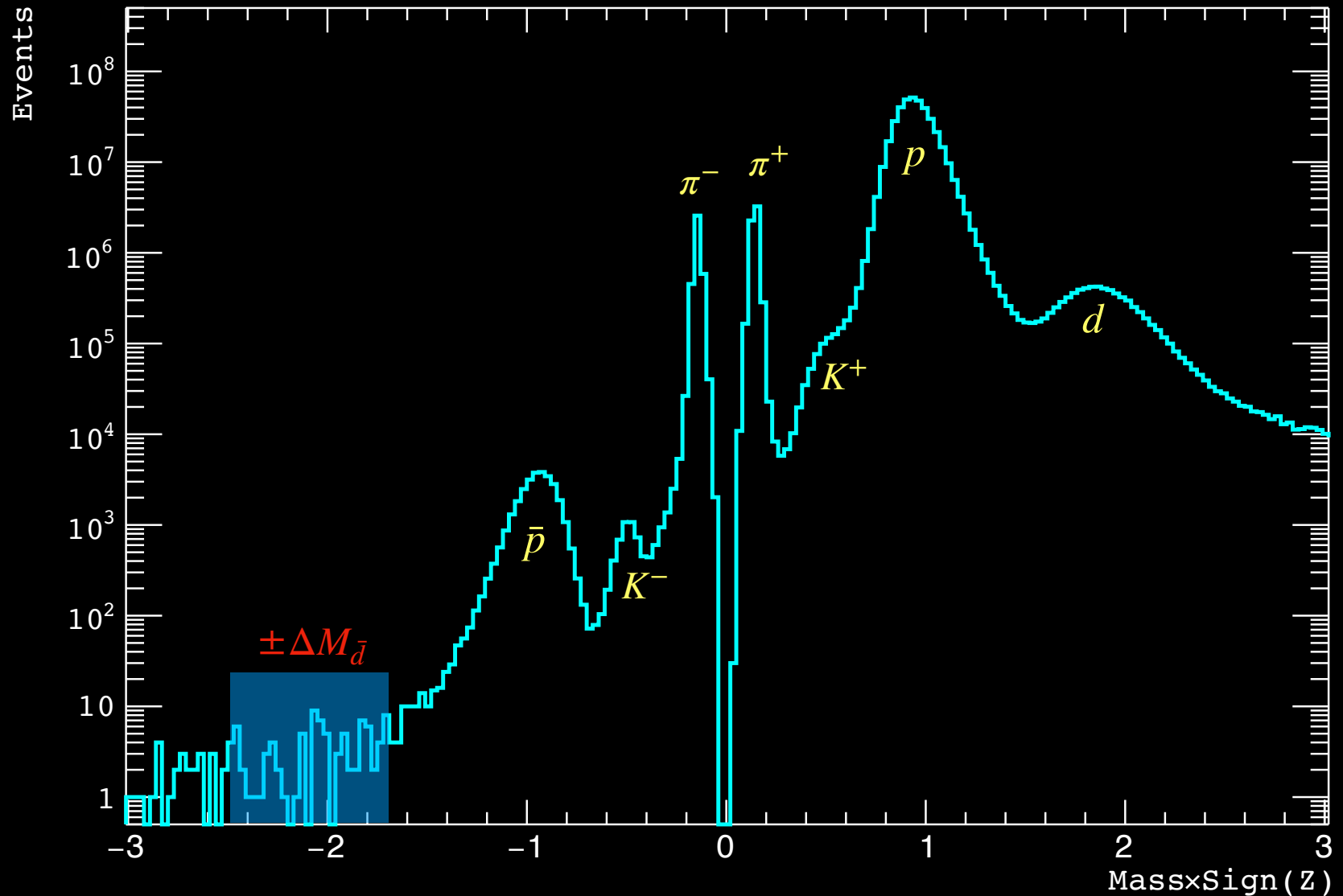


Anti-Proton / Proton Ratio

7 years update of M. Aguilar et al., PRL **117** (2016) 0911003



Antideuteron Search



Timing in an astro-particle tracker

Including the timing into the Tracker of an astro-particle detector permits to:

- substitute (or provide full redundancy to) any other **ToF detector** (i.e. planes of scintillators) in measuring $\beta \rightarrow$ isotopic composition for nuclear species (combined with E or p measurement);

$$M = \frac{E}{\gamma} = \sqrt{1 - \beta^2} E \quad \frac{\delta M}{M} = \frac{\delta E}{E} \oplus \beta^2 \gamma^2 \frac{\delta \beta}{\beta}$$

With 20% energy resolution (doable @5-10 GeV for protons?) the mass resolution cannot be never below 20%

With $\delta\beta/\beta = 2\%$ (i.e. 60 ps @ 1 m) the velocity uncertainty term dominates if $\gamma^2 > O(10) \rightarrow \gamma > O(3) \rightarrow E_p > 3 \text{ GeV}$

\rightarrow d/p doable maybe...

\rightarrow ${}^3\text{He}/{}^4\text{He}$ already ruled out...